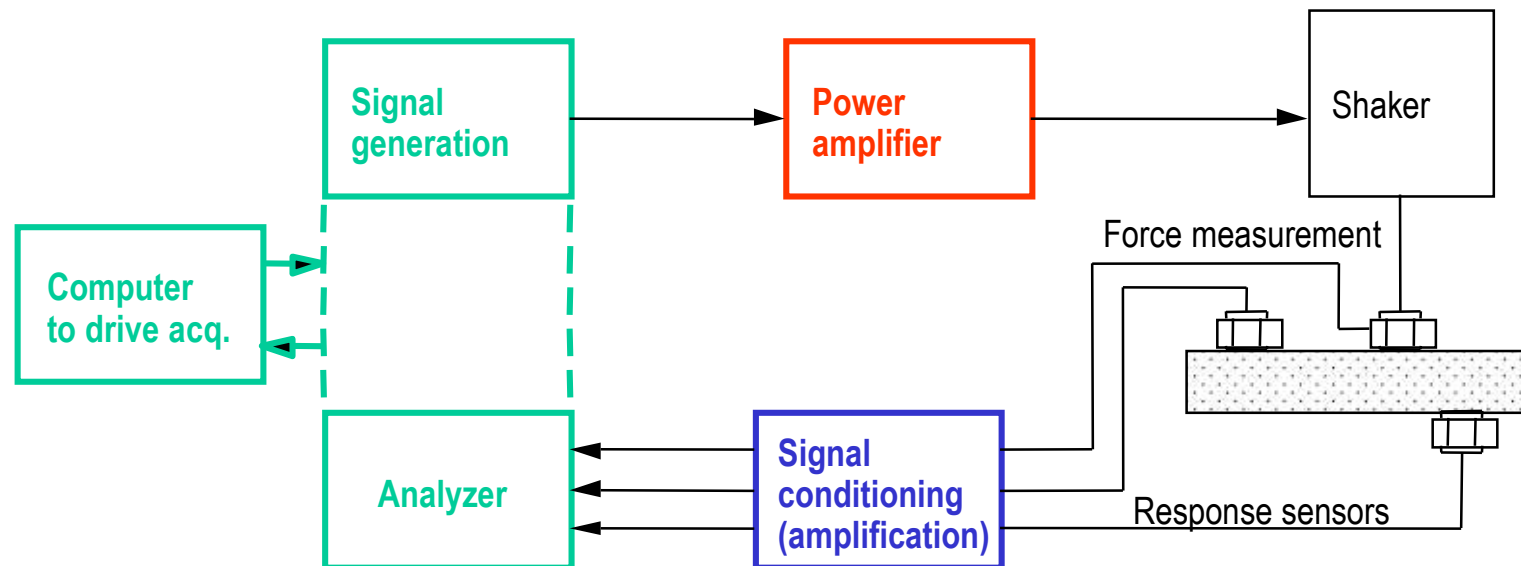
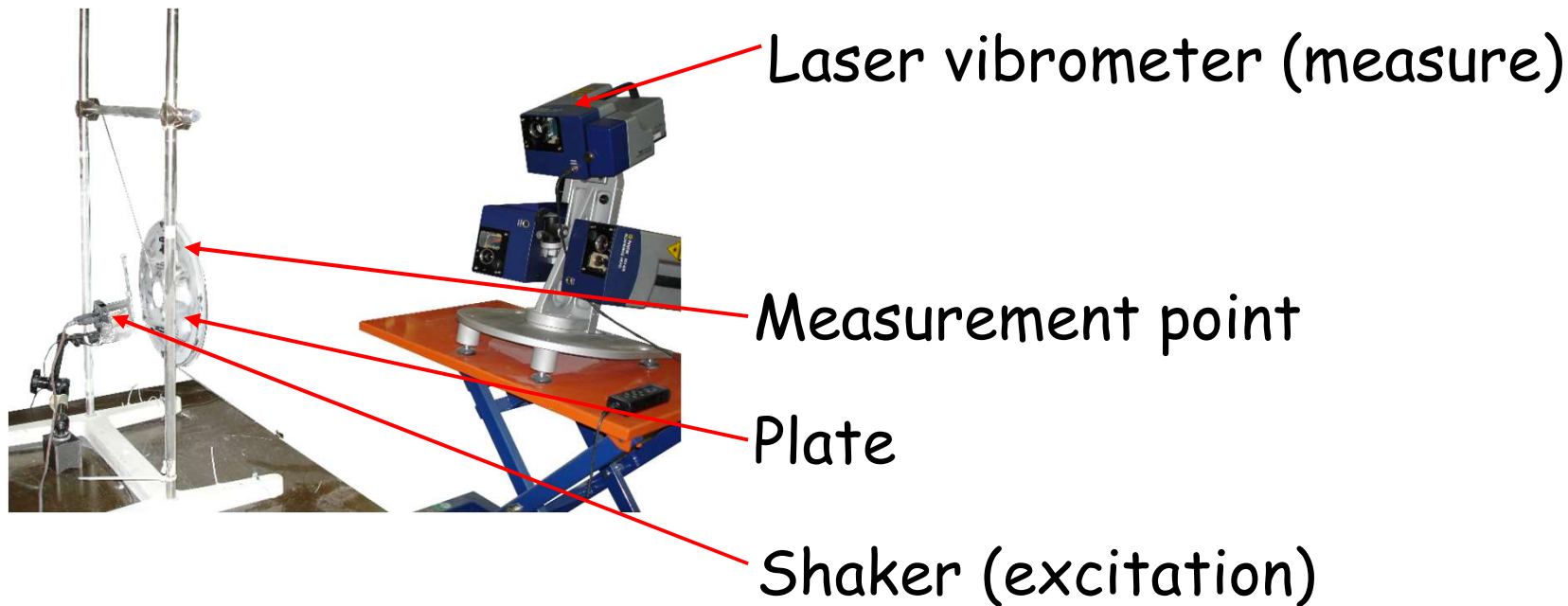


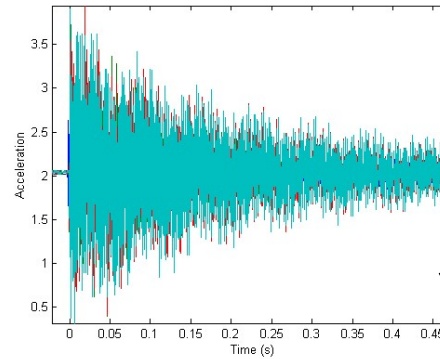
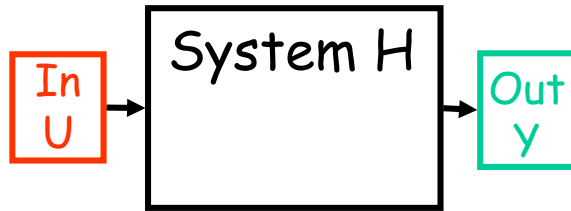
Identification (notes chapters 6-7, slides [Identification_EMA.pdf](#))

1. Identification demo
2. Inverse problems : model forms, data ([s 7.1](#))
3. Model forms for identification, discussion of residual terms
4. Frequency domain least-squares solution, implicit NL
5. Evaluation of results ([ch 8](#))

