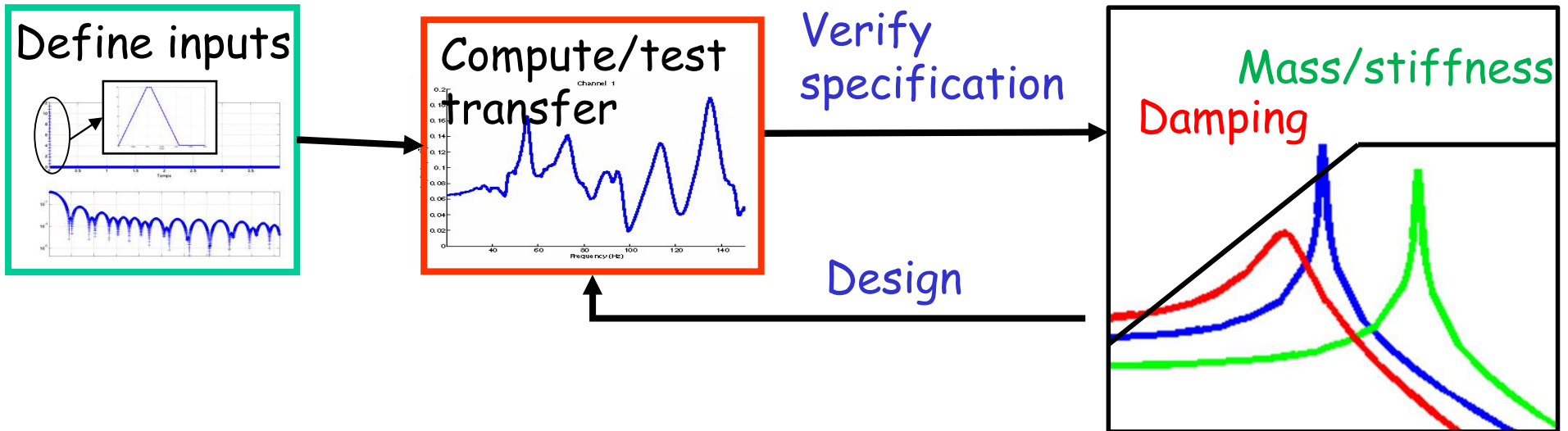


Damping and complex modes

1. Devices
2. Coupling/energy levels
3. Damping mechanisms, constitutive laws
4. Reduced models for damping: from material to structure

Design process



Designing for damping is difficult

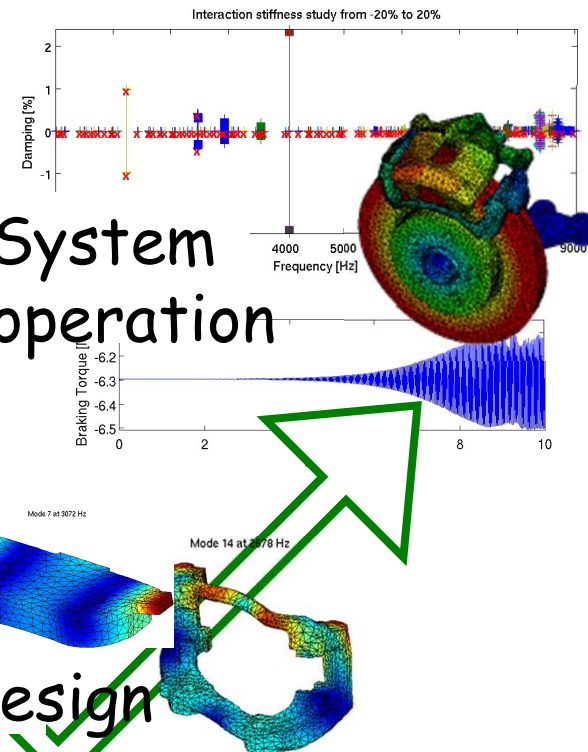
- Physical damping models are computer intensive
- NVH response typically evaluated at system level

Concept

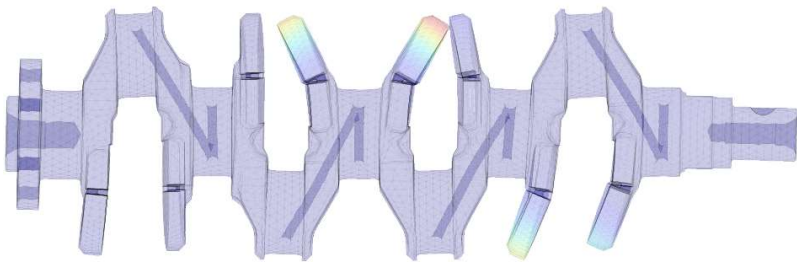
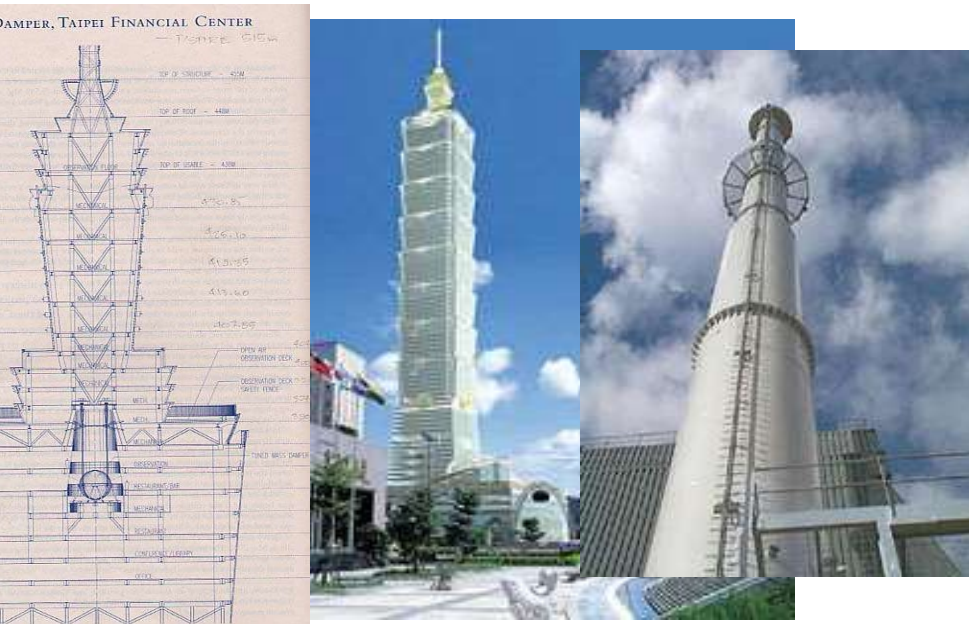
Requirements
&
architecture

Component design

System
operation

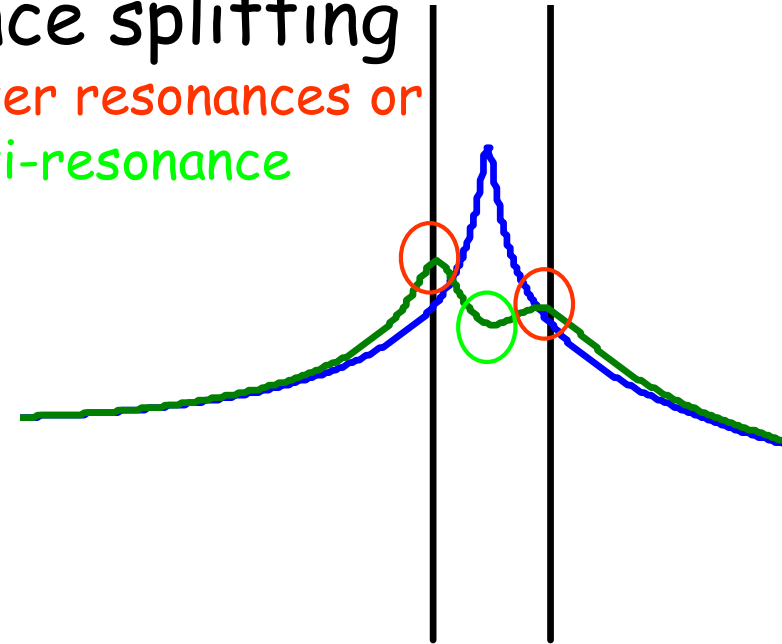


Tuned mass dampers / vibration absorber



- Resonance splitting

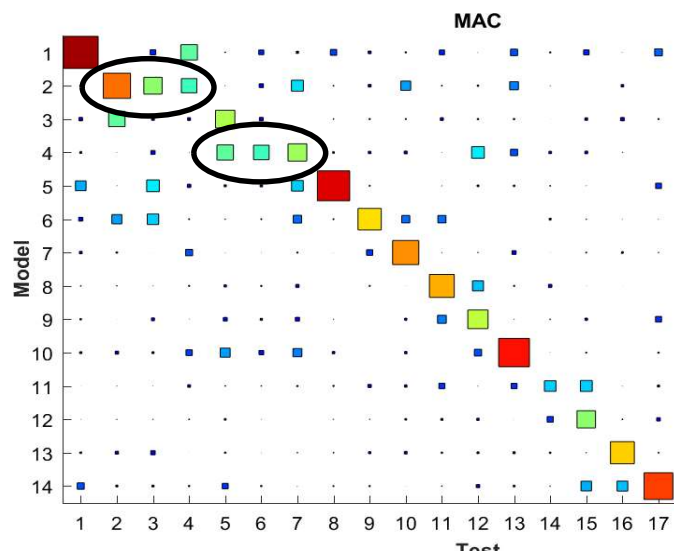
- Two lower resonances or
- One anti-resonance



- Countless applications :
helicopters, buildings, lamp
posts, cars, ...
- Current trend : non-linear
absorber (self tuning)

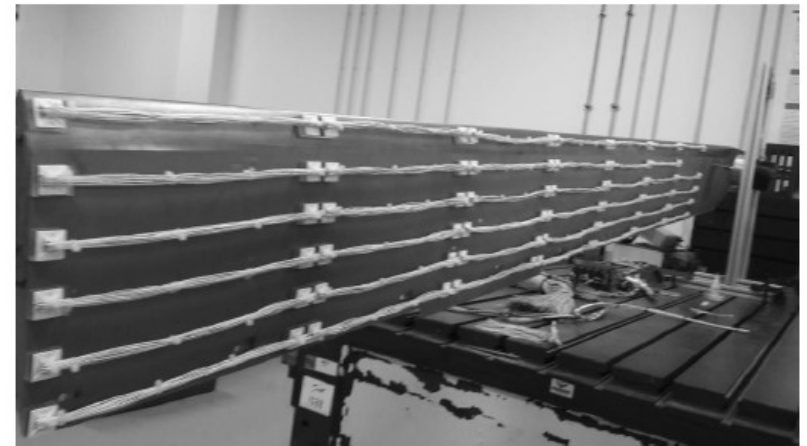
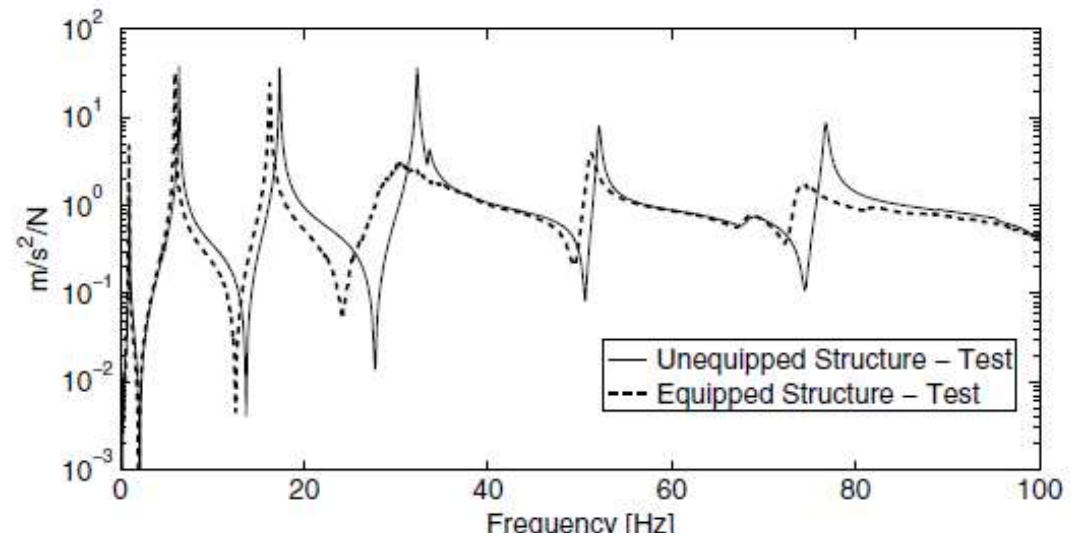
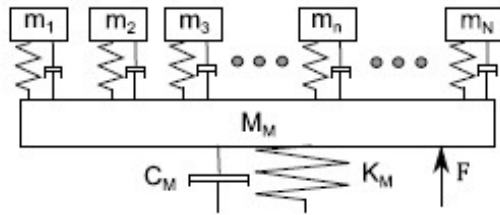
Sample dampers

- Fluid sloshing : No moving parts
- Rings, cables, side masses, ...
- Electrical wires