Experimental modal analysis : data



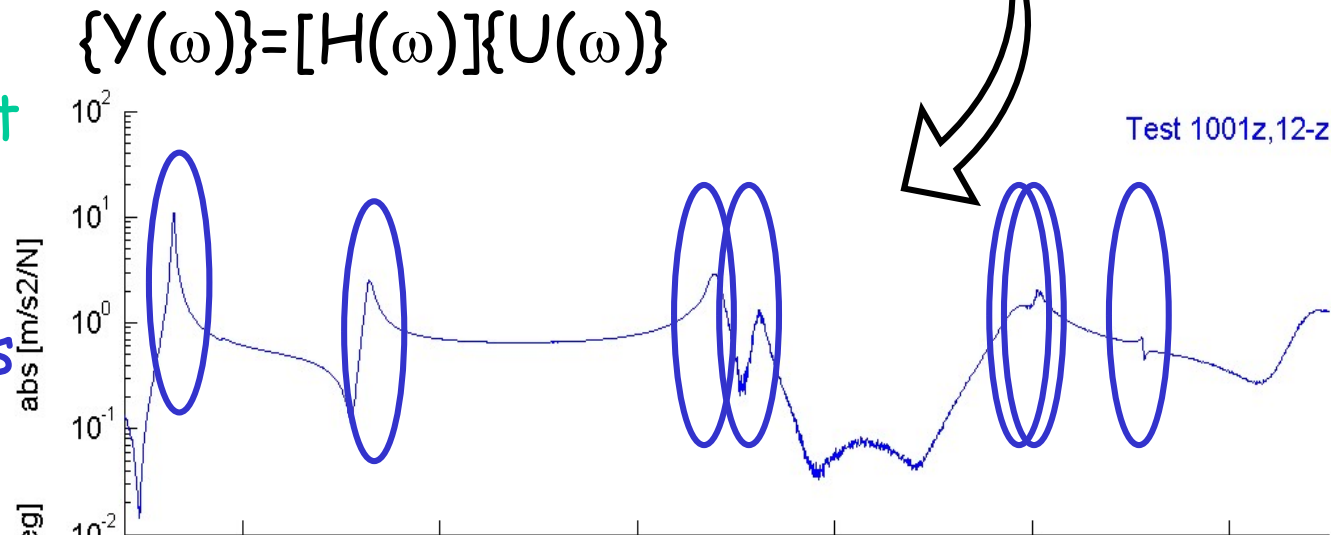
Assumption LTI system = transfers exist



Transfers estimated from time response

ONE input
ONE output

MANY resonances



$$\{Y(\omega)\} = [H(\omega)]\{U(\omega)\}$$

Test 1001z,12-z

Some standard texts for the vibration community

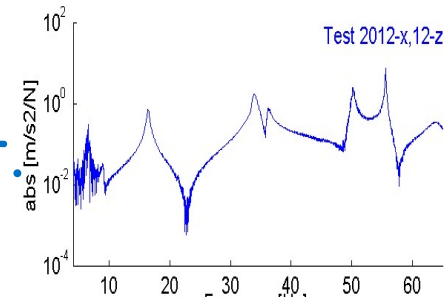
- [1] D. J. Ewins, Modal Testing: Theory and Practice. John Wiley and Sons, Inc., New York, NY, 1984.
- [2] W. Heylen et P. Sas, Modal analysis theory and testing. Katholieke Universiteit Leuven, Departement Werktuigkunde, 2006
- [3] K. G. McConnell, Vibration Testing. Theory and Practice. Wiley Interscience, New-York, 1995

Experimental modal analysis: an inverse problem

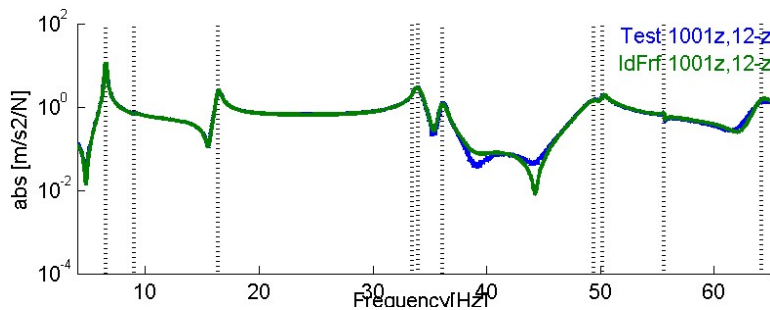
See also chapter 5 (signal processing & non-parametric ID) & Signal_Test.pdf

Data

- Transfer
- Hankel mat.

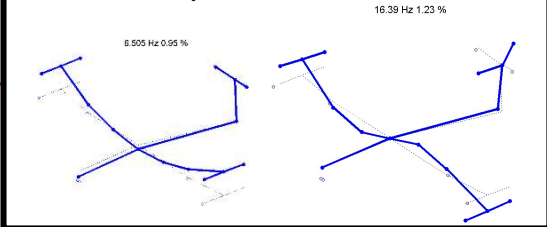


Objective $H_{\text{test}} - H_{\text{id}}$



Optimization

Result : parameters



Family of models

SVD $\sum_i \left\{ \begin{matrix} U_i \\ \downarrow \\ \text{sensors} \end{matrix} \right\} \{a_i(t) \rightarrow \text{time}\}$

State space $\{\dot{x}\} = [A]\{x\} + [B]\{u\}$

Pole residue $\sum \frac{[R_j]_{NS \times NA}}{s - \lambda_j}$

Rational fraction, modal model, second order ...

LTI model forms

- See MATLAB [control](#) or [scipy.signal.lti](#)

- **ss** state-space
$$\begin{aligned}\{\dot{x}\} &= [A]_{N \times N}\{x\} + [B]\{u\}_{NA \times 1} \\ \{y\}_{NS \times 1} &= [c]\{x\} + D\{u\}\end{aligned}$$

- Mechanical SS
$$\begin{aligned}\begin{Bmatrix} \dot{p} \\ \ddot{p} \end{Bmatrix} &= \begin{bmatrix} 0 & I \\ -[\omega_j^2] & -\Gamma \end{bmatrix} \begin{Bmatrix} p \\ \dot{p} \end{Bmatrix} + \begin{bmatrix} 0 \\ \phi_j^T b \end{bmatrix} \{u(t)\} \\ \{y(t)\} &= [c\phi_j \quad 0] \begin{Bmatrix} p \\ \dot{p} \end{Bmatrix}\end{aligned}$$

- **tf**
$$H = \frac{\sum_i [b_i]_{NS \times NA} s^i}{\sum_j a_j s^j}$$

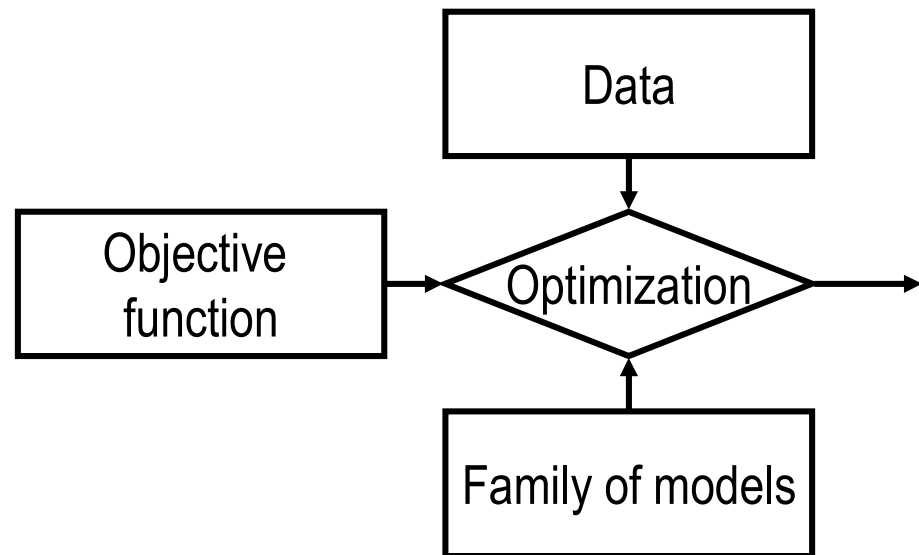
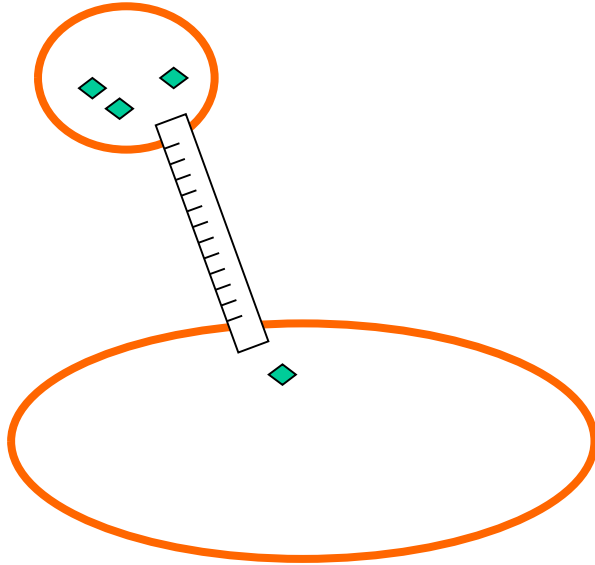
$$\begin{aligned}\begin{Bmatrix} \dot{q} \\ \ddot{q} \end{Bmatrix} &= \begin{bmatrix} 0 & I \\ -M^{-1}K & -M^{-1}C \end{bmatrix} \begin{Bmatrix} q \\ \dot{q} \end{Bmatrix} + \begin{bmatrix} 0 \\ M^{-1}b \end{bmatrix} \{u(t)\} \\ \{y(t)\} &= [c \quad 0] \begin{Bmatrix} q \\ \dot{q} \end{Bmatrix}\end{aligned}$$

- **zpk** (zero pole gain)
$$H = k \frac{\prod_{k=1}^N (s - [z_k]_{NS \times NA})}{\prod_{j=1}^N (s - p_j)}$$

- **residue** (partial fraction expansion = modal form)

$$[H(s)]_{NS \times NA} = \sum_{j=1}^{N/2} \left(\frac{[R_j]_{NS \times NA}}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j} \right)$$

Inverse problem

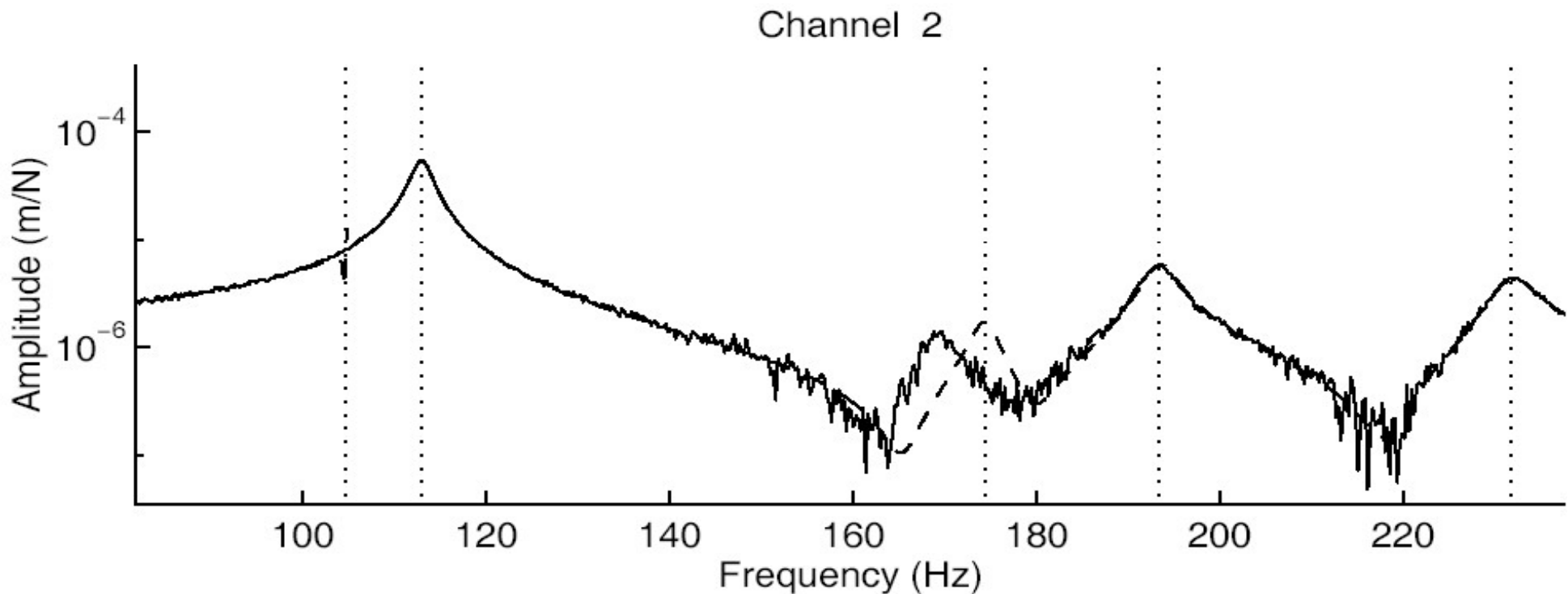


Objective function

- Least squares
 - Objective : minimize distance
 - Variants : Log-least squares, output-error, ...
- Singular value decomposition
 - Objective : constrained rank approximation

Example : FD polynomial

- This is done on the blackboard
- Section 7.4.1 (frequency domain ARX)