www.multitech-fr.com

Damping due to local modes



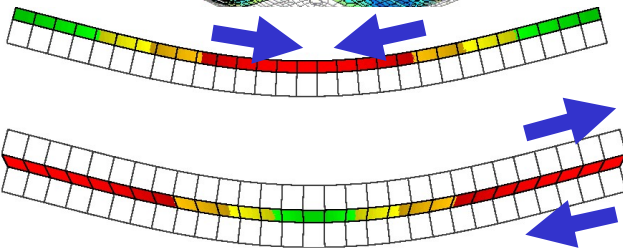
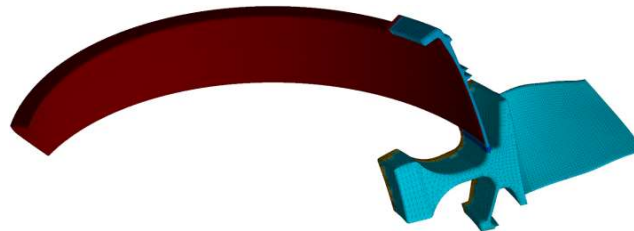
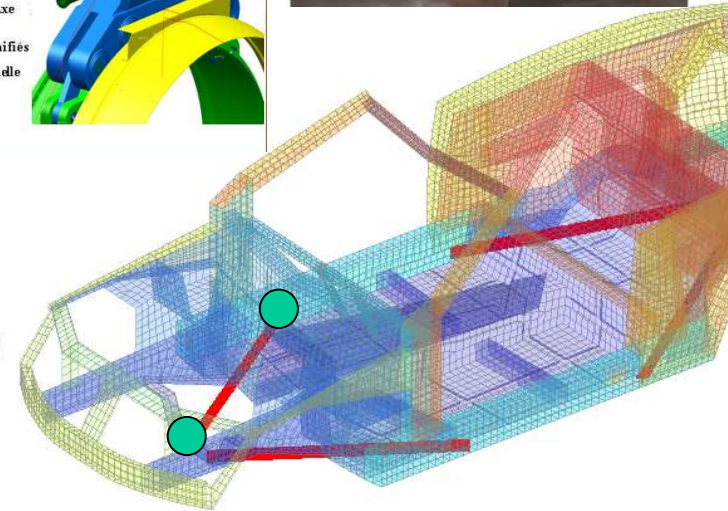
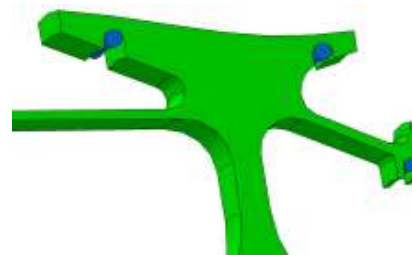
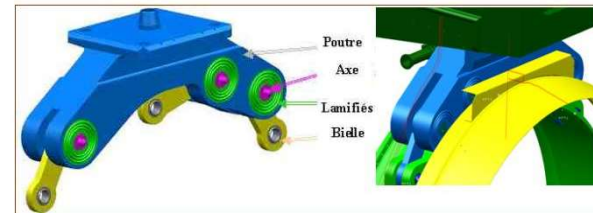
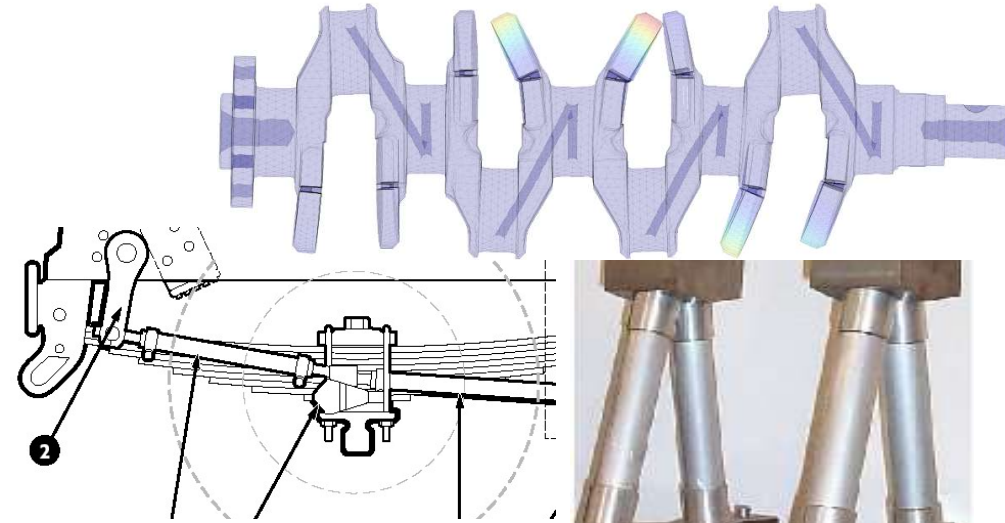
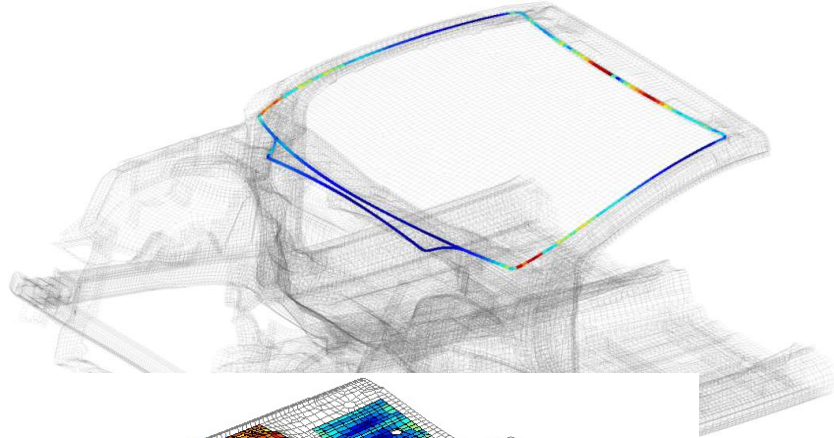
Soize "structural fuzzy" 1987

Arnoux, A. Batou, C. Soize, L. Gagliardini. JSV 2013

Loukots, Passieux, Michon. Journal of Aircraft 2017

Damping devices

- Resonant mass
- Struts, point connections
- Line/surface treatments



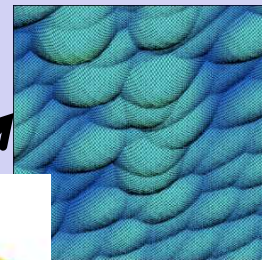
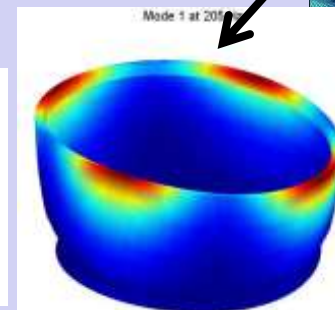
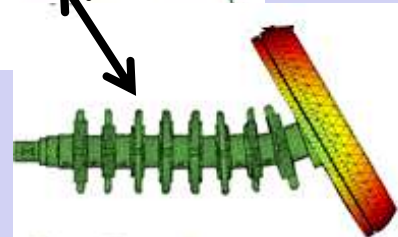
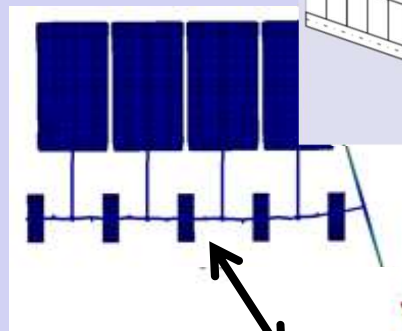
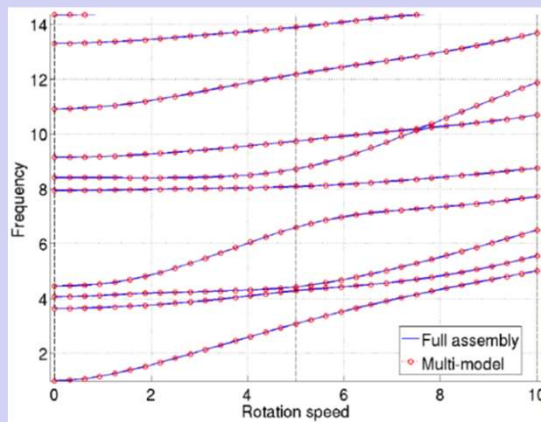
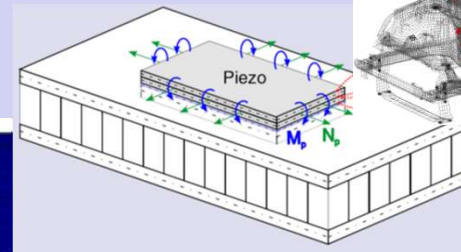
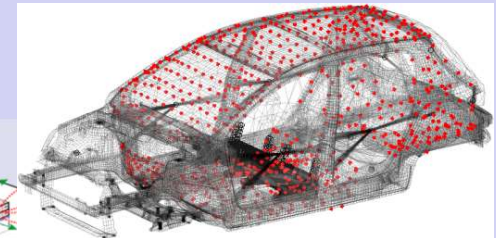
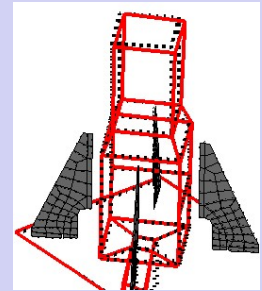
Q2 : undamped nominal + coupling

In

Undamped
reduced model

Sensors

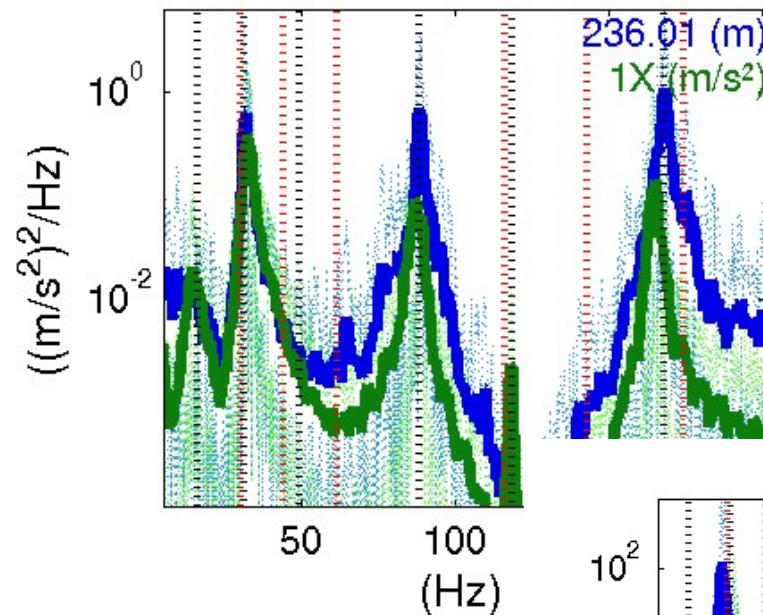
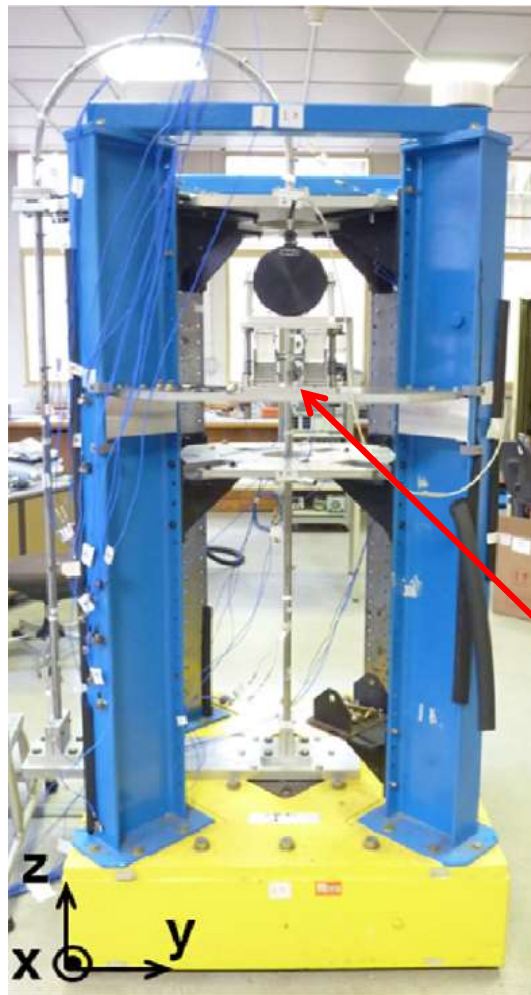
- **Coupling** : test/FEM, fluid/structure active control, ...
- **Local non-linearities** : machining, bearings, contact/friction, ...
- **Environment**/design/uncertainty



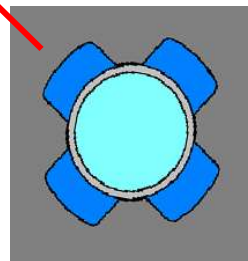
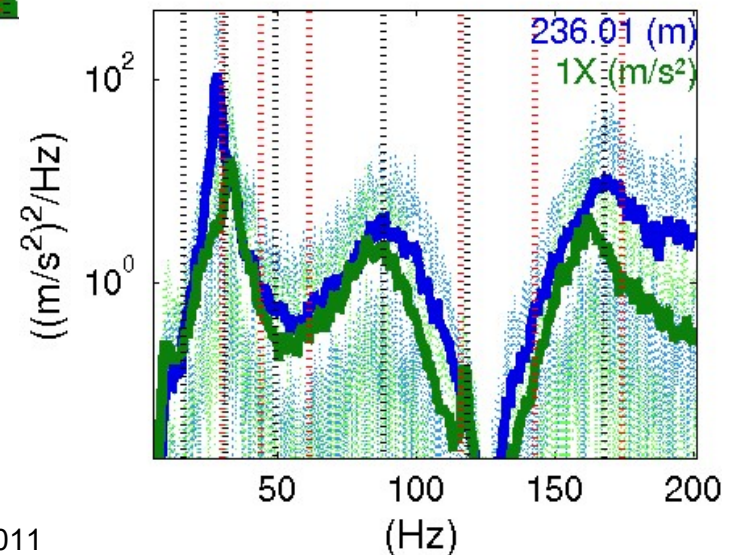
Energy coupling & NL system

Bending beam with gap & contact
Apparent damping with pure contact

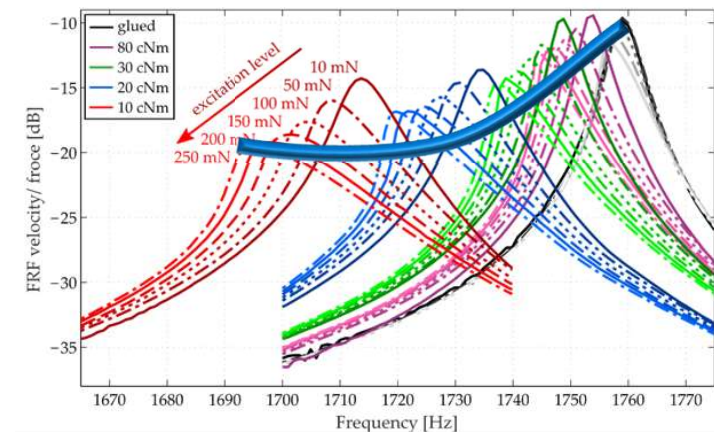
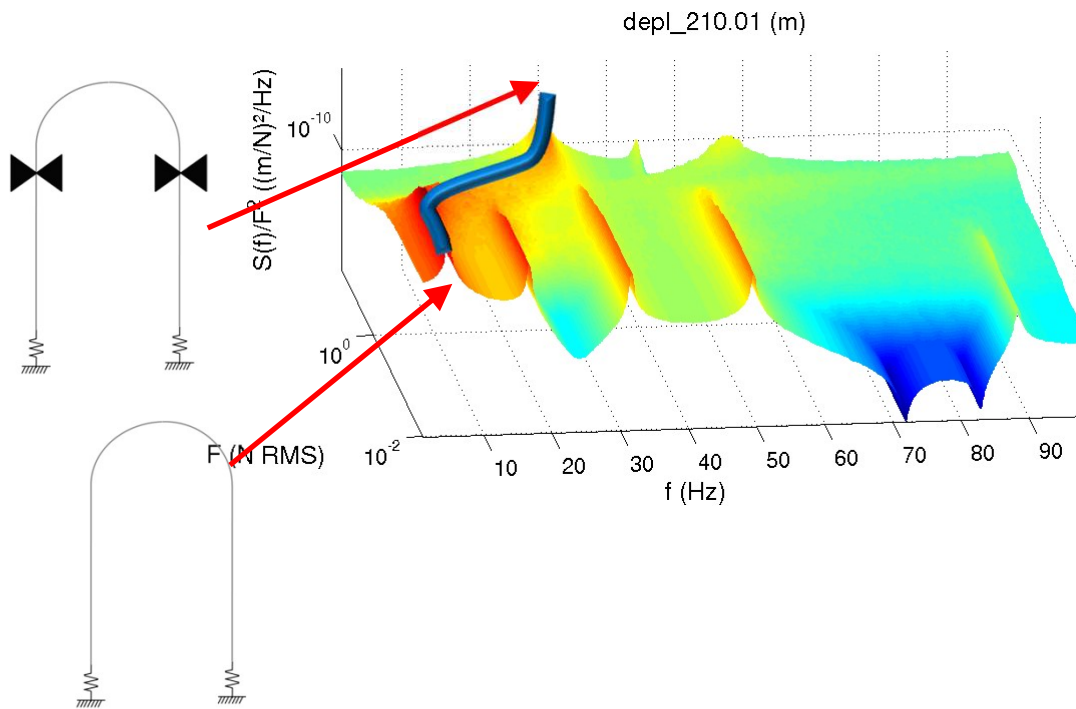
($F=0.33$ N RMS)



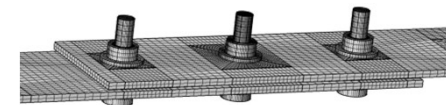
($F=2.52$ N RMS)