Circle fit / -3 dB

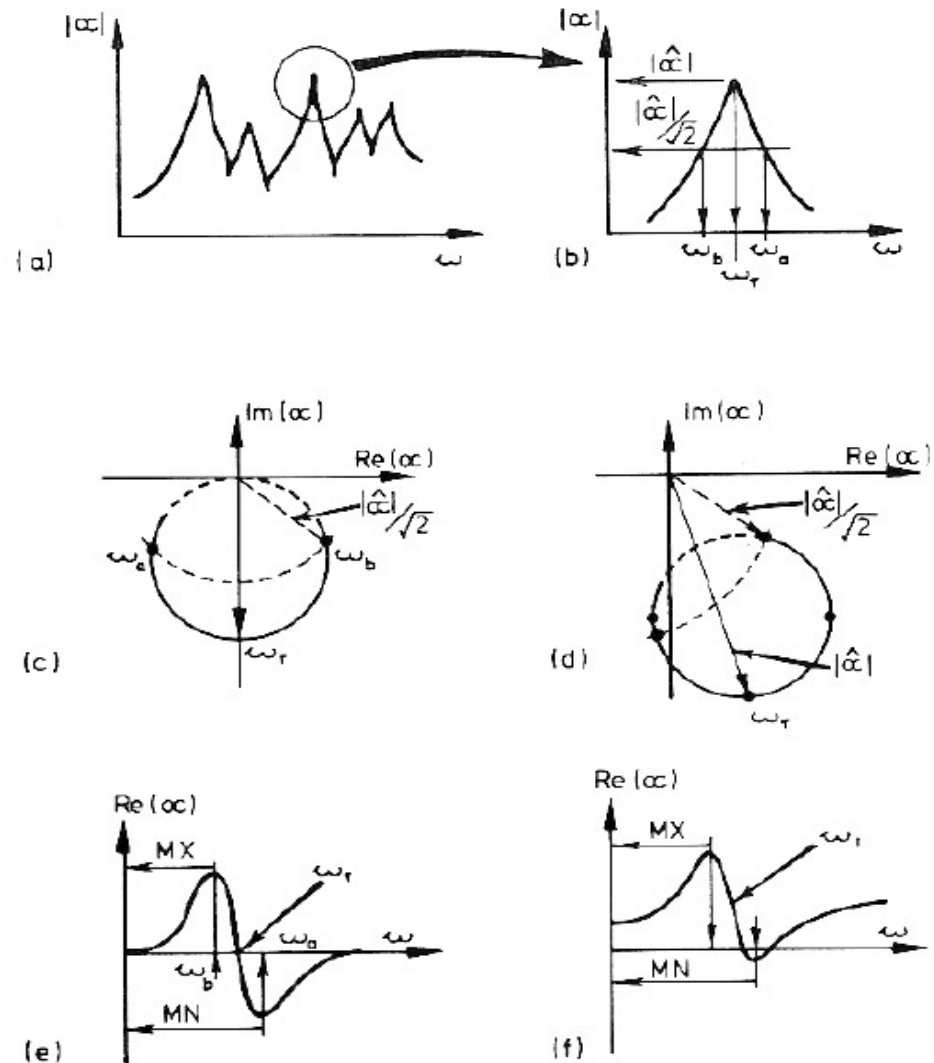
- Assumed model form

$$[H(s)] \approx \frac{A_j}{s^2 + 2\zeta_j\omega_j s + \omega_j^2}$$

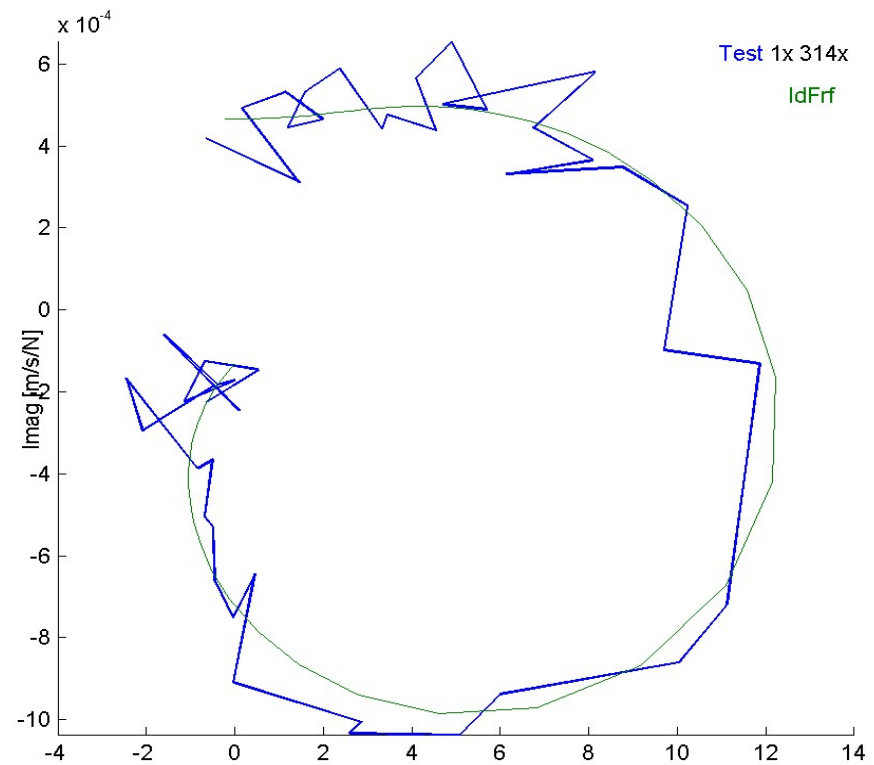
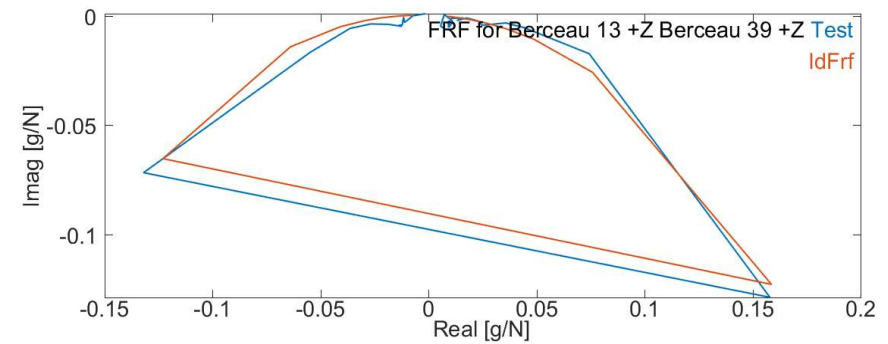
- Geometrical properties of Nyquist give ω, ζ

$$\zeta_j \approx \frac{\omega_b - \omega_a}{2\omega_n}$$

- Poor accounting of other contributions

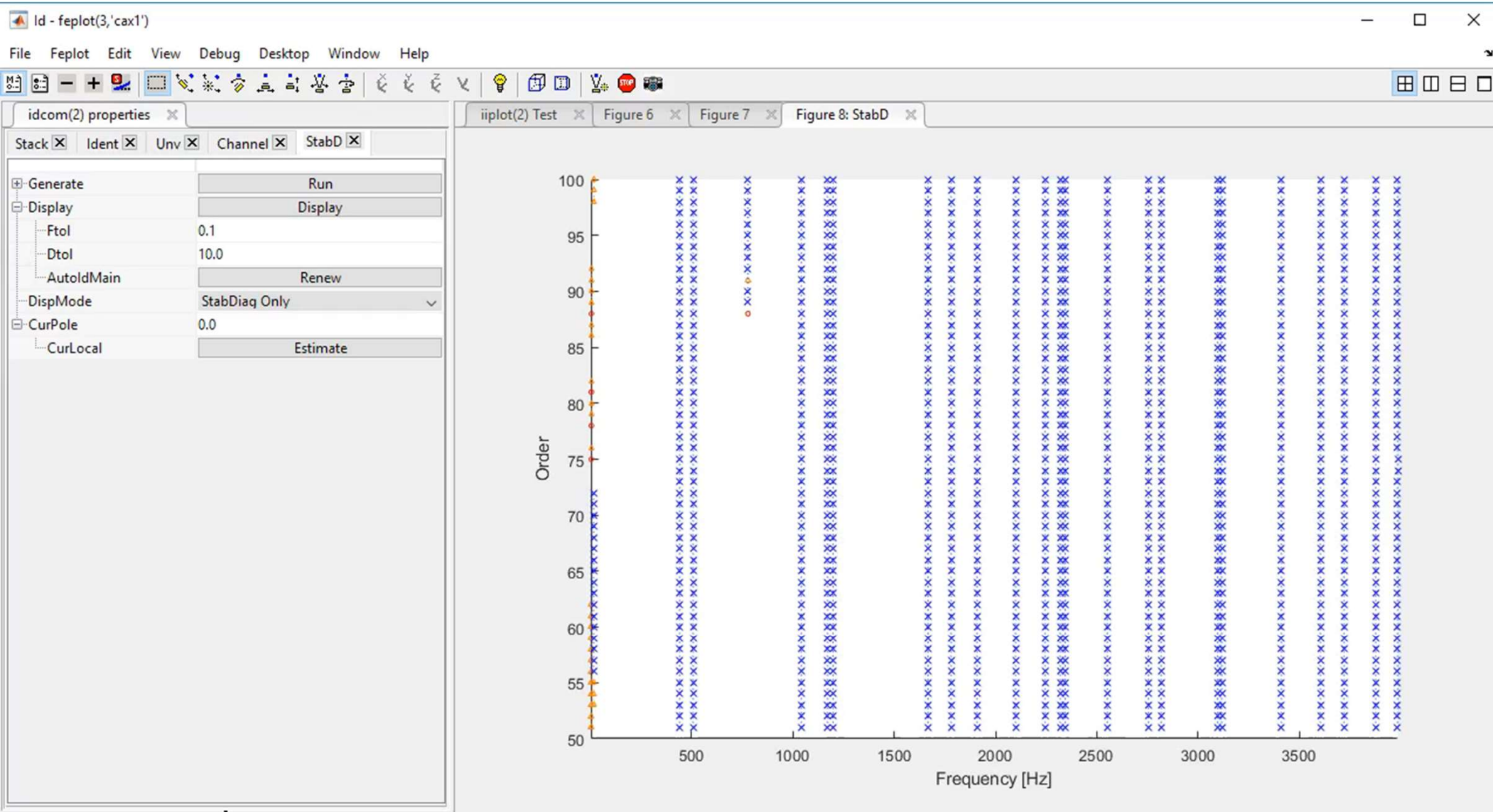


-3dB failures



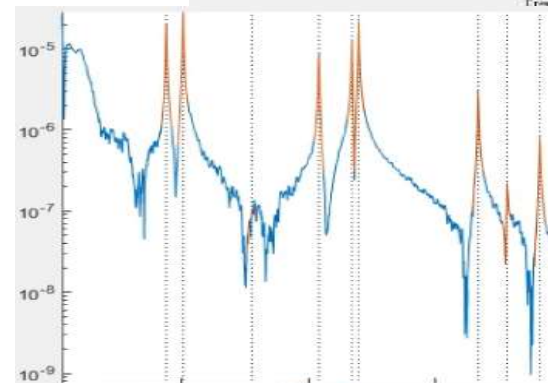
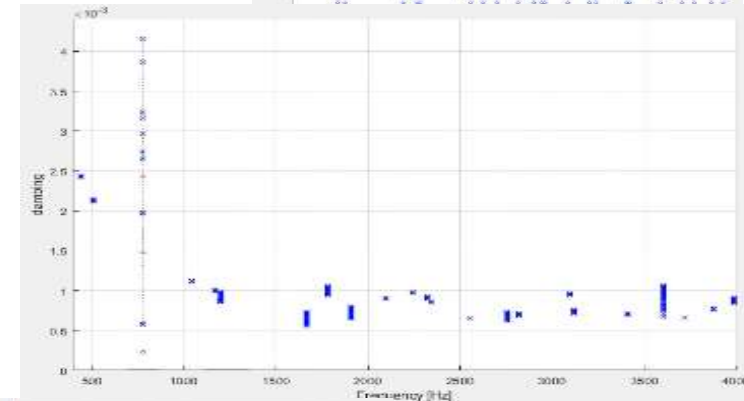
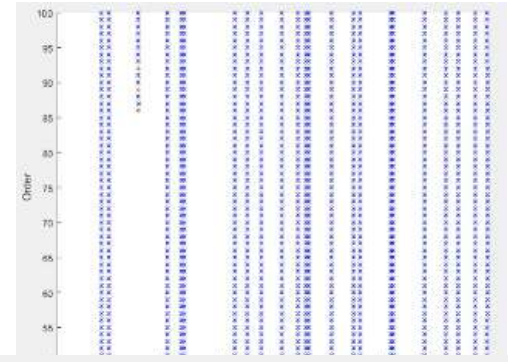
This is done on the blackboard 10