Damping occurs in the transition

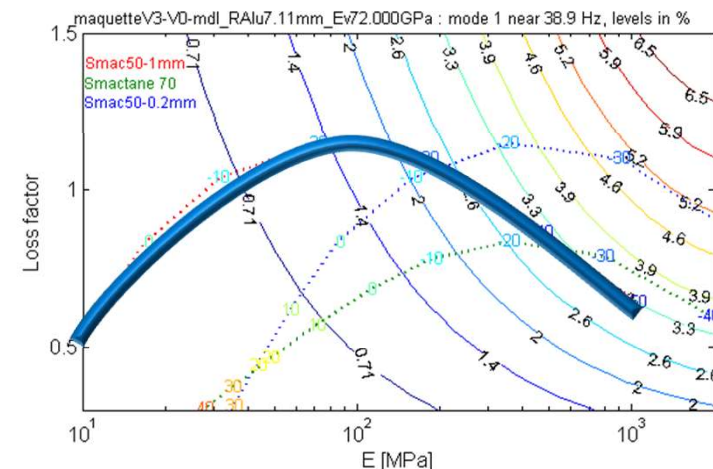


FEMTo-ST Orion Testbed



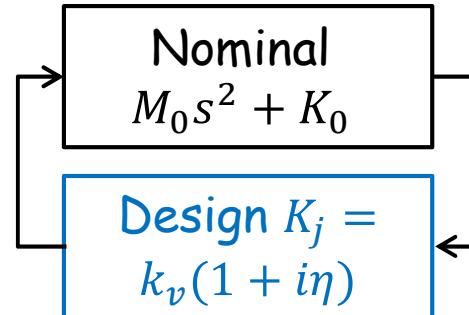
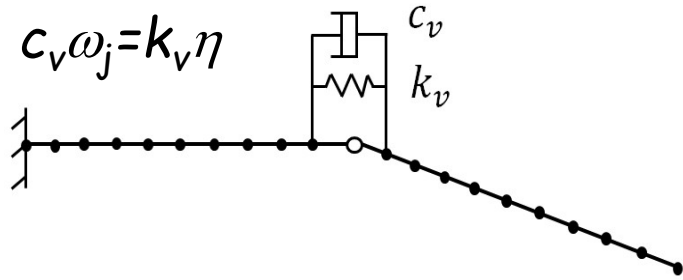
Transition

- Contact with broadband excitation **level**
- Friction : **amplitude/normal force** dependence
- Viscoelasticity : **temperature/frequency** dependence

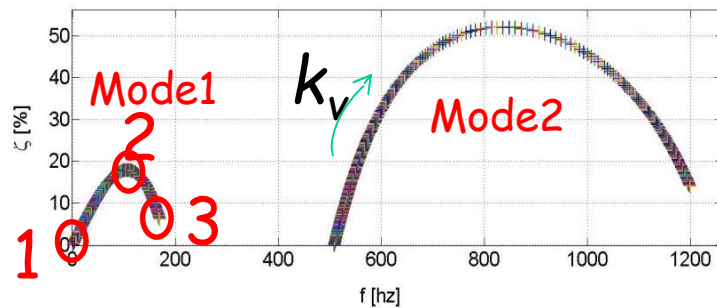


ENSAM/PhD. Hammami 2014
Hammami/Balmes/Guskov, MSSP 2015

A mechanical root locus (coupling)



k_v variable, $\eta=1$ constant loss
Output : $f(k_v, \eta) \zeta(k_v, \eta)$



- | | | |
|---|------------------|-------------------|
| 1 | 1.113 Hz 0.28 % | Soft joint |
| 2 | 98.06 Hz 17.79 % | Dissipative joint |
| 3 | 181.2 Hz 0.11 % | Stiff joint |

Damping can only exist
if joint is « working »

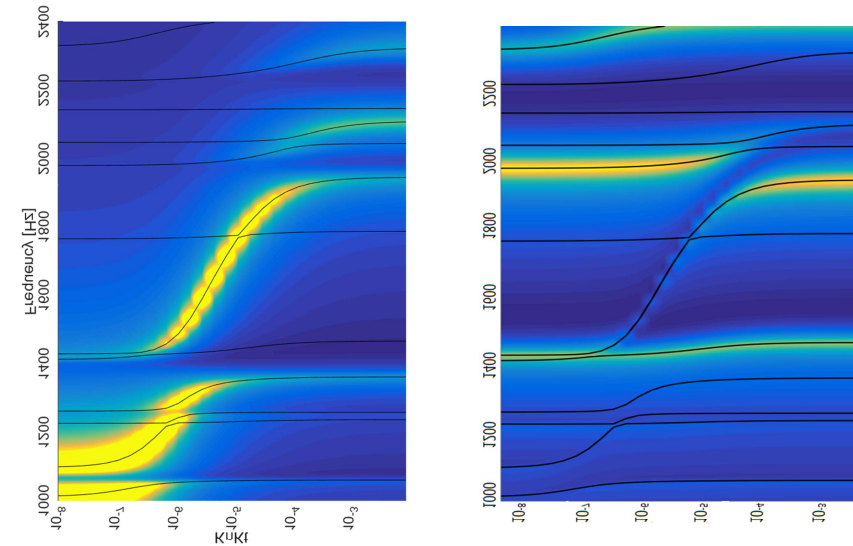
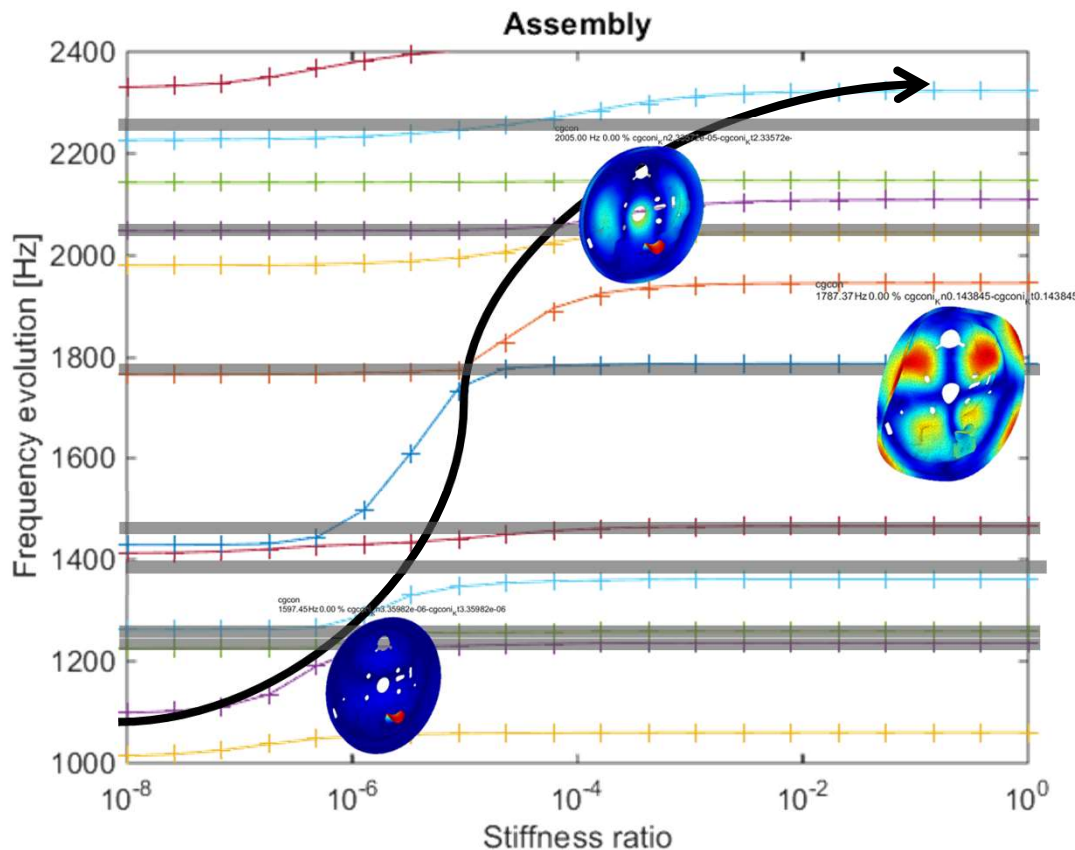


Energy fraction
in joint must be « sufficient »
(depends on k_v)

Related ideas : electro-mechanical coupling coefficient, pole/zero distance,
modal strain energy

Coupling & mode crossing

- Variable contact **surface**, **contact**, **sliding** can generate coupling
- Horizontal lines : global modes
- S curve : local cable guide mode



Sensors can help
But can we track modes better ?

Lab work 3 : tracking & reanalysis

Q3 : Damping mechanisms

Damping : mechanism by which the energy in the considered system is diminished

- Non elastic behavior
 - Material : viscoelastic, plastic, ...
 - Joint friction
 - Inter-frequency coupling through non-linearities
- Coupling with other systems
 - Fluid/structure, particle, soil, thermal, ... interactions
 - Active control systems (piezo, electromagnetic, ...)
 - Rotating parts

Linear, Non-linear

Viscoelasticity = LTI material

- Viscoelasticity (Bolzmannian material), stress f (strain history) linear in amplitude
- Viscous relaxation : convolution with displacement or velocity (**non-parametric**)

$$F = \int_{-\infty}^{+\infty} K(t-r) \mathbf{x}(r) dr = \int_{-\infty}^{+\infty} R(t-r) \dot{\mathbf{x}}(r) dr$$

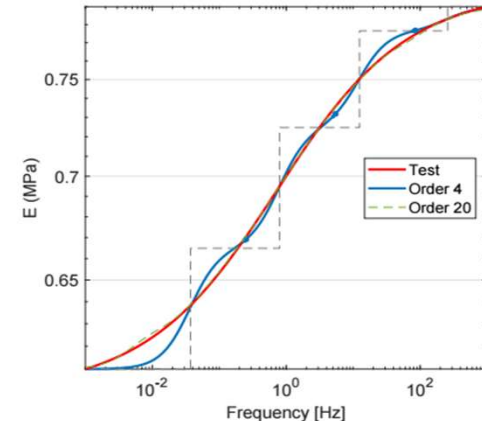
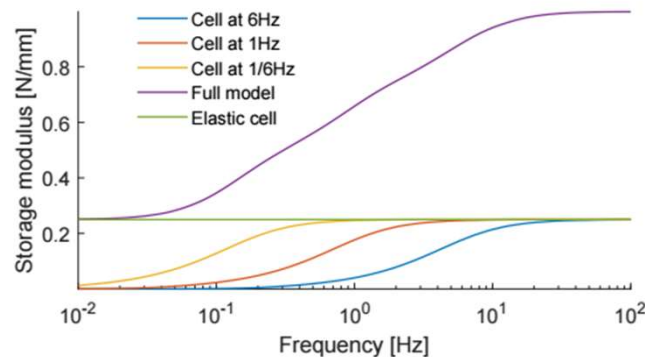
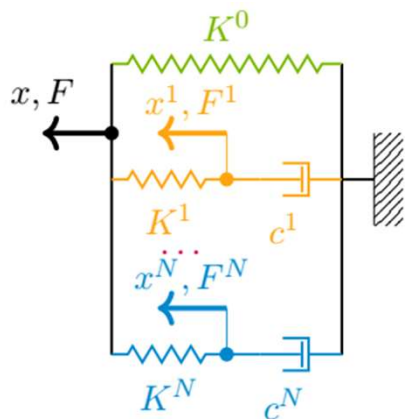
- **Parametric** model representation (Maxwell, rational fraction)

$$K(s) = K^0 \prod_{i=1}^N \frac{1 - s/z^i}{1 - s/p^i} = K^\infty \left(1 + \sum_{i=1}^N -\frac{g^i}{s/\omega^i + 1} \right) = K^\infty \left(g^0 + \sum_{i=1}^N -\frac{g^i s}{s + \omega^i} \right)$$

- Differential equation formulation (stress or stress rate relaxation)

$$\frac{\dot{F}^i}{\omega^i} + F^i = -g^i F^\infty(x), \text{ or } \dot{F}^i + \omega^i F^i = g^i \dot{F}^\infty(x)$$

- Order selection: place poles and zeros in frequency domain ($> 1/\text{decade}$)



Viscoelasticity 2 : frequency domain

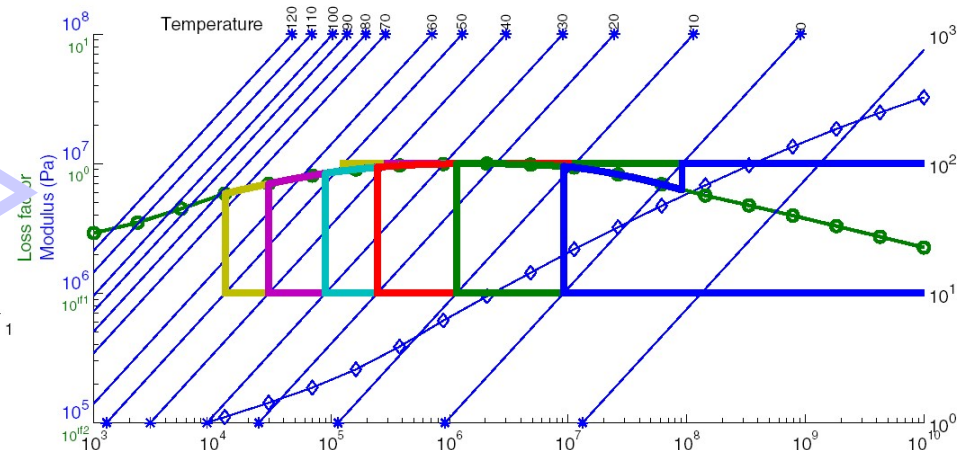
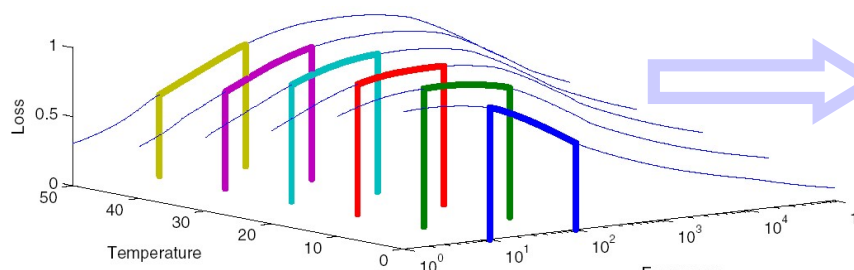
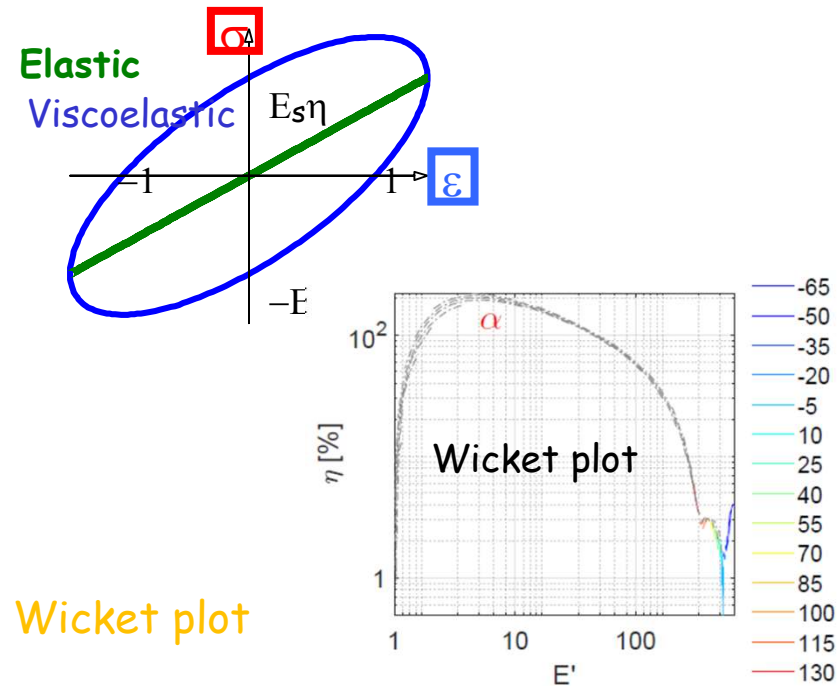
- Time convolution $\sigma(t) = \int_{-\infty}^t C(t - \tau)\epsilon(\tau)$
- Frequency complex modulus $\sigma = E(\omega, T, \epsilon_S)\epsilon$

$$E = E'(1 + i\eta)$$

E' storage modulus, $\eta = \tan(\delta)$ loss factor

- Temperature/frequency equivalence

$$E(\omega, T, \epsilon_S) = E(\omega\alpha(T, \epsilon_S)) \quad \text{generalized coordinate in Wicket plot}$$

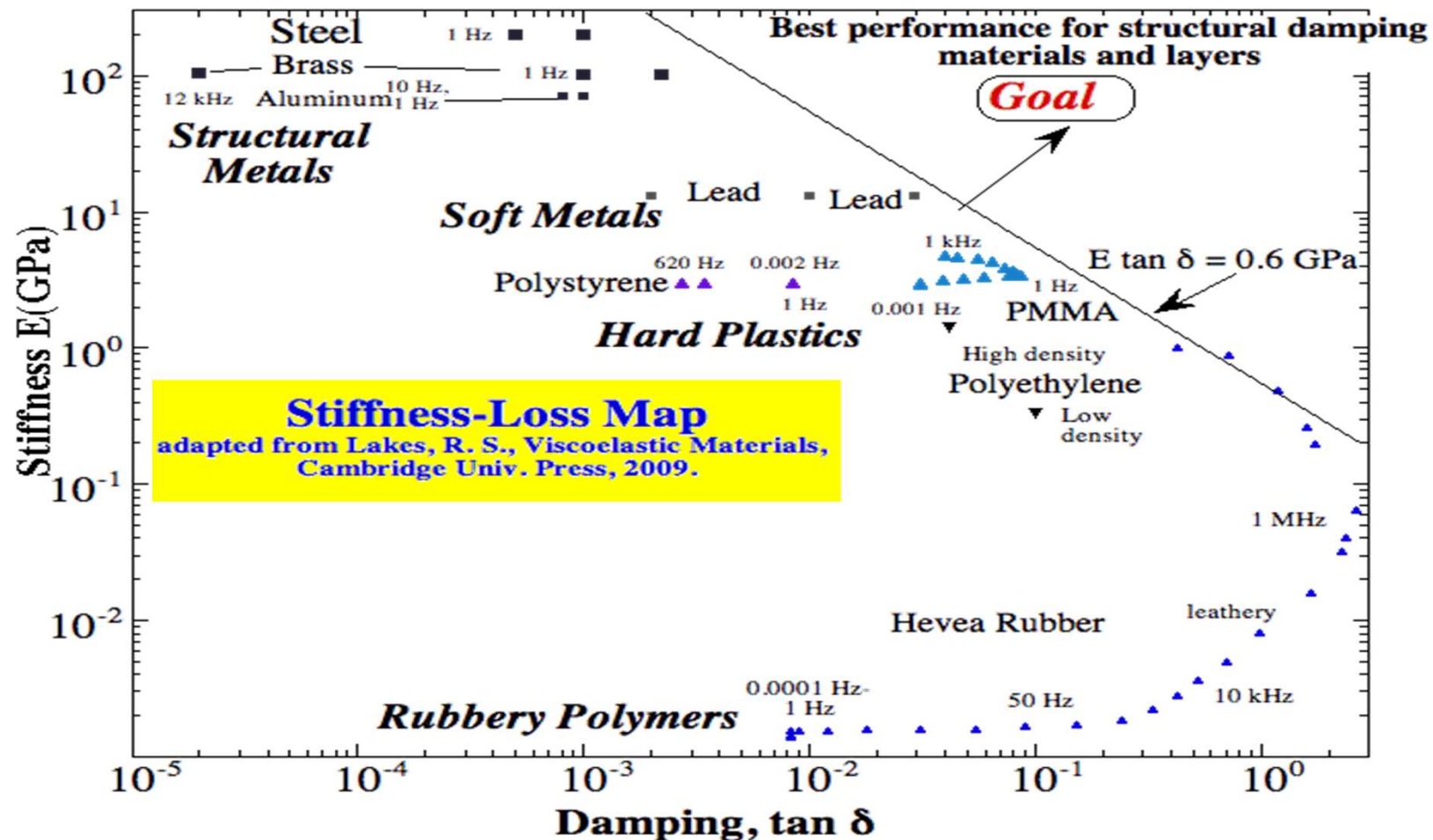


[1] A. D. Nashif, D. I. G. Jones, and J. P. Henderson, *Vibration Damping*. John Wiley and Sons, 1985.

[2] J. Salençon, *Viscoélasticité*. Presse des Ponts et Chaussées, Paris, 1983.

[3] SDT/Visc www.sdtools.com/pdf/visc.pdf

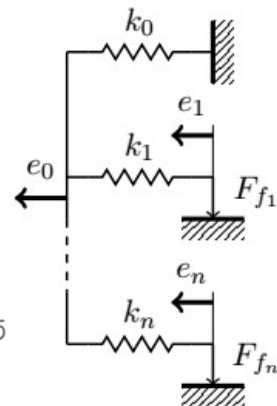
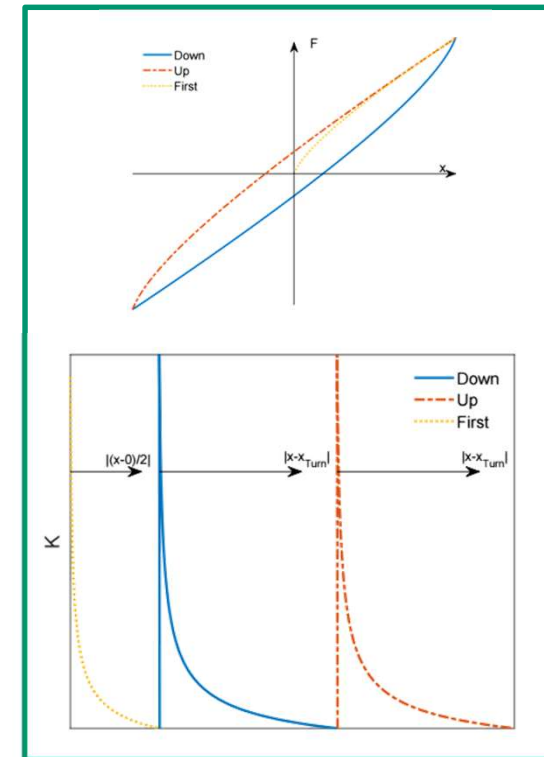
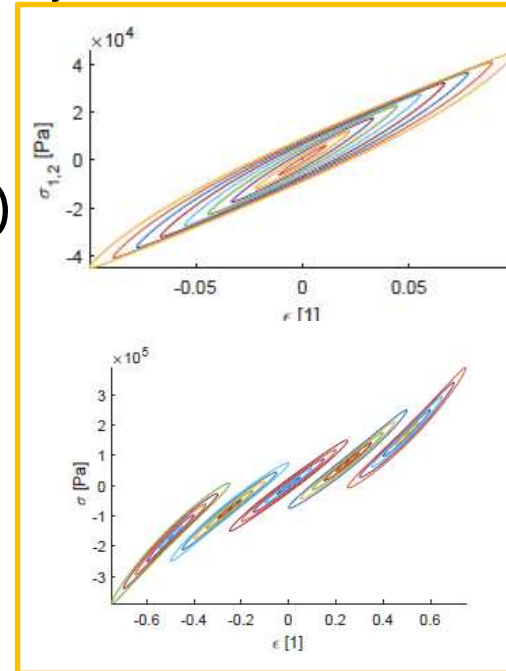
Modulus / loss map of common materials



$$\sigma(\omega) = E_{\text{complex}} \epsilon(\omega) = E_{\text{real}}(1 + i\eta(\omega))\epsilon = E_{\text{real}}(1 + i \tan(\delta)(\omega))\epsilon$$

Hysteretic damping : keywords

- Stress = fct (strain history)
but independent on rate
- Non parametric : triangular testing
 - Masing rule (homothety + offset)
 - Madelung assumption (loop closure)
 - Payne effect ($E \downarrow$ amplitude $\uparrow$)
 - Mullins (first loading adaptation)
- Parametric models :
Iwan, Dahl, Berg



- [1] D. J. Segalman, "A Four-Parameter Iwan Model for Lap-Type Joints," *Journal of Applied Mechanics*, vol. 72, no. 5, pp. 752-760, Sep. 2005
 [2] B. Bourgeteau, "Modélisation numérique des articulations en caoutchouc de la liaison au sol automobile en simulation multi-corps transitoire," Ecole Centrale Paris, 2009
 [3] R. Penas, "Models for dissipative bushings in multibody dynamics ", ENSAM, 2021

Rate independent hysteresis

- Hysteretic relaxation (**non-parametric**)

$$F(x) = F(x_{turn}) + \int_0^{|x-x_{turn}|} K_f(r) dr$$

- Selected order **parametric** representation
Jenkins cells, in an Iwan series arrangement
- Force formulation

$$F^i = \frac{g^i}{g^0} F^0, \text{ if } F^i \text{sign}(\dot{x}) < F_f^i, \text{ and } F^i = \text{sign}(\dot{x}) F_f^i, \text{ if } \frac{g^i}{g^0} |F^0| \geq F_f^i$$

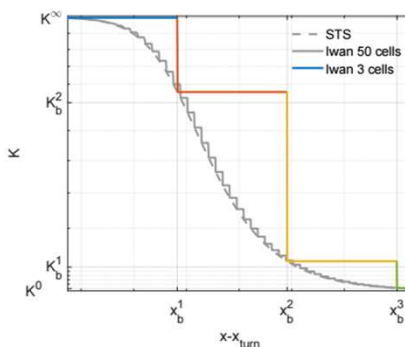
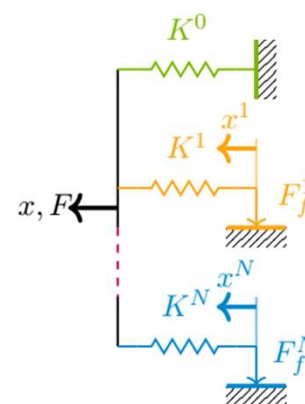
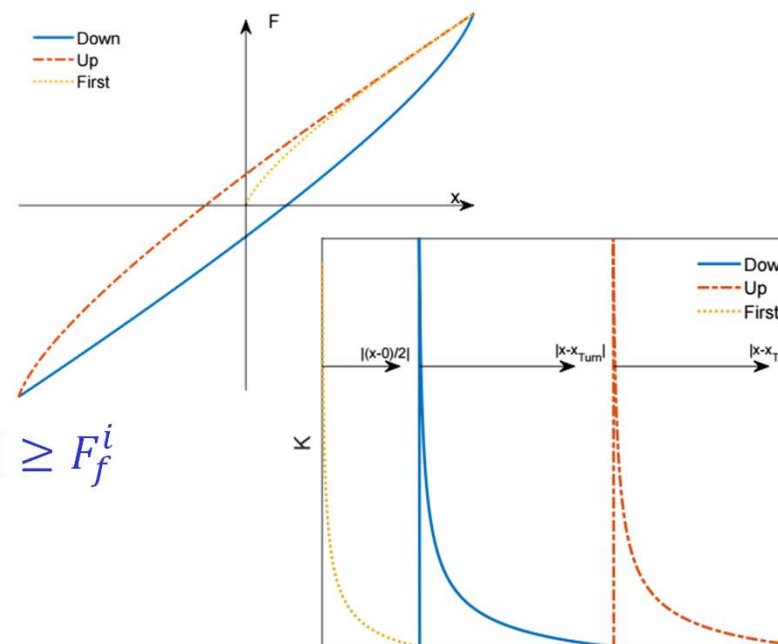
- Displacement formulation

$$\dot{x}^i = 0, \text{ if } |x - x^i| < \frac{F_f^i}{K^i}, \text{ and } \dot{x}^i = \dot{x}, \text{ if } |x - x^i| > \frac{F_f^i}{K^i}$$

- Definition of break points and stiffness

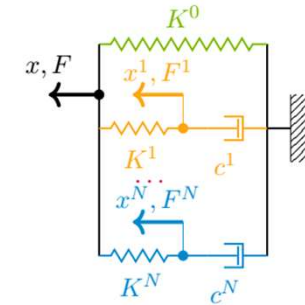
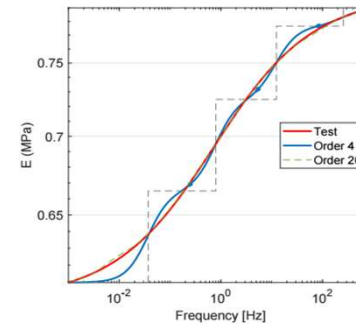
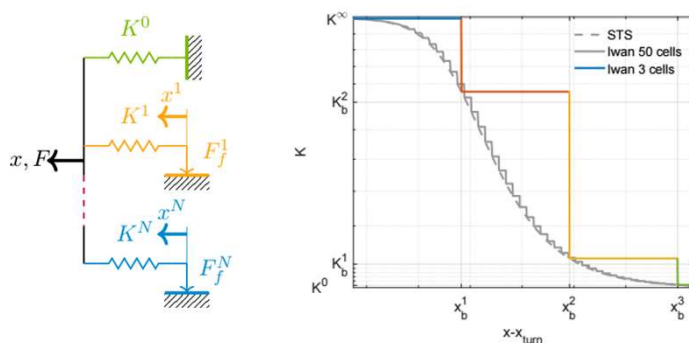
$$K_b^k = \sum_{i < k} K^i, \text{ and } x_b^k = x_{turn} + \frac{F_f^k}{K^k}$$

- Order selection: place x_b and K_b in the hysteretic relaxation curve



Parallel between viscoelasticity & hysteresis

- Fit non-parametric curves : discretize with selected order



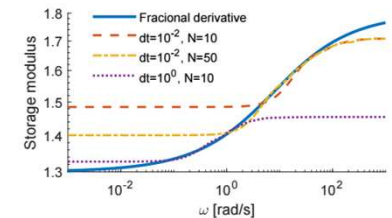
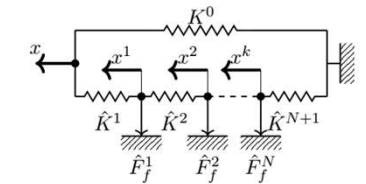
- Identify order independent parameters

- STS
$$F - F_{turn} = \left(\sqrt{NC|x - x_{turn}| + \frac{NC}{2\hat{K}^0} - \frac{NC}{2\hat{K}^1}} \right) \text{sign}(x - x_{turn})$$

- Segalman (4 parameter friction)

- Fractional derivative

$$F(s) + \frac{1}{\omega_c} \frac{d^\alpha F}{dt^\alpha} = K^0 x + \frac{1}{\omega_c^\alpha} K^\infty \frac{d^\alpha x}{dt^\alpha}$$