Classical linear system Id



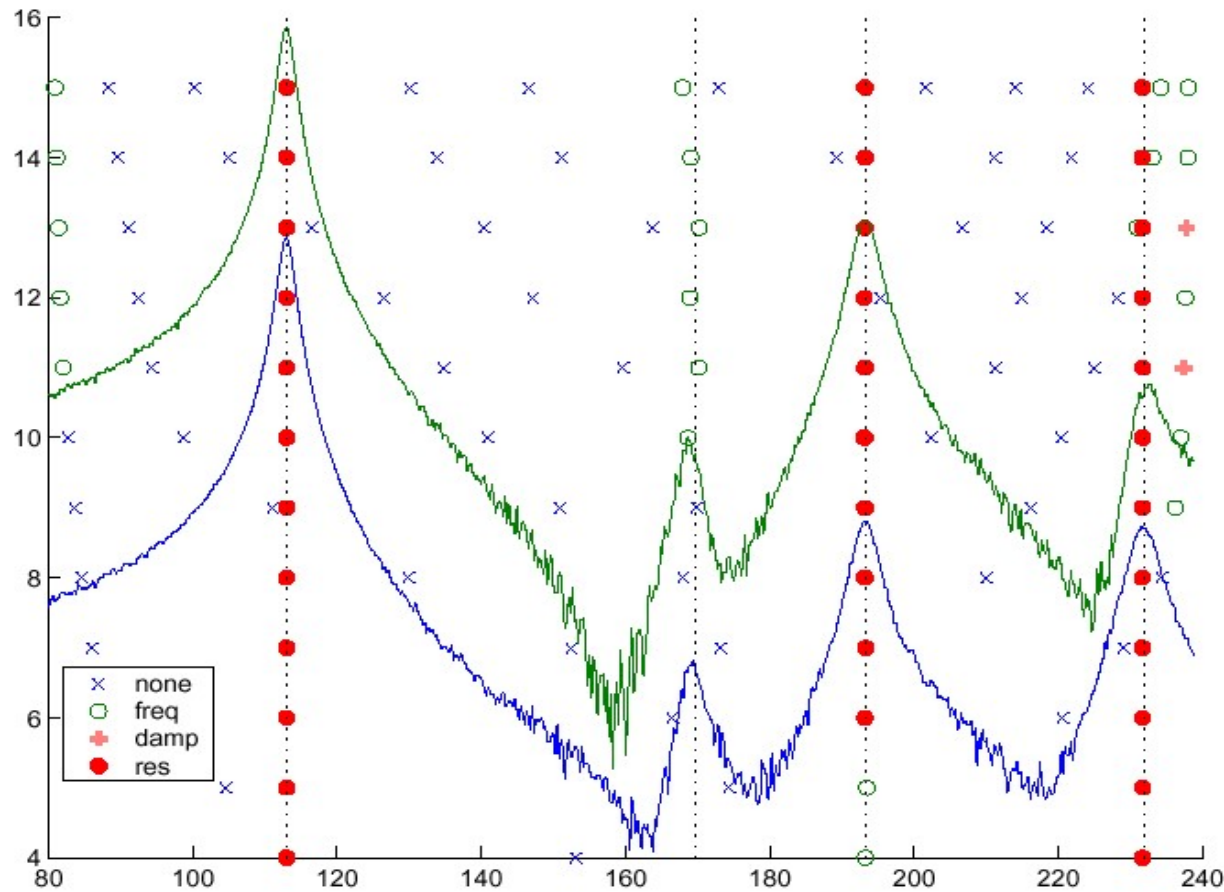
Identification phases

- Initialization pick from :
 - Single pole estimate [1]
 - Stabilization diagram [2]
- Estimate by band (why ? [1,3])
- How can problems be detected ? [3-4]
- Re-optimize poles (why ?)



- [1] E. Balmes, « Frequency domain identification of structural dynamics using the pole/residue parametrization », IMAC 1996
- [2] P. Verboven, « Frequency-domain system identification for modal analysis », Ph.D. thesis, 2002.
- [3] G. Martin, « Méthodes de corrélation calcul/essai pour l'analyse du cisement », Ph.D. CIFRE SDTools, Arts et Metiers ParisTech, Paris, 2017
- [4] G. Martin, E. Balmes, et T. Chancelier, « Characterization of identification errors and uses in localization of poor modal correlation », MSSP 2017.