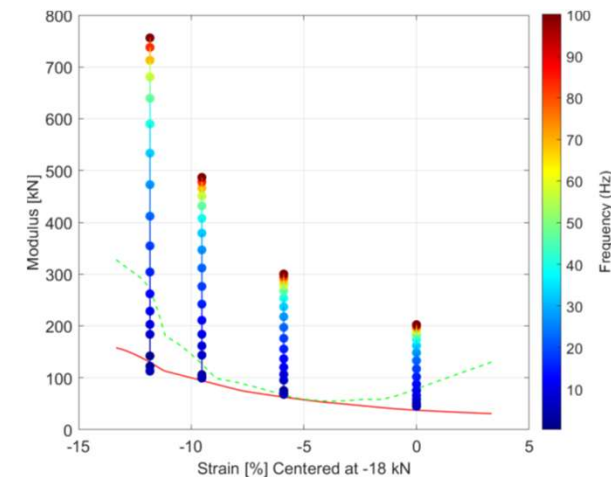
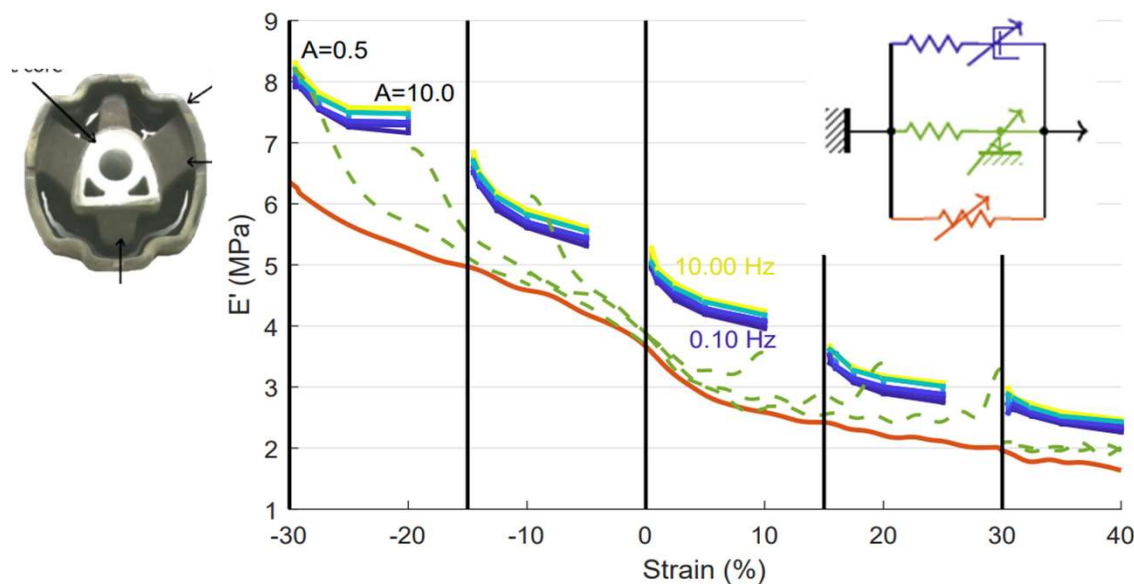


- Implementation requires discretization = order selection

Adding hyperelasticity & combination

Material/mount model split in **static**, **path** and **dynamic** effects

- Non-parametric models for: **hyperelasticity**, **hysteresis** and **viscoelasticity** $E(\omega, A, \epsilon_0)$

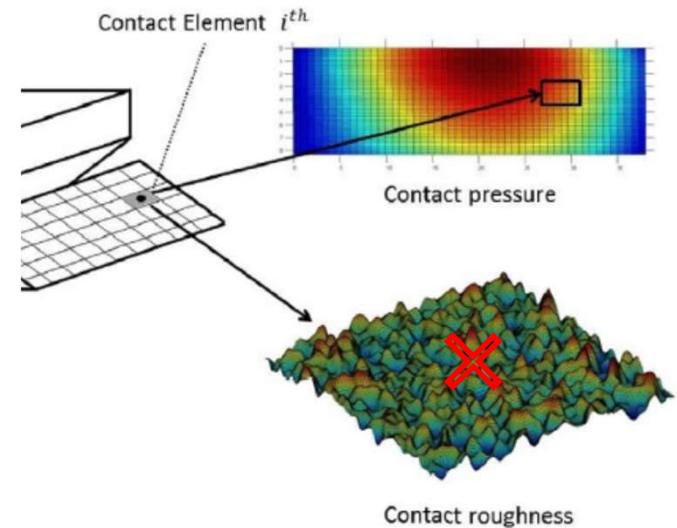
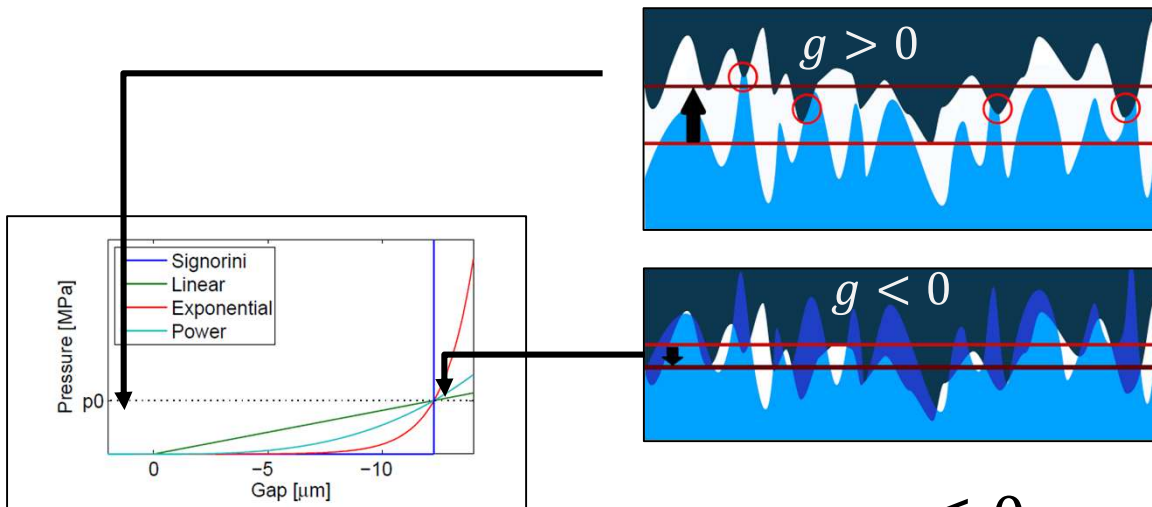


- Among all cell models that combine **HE**, **HYST** and **VE**
 - stress rate relaxation $\dot{F}^0(x)$ is needed
 - NL relaxation times $\omega^i + \frac{|\dot{x}|}{x_f^i}$ explain Paine effect

$$\dot{F}^i + \left(\omega^i + \frac{|\dot{x}|}{x_f^i} \right) F^i = \frac{g^i}{g_0} \dot{F}^0(x)$$

Contact constitutive law + friction

- Macro-scale surface not flat (**1 gauss**)
- Macro load function of gauss strain

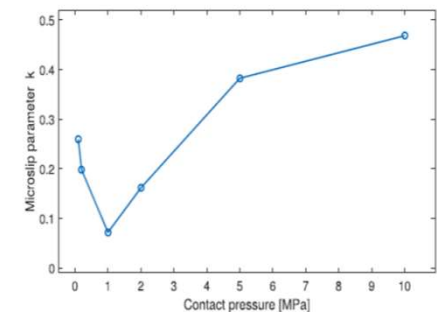
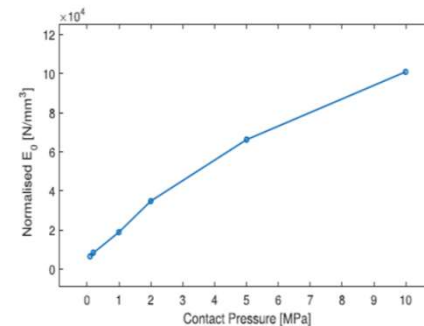


From : L. Pesaresi. JSV 2018

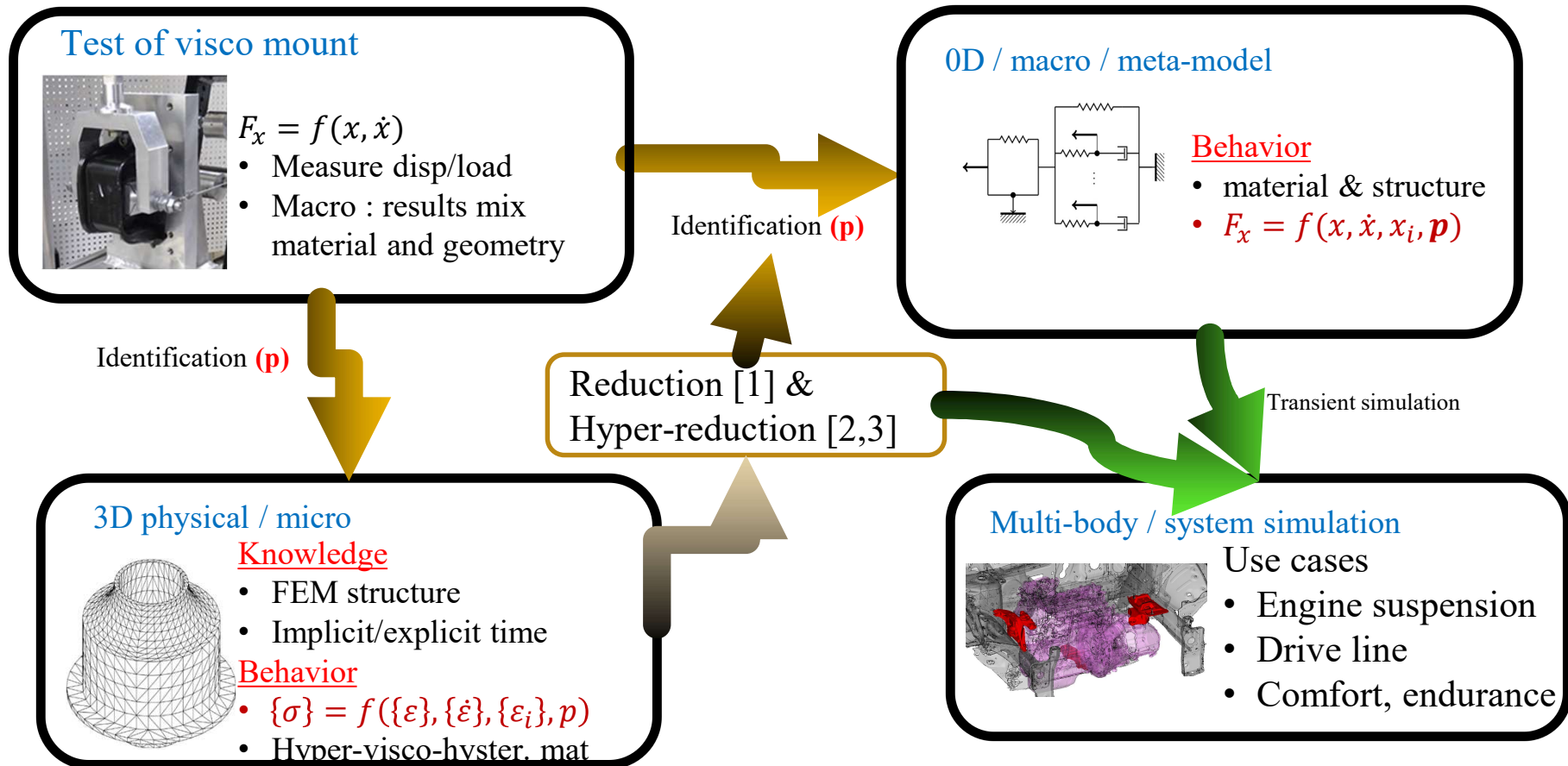
- Idealization Signorini
- Reality $p(g)$

$$\begin{aligned} g &\leq 0 \\ p &> 0 \\ pg &= 0 \end{aligned}$$

- Friction : coulomb $\sigma_t = -\mu p \frac{v_t}{\|v_t\|}$



Q4 : from material to structure model



- [1] Hammami, Balmes, Guskov, « Numerical design and test on an assembled structure of a bolted joint with viscoelastic damping », *MSSP*, v70, p.714–724, 2015
- [2] Farhat, Avery, Chapman, Cortial, « Dimensional reduction of nonlinear finite element dynamic models with finite rotations and energy-based mesh sampling and weighting for computational efficiency », *IJNME*, vol. 98, n° 9, p. 625–662, 2014.
- [3] F. Casenave, N. Akkari, F. Bordeu, C. Rey, and D. Ryckelynck, “A nonintrusive distributed reduced-order modeling framework for nonlinear structural mechanics—Application to elastoviscoplastic computations,” *IJNME*, vol. 121, no. 1, pp. 32–53, Jan. 2020, doi: 10.1002/nme.6187.

Historical method : V_0 = reduction modes (poor)

MSE (modal strain energy) is a kinematic reduction using **elastic modes** and **general damping**

$$\{q\} = [T] \{q_r\} = [\phi_1 \dots \phi_{NM}] \{q_r\}$$

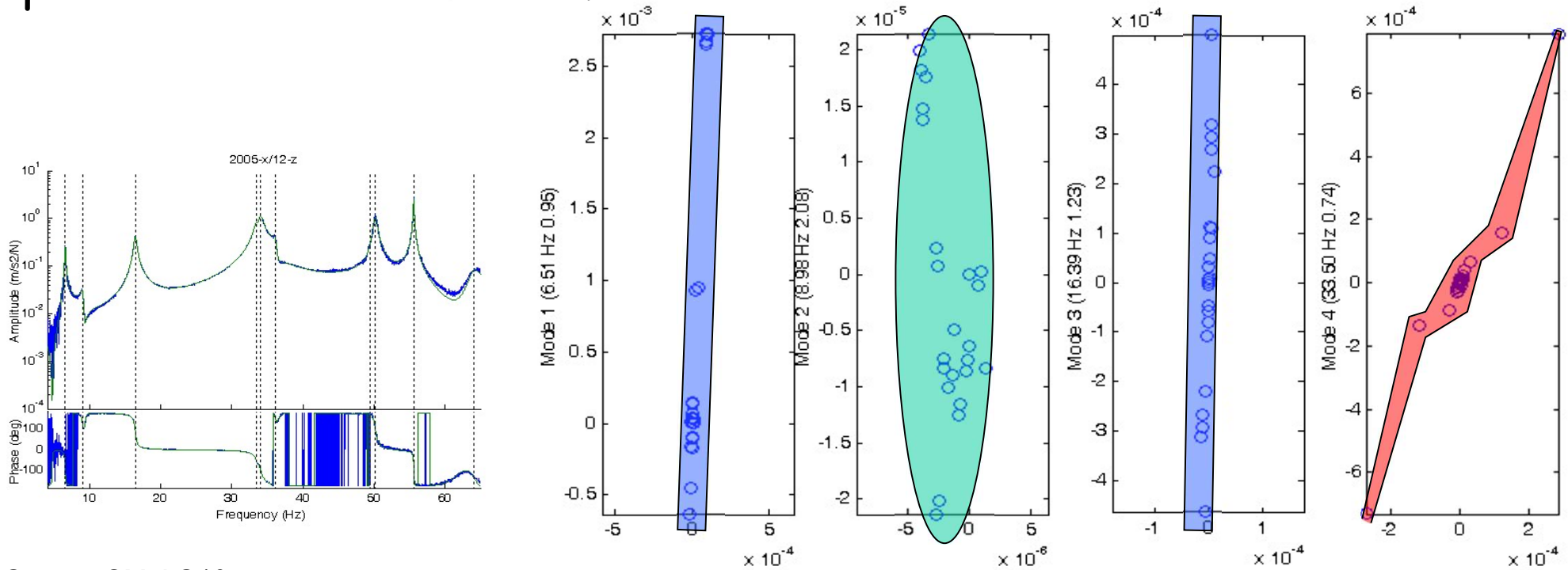
$$\left[s^2 [I] + [\phi]^T [Im(Z(s))] [\phi] + [\omega_j^2] \right] \{p\}(s) = [\phi]^T \{F(s)\}$$

NASTRAN

- SOL110 modal complex modes
- SOL111 modal frequency response
- SOL112 modal transient

When does MSE work ?

Experimental modes are often “real”



Garteur SM-AG19 test

“Real modes”

Residues for I/O pairs line up

“Complex modes”

Residues have a phase spread

Poor modes

Have complex residues

When are complex modes nearly real ?

- Uncoupling criterion (Hasselman) $\Leftrightarrow$

$$\min(\zeta_1\omega_1, \zeta_2\omega_2)/|\omega_1 - \omega_2| \ll 1$$

corresponds to non overlap of peaks : using real modes is then OK

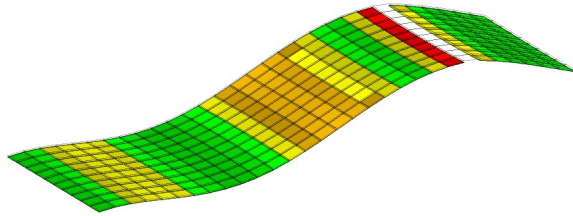
- Proof based on damping matrix positiveness

$$\gamma_{12}\gamma_{21}/(\gamma_{11}\gamma_{22}) < 1$$

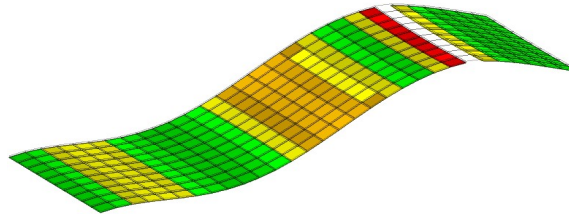
- When damping is enhanced by design, modes are often complex

Historical V_1 = reduction modes + enrichment (good)

Modal

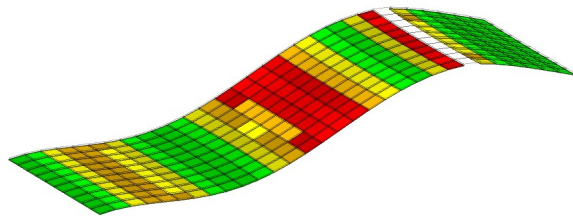


MSE

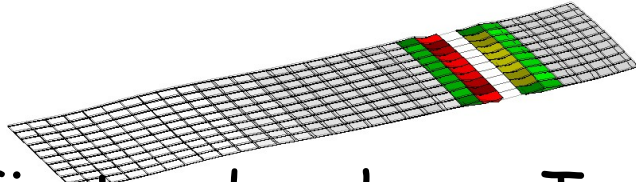
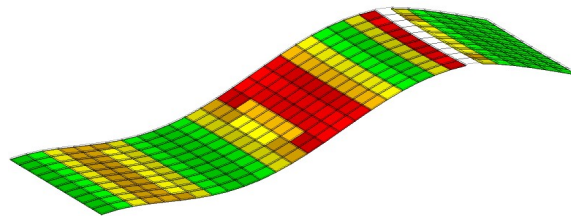


- Forced response around a resonance
- Objective correct strain energy

First order



Exact



First order shape : $T = [K_0^{-1} [Im(Z-Z_0)] \phi_4]_{orth}$

- Modes (MSE)
- Multi-model
- Modes + static

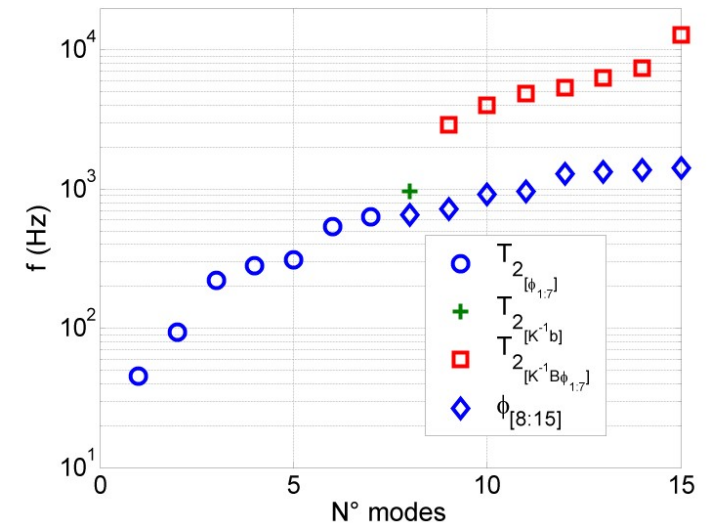
$$T = [\Phi(p_0)]$$

$$T = [\Phi(p_1) \Phi(p_2)]$$

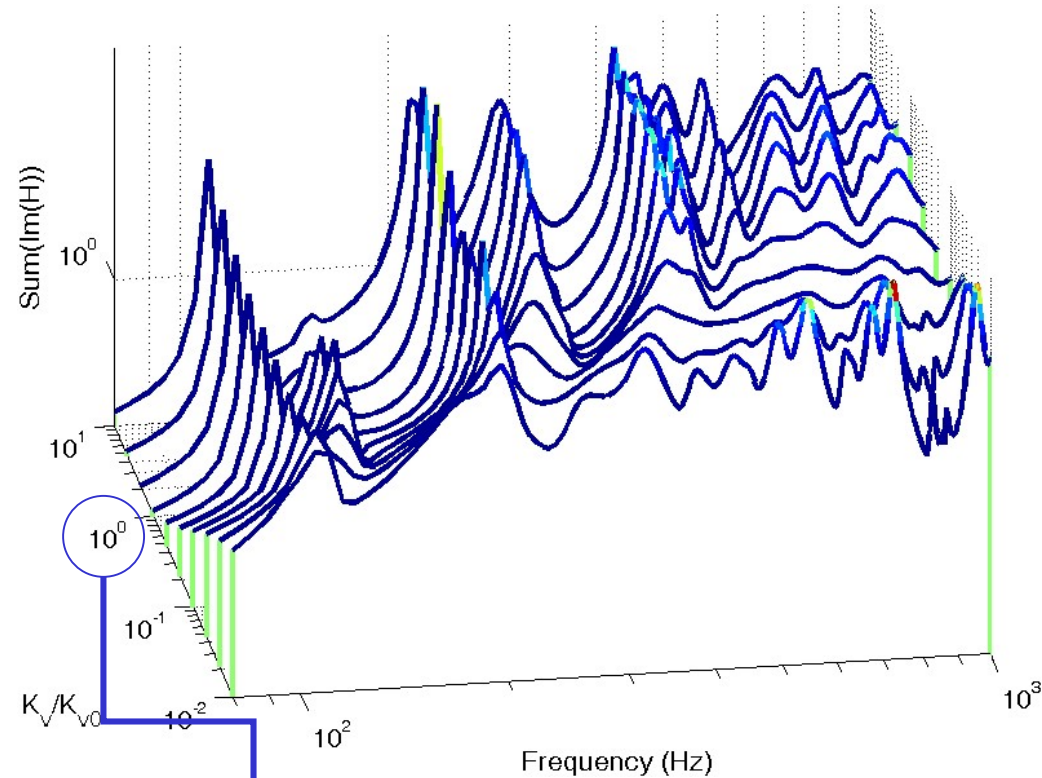
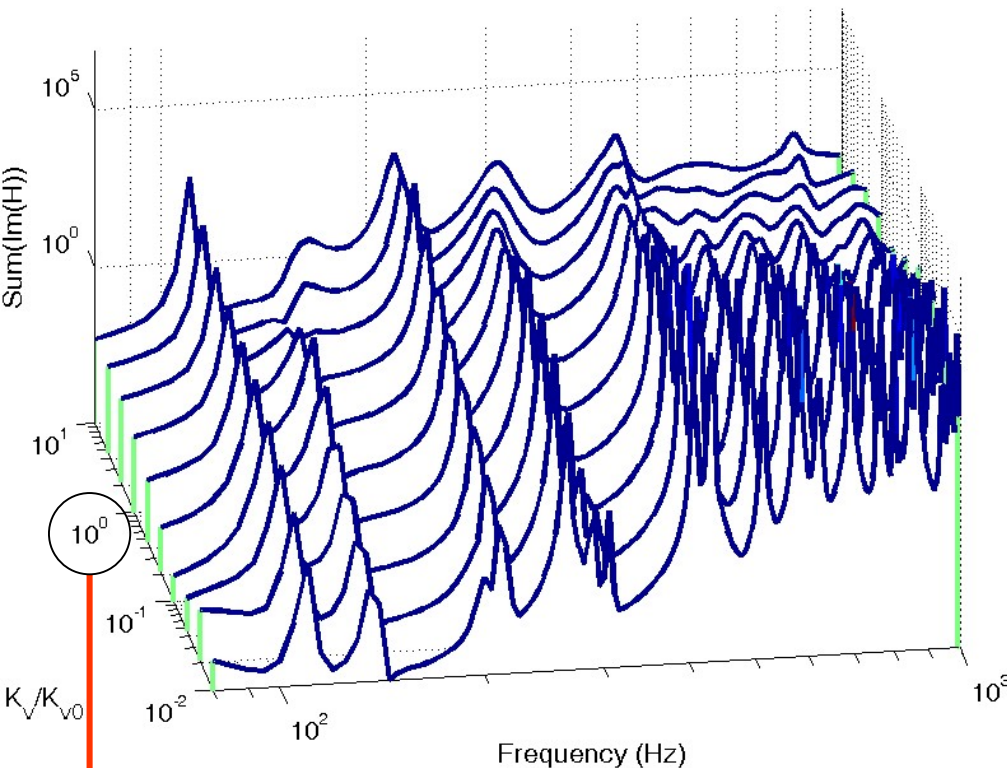
$$T = [\Phi(p_0) K_0^{-1} [Im(Z-Z_0)] \Phi(p_0)]$$

Reduction Bases/Starting Point

- Modal $T = [\Phi(p_0)]$
- Modal + static Responses to input
 $T = [\Phi(p_0) \ K^{-1}b]$
- Modal + static responses
 to representative loads
 $T = [\Phi(p_0) \ K^{-1}[b \ R_j] \ \Phi(p_k)]$
- Frequencies of corrections $\square$
 differ from modes $\blacklozenge$: the added
 information is VERY different



MSE validity FRF prediction



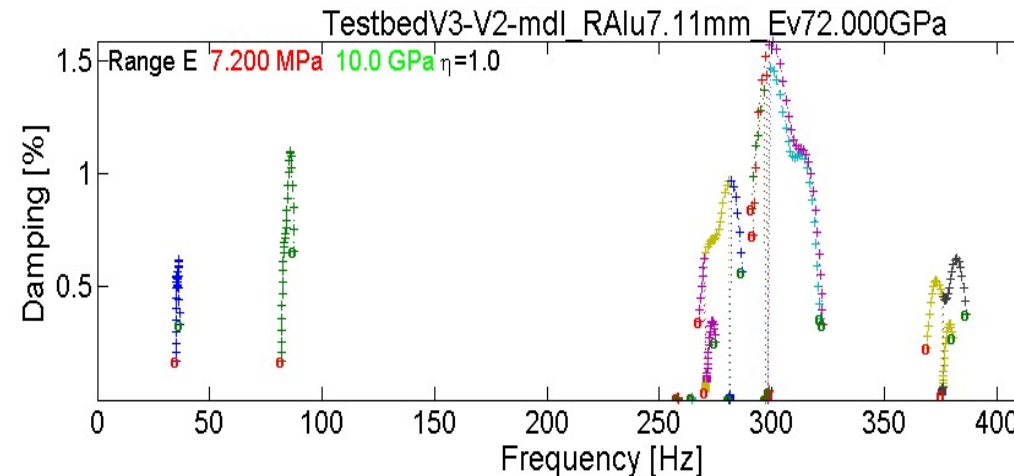
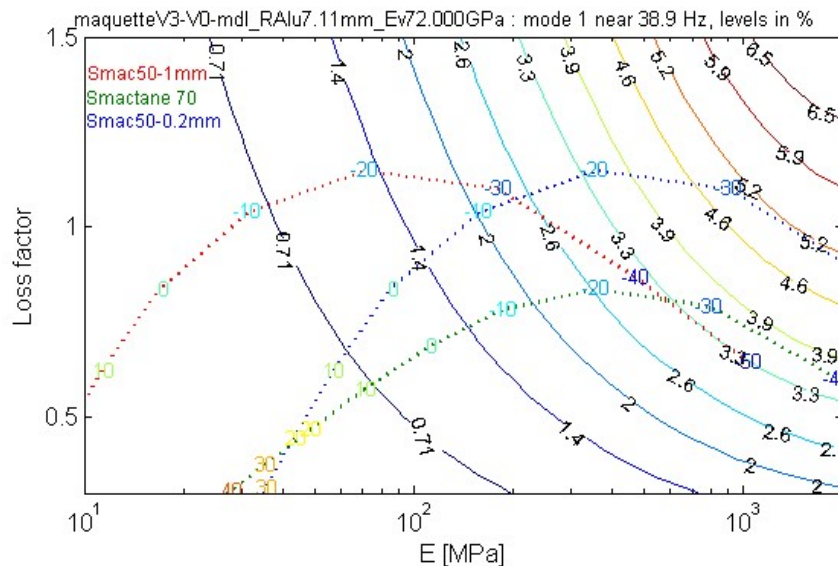
$$T = \boxed{\phi(k_v)_{1,NM}} K_0^{-1} K_v \phi(k_v)_{1,NM}$$

Fixed nominal modes give a poor approximation



Eigenvalue extraction

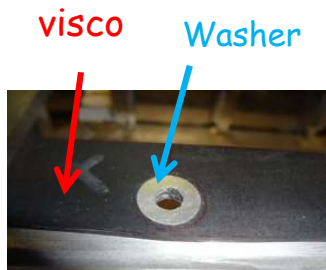
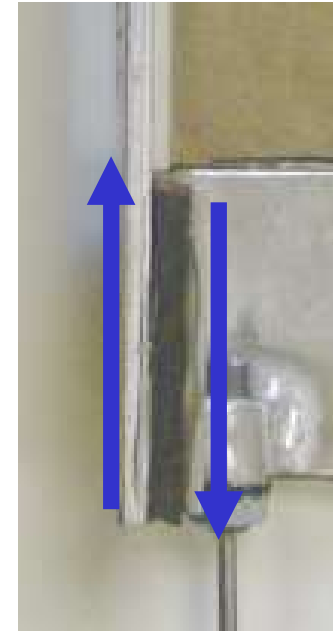
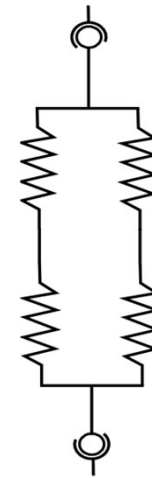
- Fixed E sparse solvers are **not robust and slow**
 $\Rightarrow$ **reduction T + full solver** $[Z(E_i, \lambda_j)]\{\psi_j\} = \{0\}$
- Analytic E(s) not common + associated sparse eigenvalue not available $\Rightarrow$ **only used for transient**
- Need to track $\{\psi_j\}, \omega_j, \zeta_j = f(E(T, \omega_j), \eta(T, \omega_j))$
 $\Rightarrow$ identify from forced response
 $\Rightarrow$ interpolate from $f(E, \eta)$



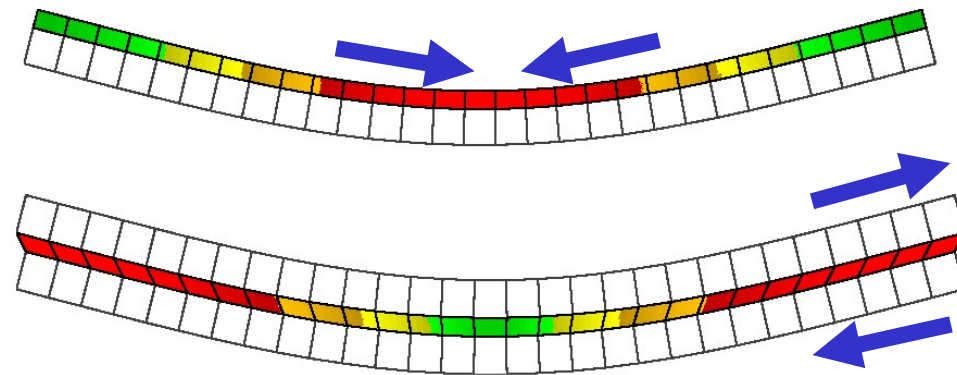
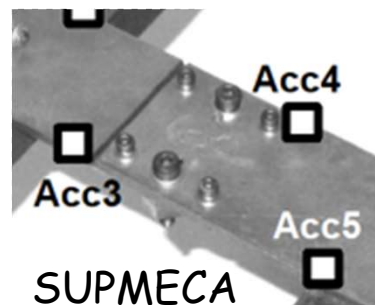
Structural mechanisms