Stabilization diagrams

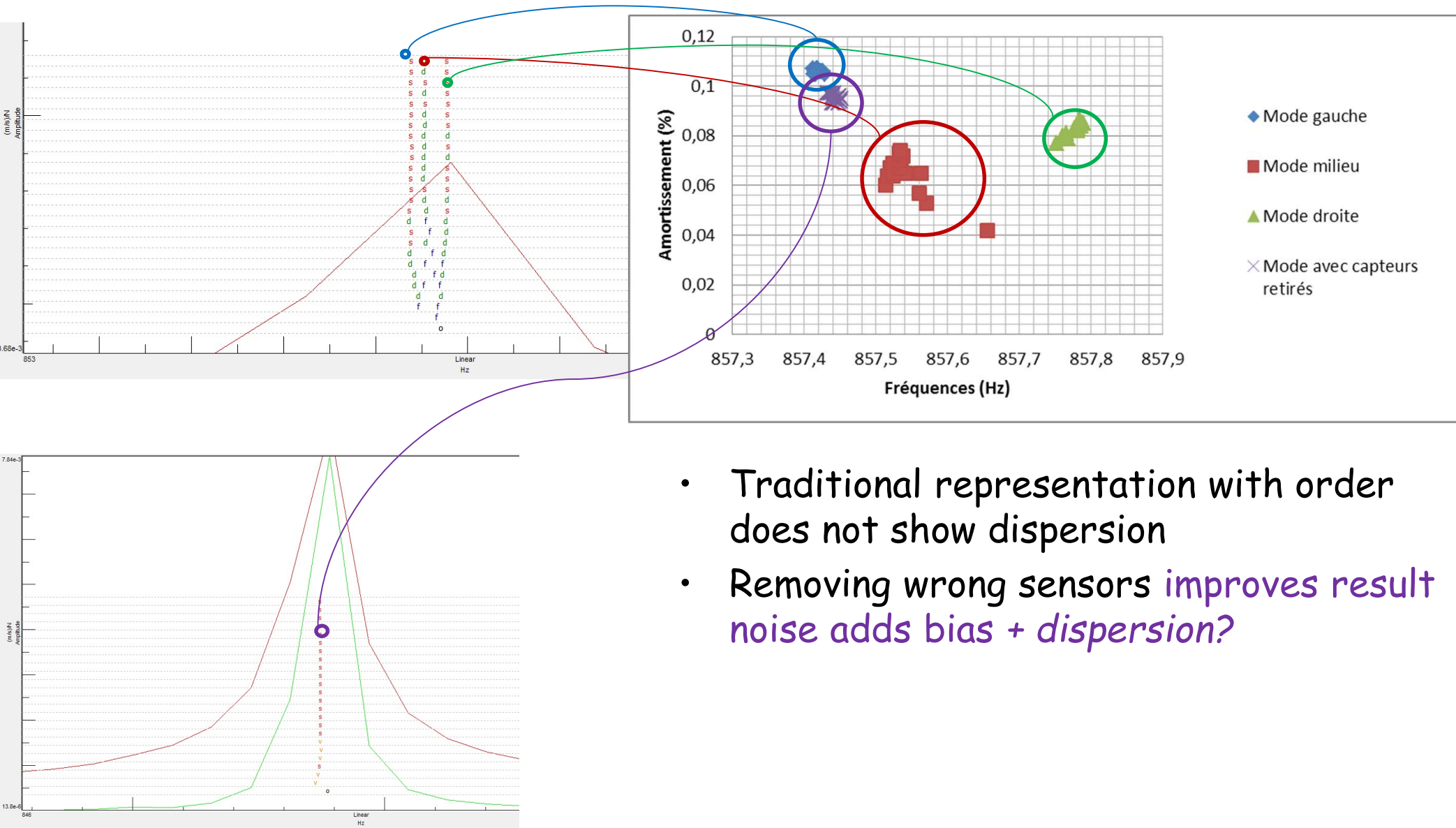


$$\Delta\omega_j/\omega_j < 1\%$$

$$\Delta\zeta_j/\zeta_j < 10\%$$

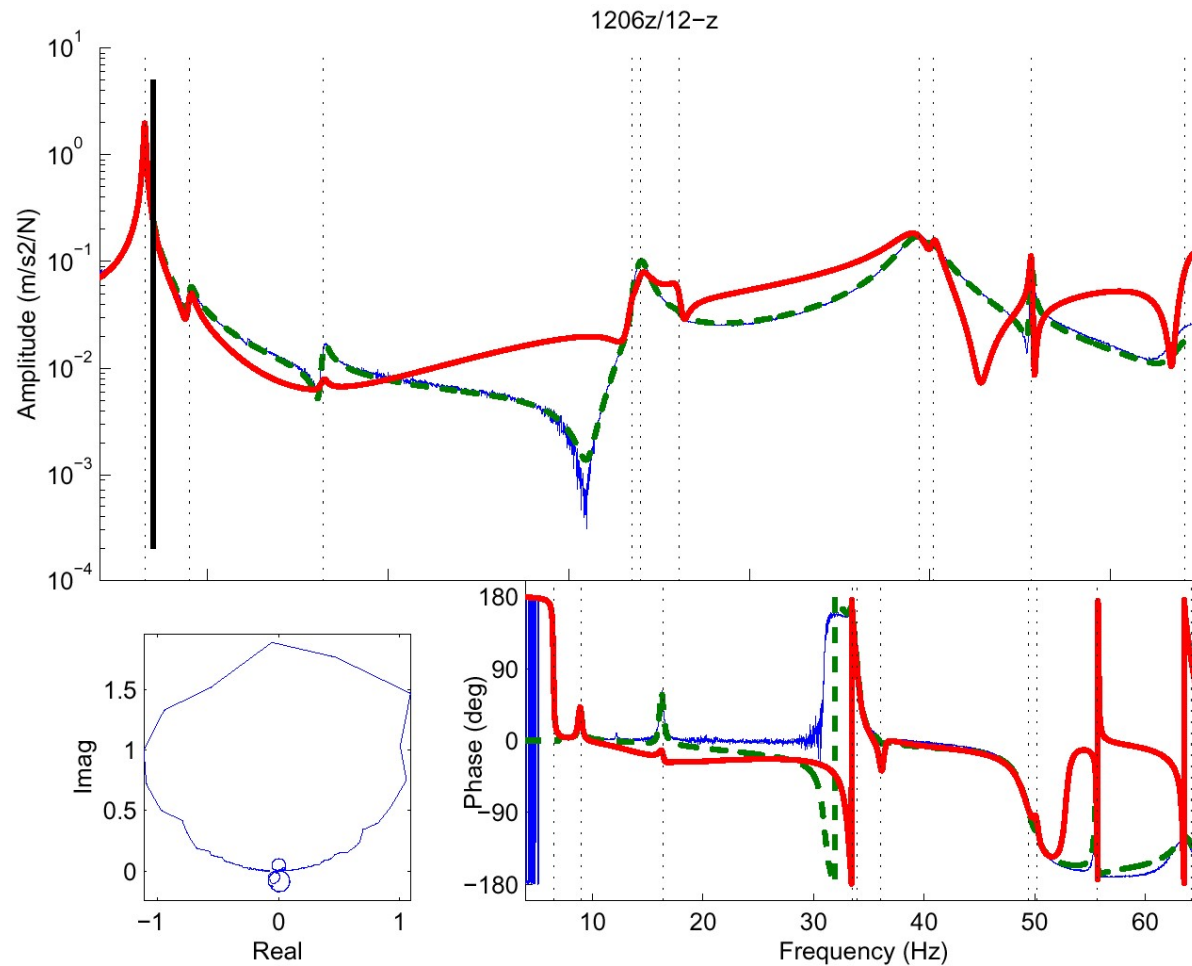
$$\Delta R_j/R_j < 10\%$$

Stabilization diagrams & bias



- Traditional representation with order does not show dispersion
- Removing wrong sensors improves result
noise adds bias + dispersion?