- Compression & shear springs
 - Integration gives **orders of magnitude choice** in functional stiffness
 - Possibly structured junctions
- Free layer bending (compression)
 - Inefficient if damping material is too soft
- Constrained layer bending (shear)
- Trough thickness dynamics
 - Only efficient in resonant (high frequency behavior)

$$k_z(\omega) = G(\omega) (1 + i\eta(\omega)) \frac{S}{\tau}$$



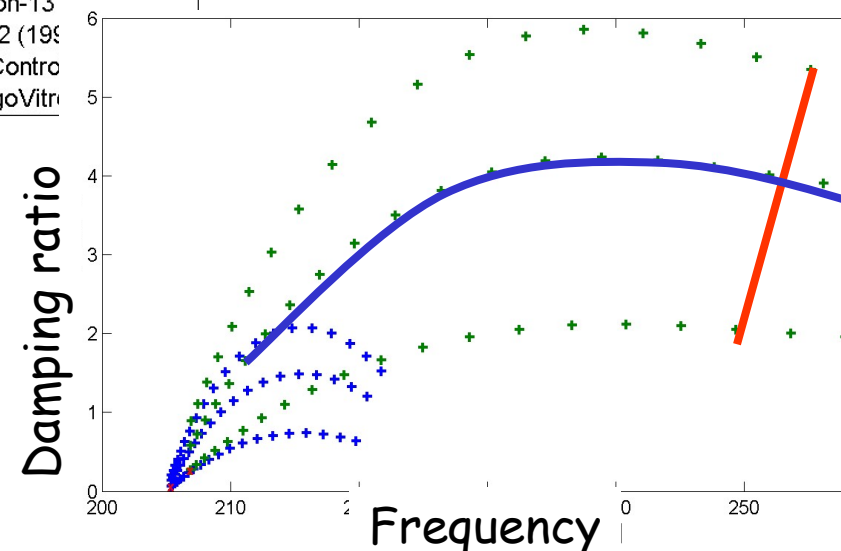
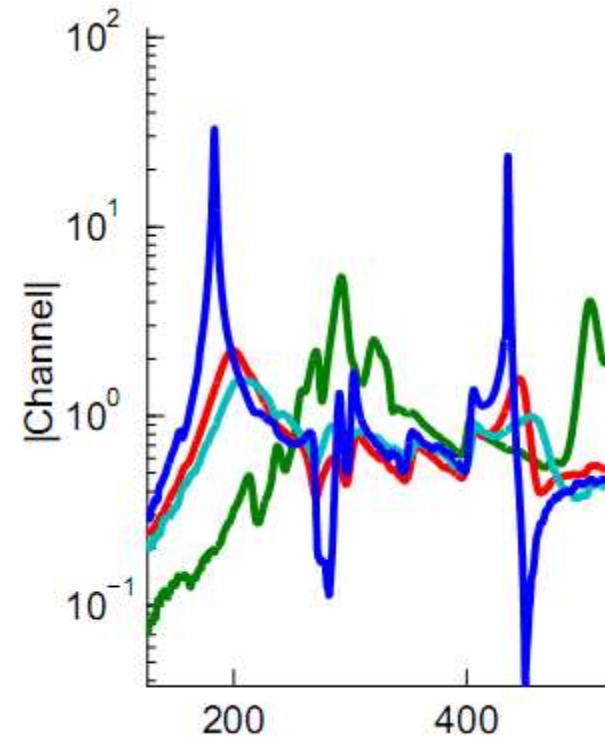
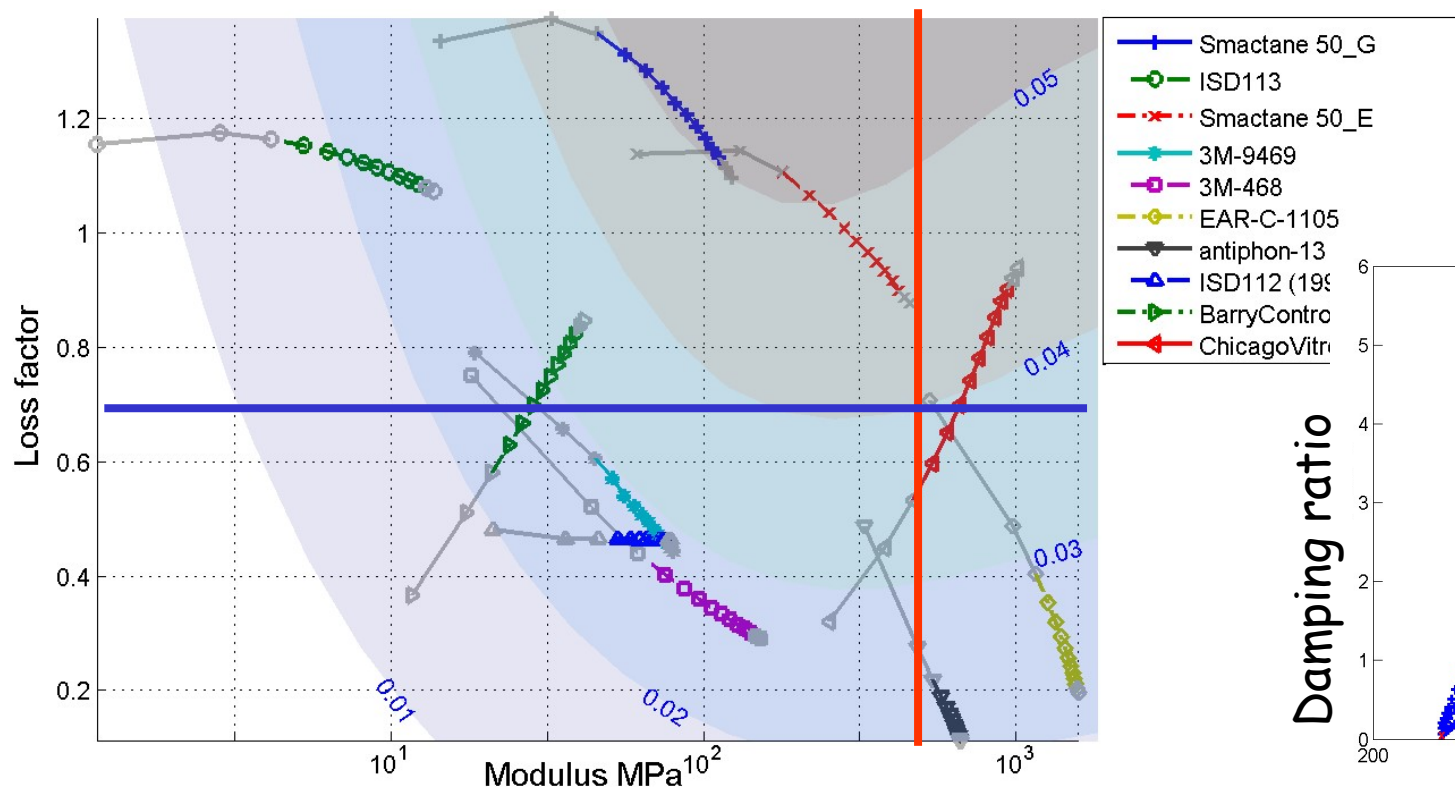
ENSAM



Optimization is necessary

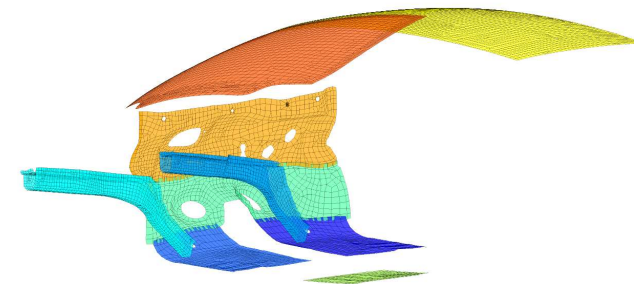
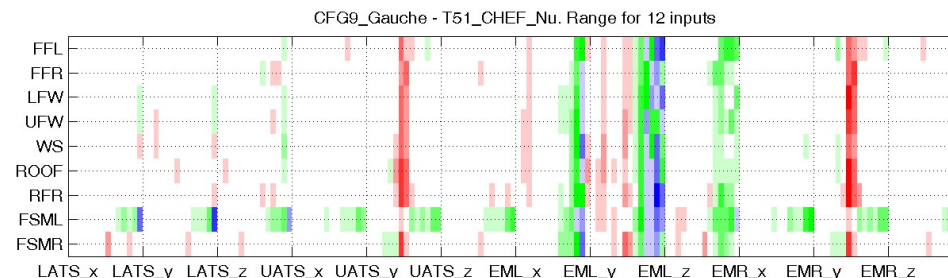
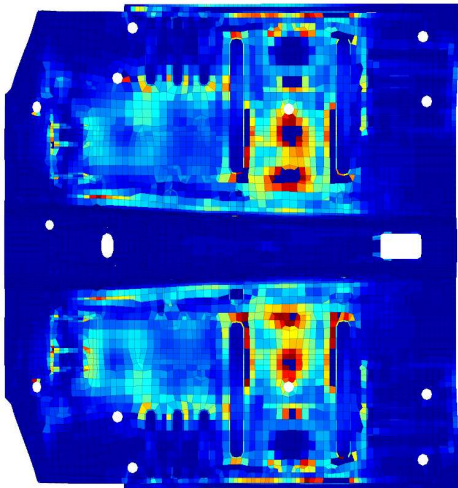
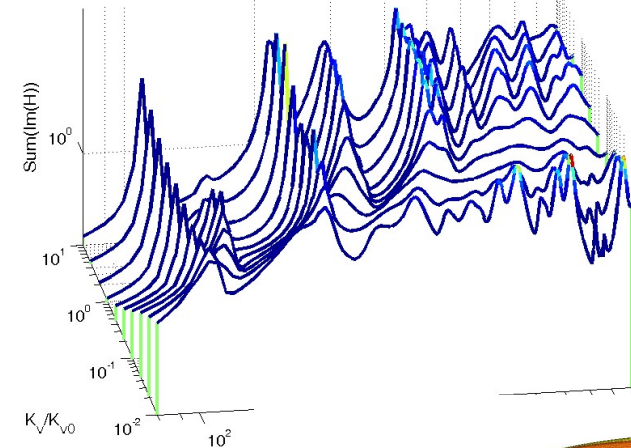
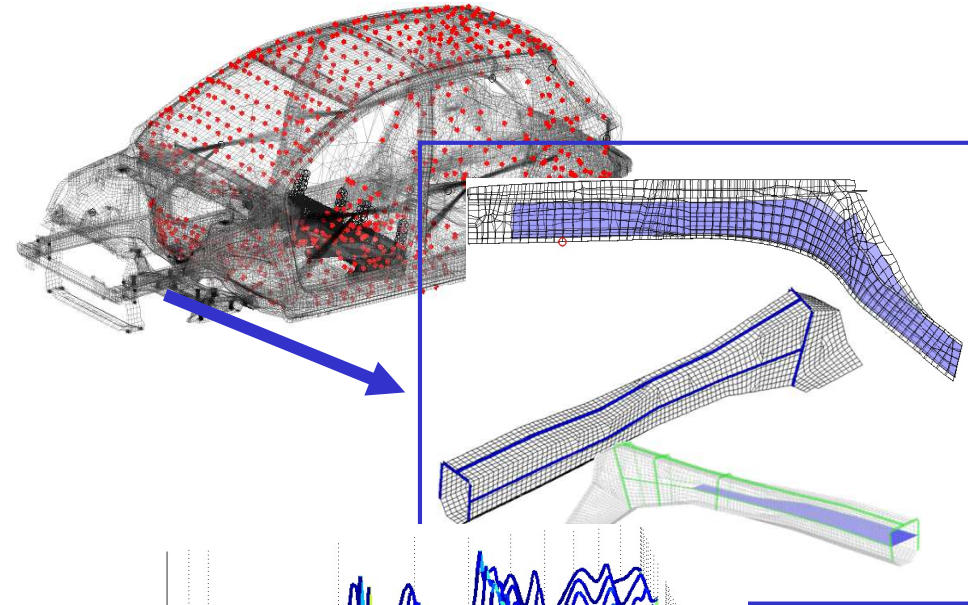
- Design space : material, thickness, surface (GS/h or EA/h)
- Multiple orders of magnitude

Damping ratio target example



Why optimization is a challenge

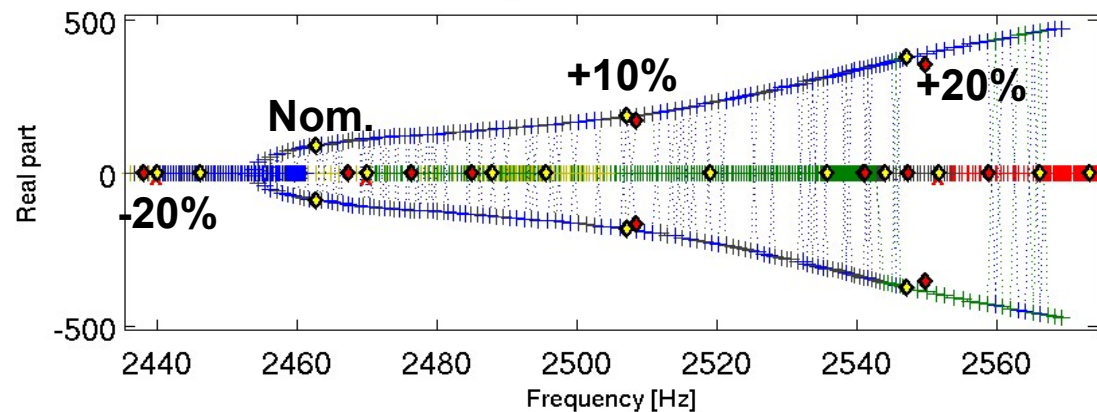
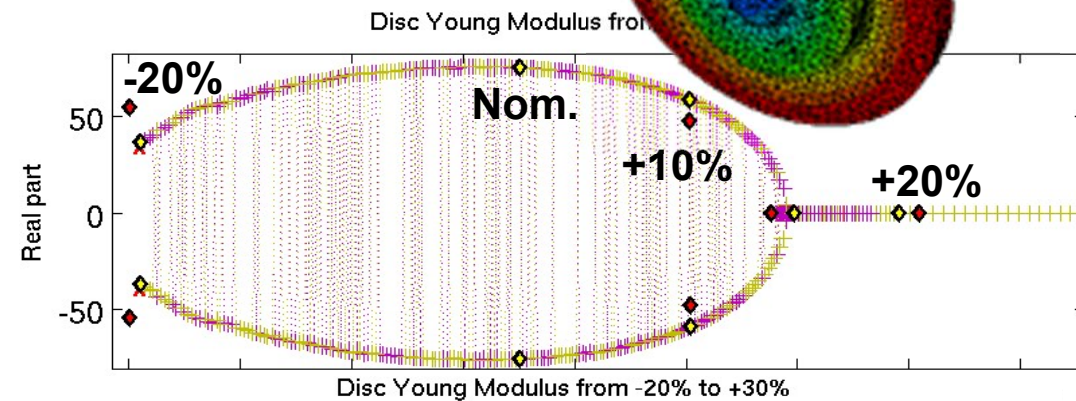
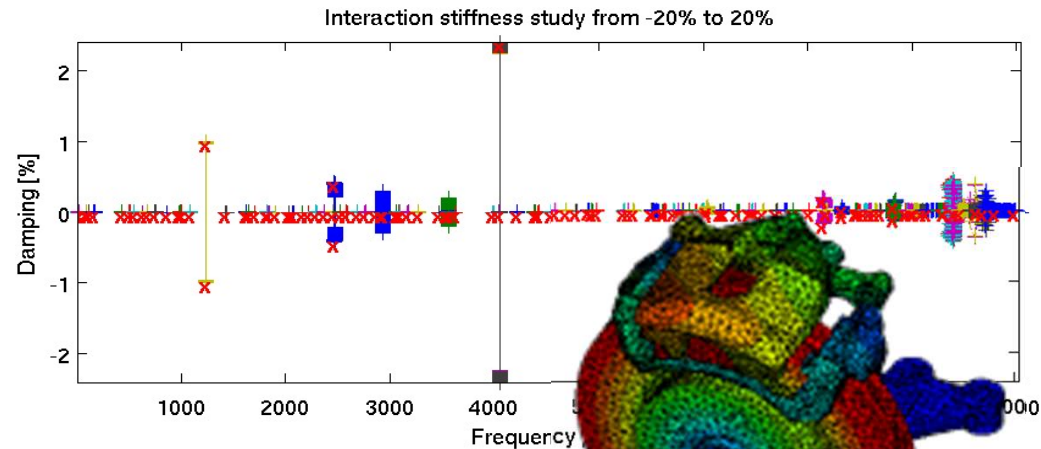
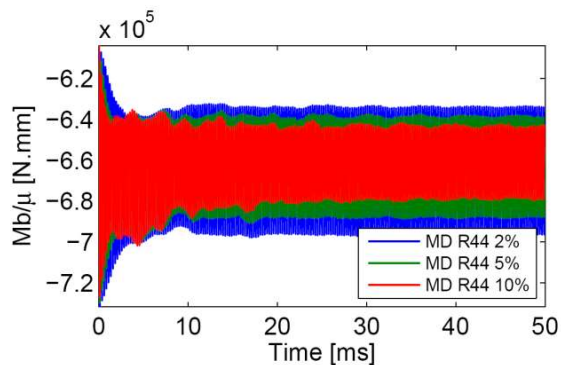
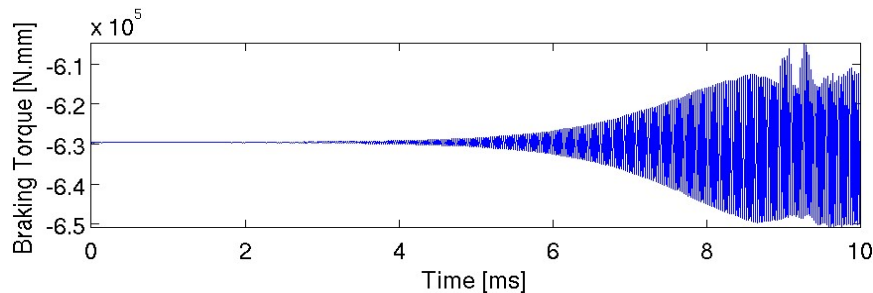
- Large models ($\times 10^6$ DOF)
- Many modes / frequencies to compute ($\times 10^3$)
- Multiple designs ($\times 10$)
- Large parametric variations (10^4)
- Complex objectives ($\times 10^3$ disp)



Historical V_2 = multi-model & CMS

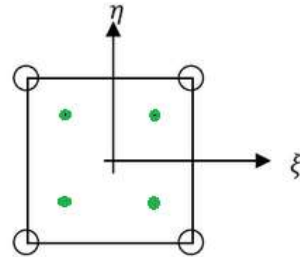
Objectives:

- complex mode stability
- Transient simulation



Historical V_3 = hyper reduction of transient NL

HR phase 1 : generic form of FEM problems



$$1. \text{ FEM kinematics } u(x, t) = [N(x)]\{q(t)\}$$

$$\epsilon = [\mathbb{C}]_{(N_{\epsilon} \times N_g) \times N} \{q\}$$

$$2. \text{ Material evolution : } S_g = f_{mat}(\epsilon_g, u_{int\ g})$$

$$3. \text{ Virtual work } M\ddot{q} + \mathcal{F}_{int}(S, \mathbf{u}_{int}) = \mathcal{F}_{ext}$$

• Integration at quadrature points g (disassembly)

$$\mathcal{F}_{int}(q, u_{int}) = \sum_g \mathbb{C}^T J_g w_g f_{mat}(\epsilon, \mathbf{u}_{int})$$

$$= [\mathbb{B}]_{N \times (N_g \times N_f)} \left\{ f_{mat} \left([\mathbb{C}]_{(N_{\epsilon} \times N_g) \times N} \{q\}, \mathbf{u}_{int} \right) \right\}$$

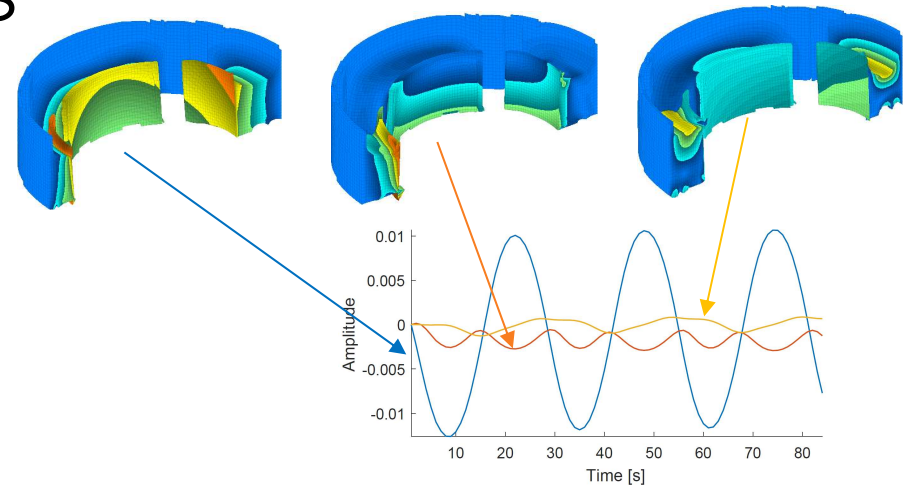
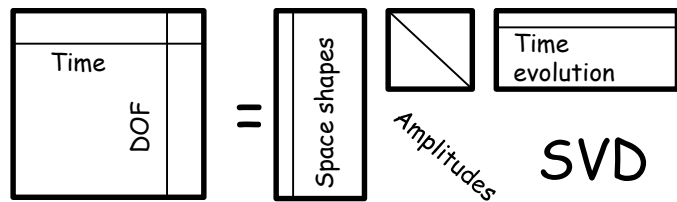
HR phase 2 : kinematic reduction

1/ **Off-line learning** (non-linear High Fidelity Simulation)

Snapshots or iterative method

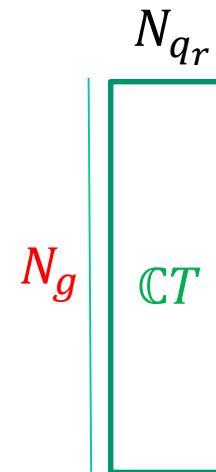
2/ **Kinematic reduction** using SVD, or CMS

$$q_{learn} = \sum_i \{U_i(x)\} (\sigma_i v_i(t))$$



Reduced equations of motion $T^T M T \ddot{q}_r + [T^T \mathbb{B}]_{N_{qr} \times N_g} f_{mat}(\mathbb{C} T q_r, u_{int}) = T^T F_{ext}$

- N_g (gauss points) remains large
- $f_{mat}(\mathbb{C} T q_r, u_{int})$ evaluation dominates



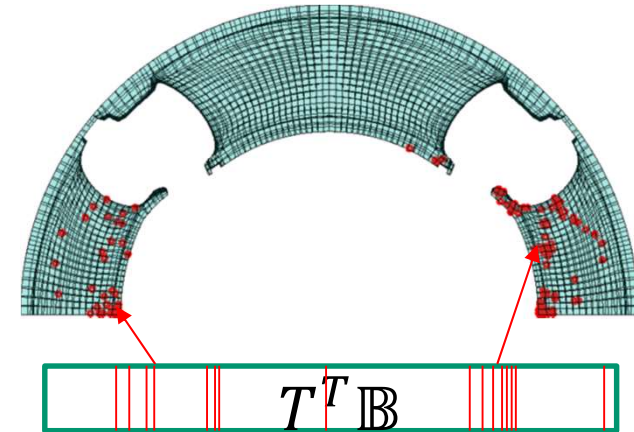
HR phase 3 : operator reduction

3/ Operator reduction (hyper-red.) = less points + new weights

$$\{F_{int}\} = \int_{\Omega} f_{mat}(x_g, t) dV \approx \sum_g \mathbb{C}^T J_g w_g f_{mat g} = \mathbb{B}(x_g) f_{mat g}(q, t)$$

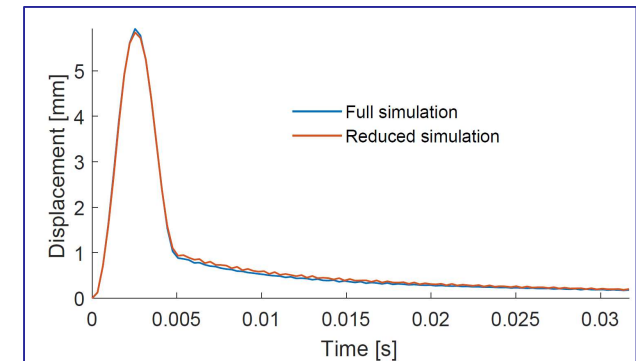
Choose NT learning points and minimize $\|w_g^*\|_0$ subject to^[1]

$$\left\| \left[[T^T \mathbb{B}] S_l \right]_{(NT \times NR)} - \left[[T^T \mathbb{B}_g] S_{lg} \right]_{(NT \times NR) \times Ng} \{w_g^*\} \right\|_2 < \varepsilon_{tol} \text{ and } w_g^* > 0$$



4/ On-line usage (29s vs. 27h)

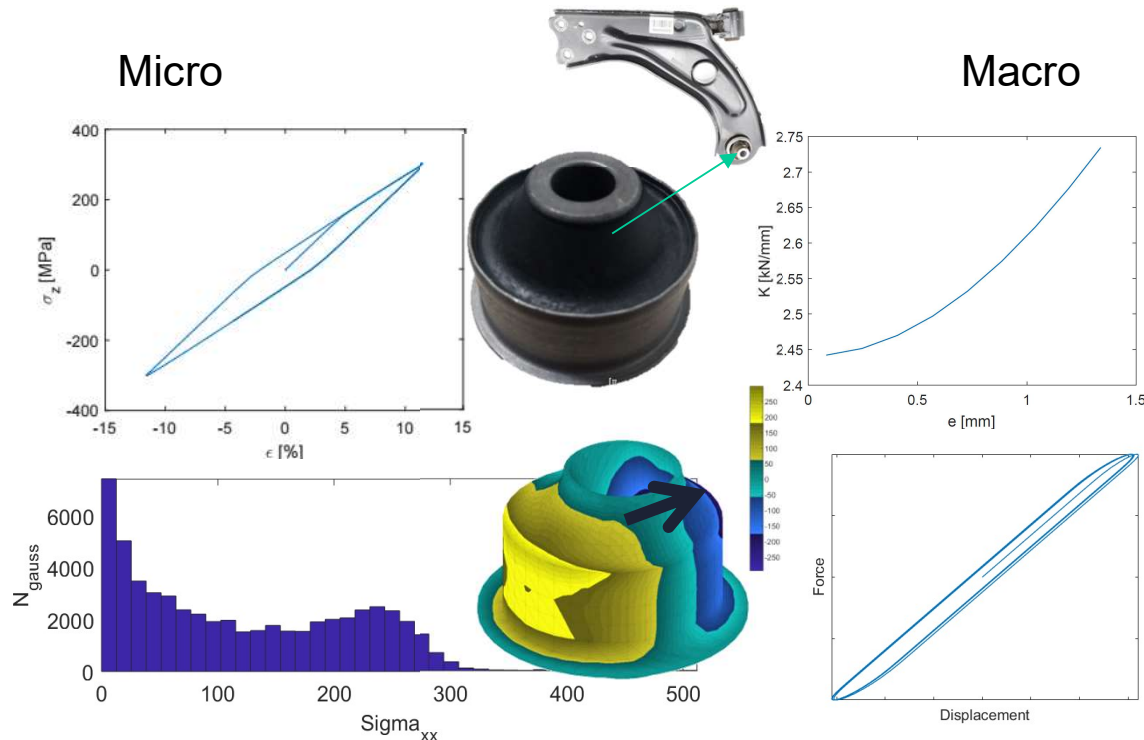
5/ Post-processing (estimate continuous fields)



- [1] Farhat, Avery, Chapman, Cortial, « Dimensional reduction of nonlinear finite element dynamic models with finite rotations and energy-based mesh sampling and weighting for computational efficiency», *IJNME*, vol. 98, n° 9, p. 625-662, 2014.
- [2] F. Casenave, N. Akkari, F. Bordeu, C. Rey, and D. Ryckelynck, “A nonintrusive distributed reduced-order modeling framework for nonlinear structural mechanics—Application to elastoviscoplastic computations,” *IJNME*, vol. 121, no. 1, pp. 32–53, Jan. 2020, doi: 10.1002/nme.6187
- [3] Penas, Rafael, *Models of dissipative bushing in multibody dynamics*, PhD ENSAM 2021

System = macro / micro

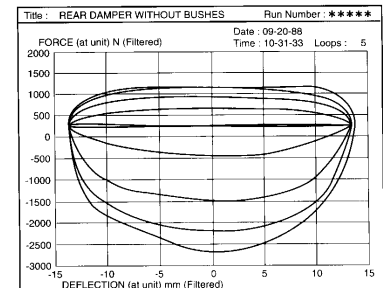
- Field distribution : macroscopic law not a simple function of material behavior



- Resultant (average in space)
- Loss factor (average in time & space)

$$\eta(X, T) = \frac{\frac{1}{2\pi} \int_0^T \sigma(X, t) \dot{\epsilon}(X, t) dt}{\frac{1}{2} \max_0^T (\sigma(X, t) \epsilon(X, t))}$$

System/macro
force-state map
 $F = f(q, t)$
Meta-model



Internal strains or internal displacement

Viscoelastic stress field

1. FEM kinematics

2. Material evolution : $S = f_{mat}(u_{nl}, u_{int})$

3. Virtual work $M\ddot{q} + \mathcal{F}_{int}(q, \mathbf{S}) = \mathcal{F}_{ext}$
 $\mathcal{F}_{int}(q, u_{int}) = [\mathbb{B}]_{N \times (N_g \times N_f)} \{f_{mat}(S)\}$

$$E(s) = E_\infty - \left(\sum_{j=1}^n \frac{E_j}{s + \omega_j} \right)$$

$$q_{vj} = -\frac{E_j}{(s + \omega_j)} q$$

Viscoelastic parametric matrix $\mathcal{F}_{int}(q, u_{int}) = \left(E^\infty + \sum_j \frac{E_j}{s + \omega_j} \right) K^u \{q\}$

Augmented state-space

$$E(s) = E_\infty - \left(\sum_{j=1}^n \frac{E_j}{s + \omega_j} \right) \quad \Rightarrow \quad q_{vj} = -\frac{E_j}{(s + \omega_j)} q$$

$$\left[\begin{bmatrix} M & 0 & 0 \\ 0 & M & 0 \\ 0 & 0 & M \end{bmatrix} s + \begin{bmatrix} 0 & -M & 0 \\ K_e + E_\infty K_v & 0 & K_v \\ E_j M & 0 & \omega_j M \end{bmatrix} \right] \begin{Bmatrix} q \\ sq \\ q_v \end{Bmatrix} = \begin{Bmatrix} 0 \\ F \\ 0 \end{Bmatrix}$$