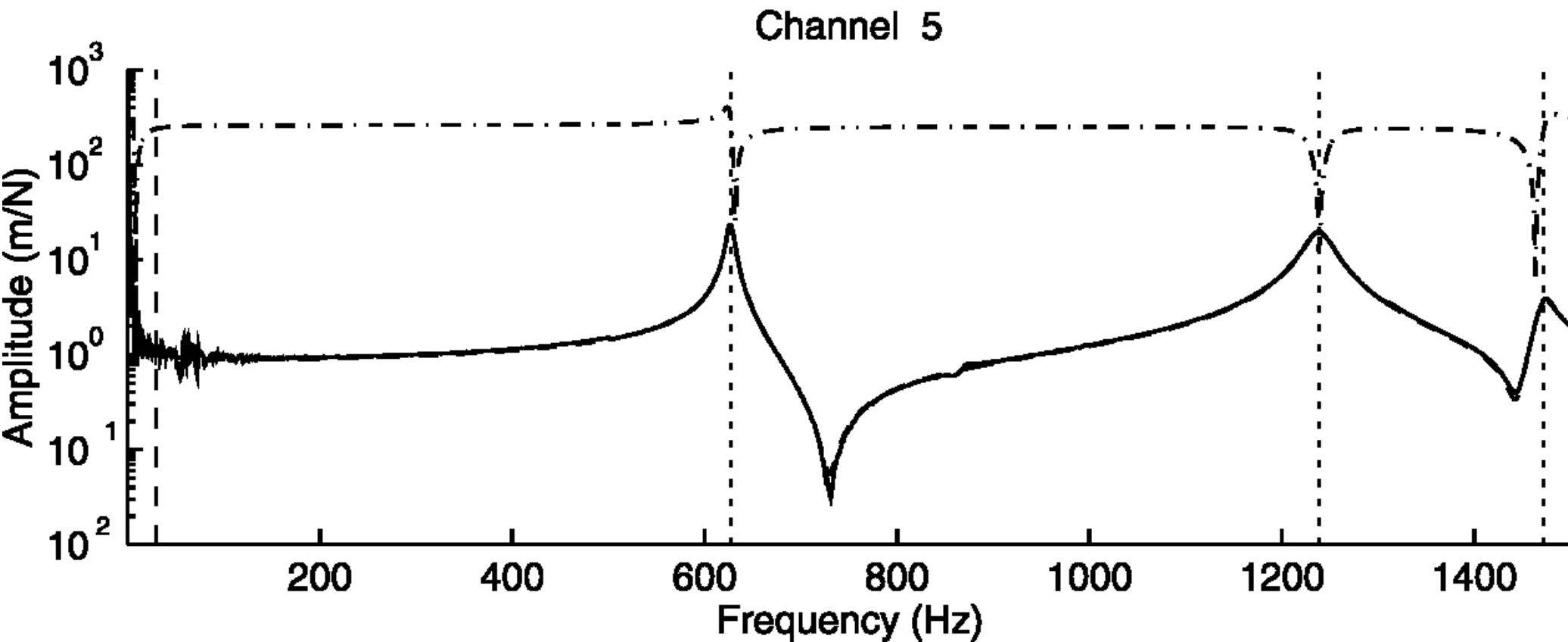
Noise induced bias -> work by band



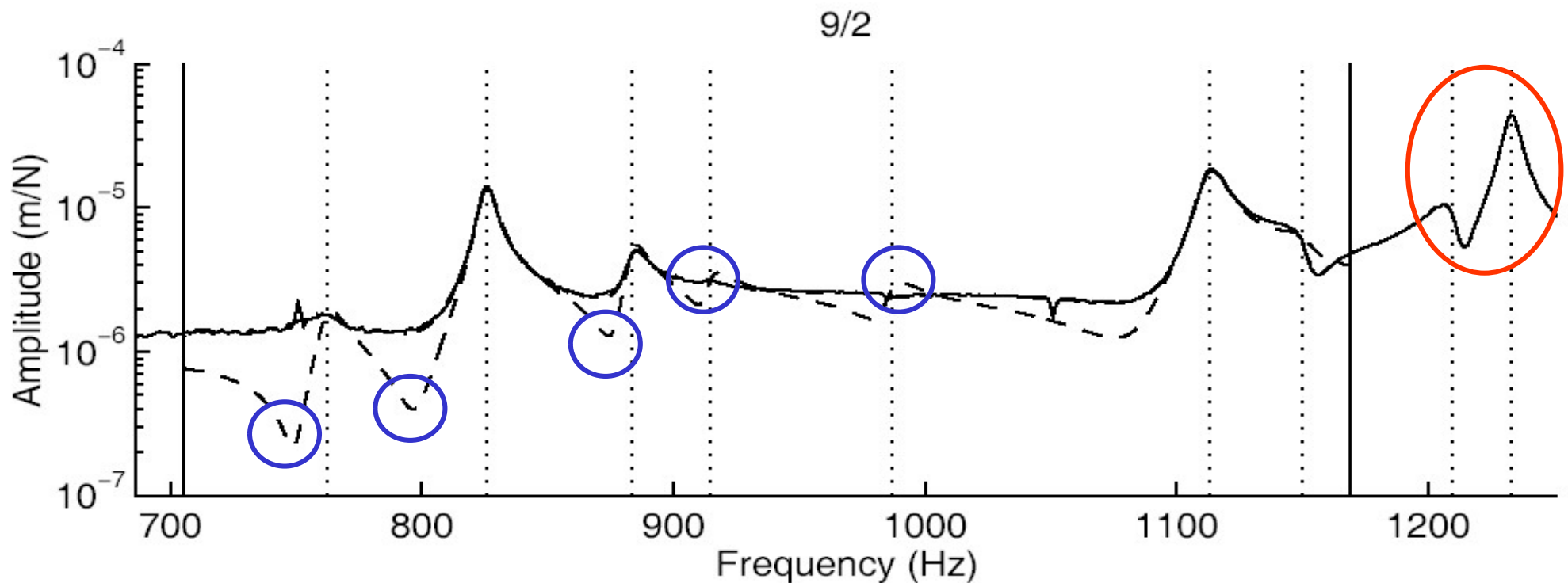
Inconsistence between data and family of models induces **bias** / variability

- [1] M. Böswald, « Analysis of the bias in modal parameters obtained with frequency-domain rational fraction polynomial estimators », Proceedings of ISMA 2016
- [2] El-kafafy, De Troyer, Peeters, Guillaume, « Fast maximum-likelihood identification of modal parameters with uncertainty intervals: A modal model-based formulation », MSSP, 2013

Low frequency noise bias



Effects of residual terms



Here : ignoring **two nearby modes** induces **bias**

In general : account for flexibility and inertia effects (lower/upper residual)

Identification model

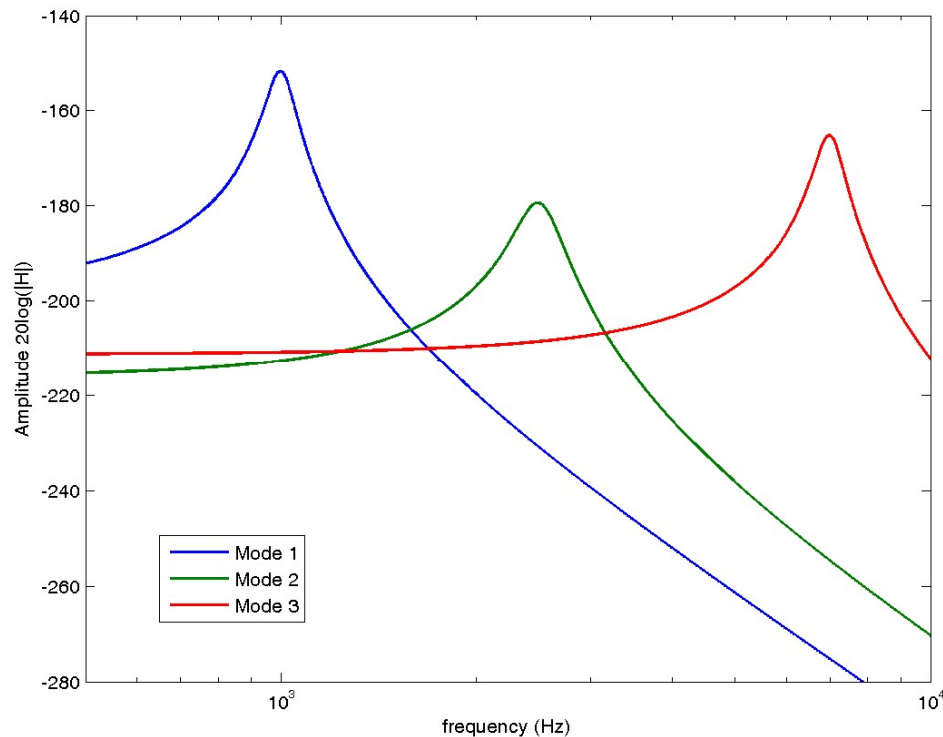
- Pole/Residue with high and low residual terms

$$[H_j(s)] = \frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j}$$

$[H_1(s)]$

$[H_2(s)]$

$[H_3(s)]$



Identification model

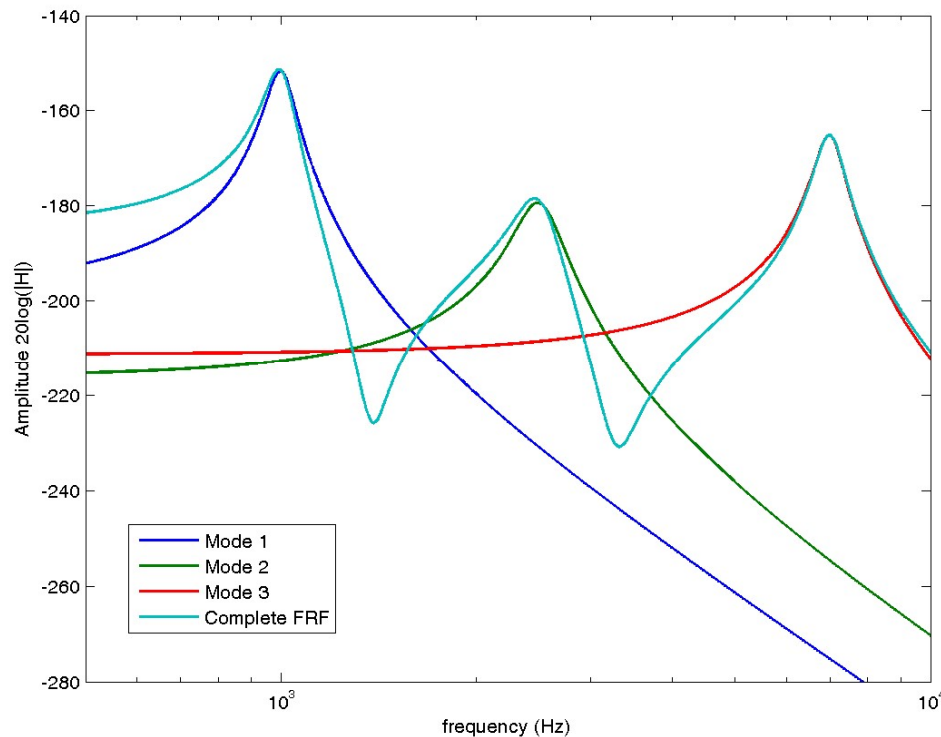
- Pole/Residue with high and low residual terms

$$[H_j(s)] = \frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j}$$

$$[H_1(s)]$$

$$[H_2(s)]$$

$$[H_3(s)]$$

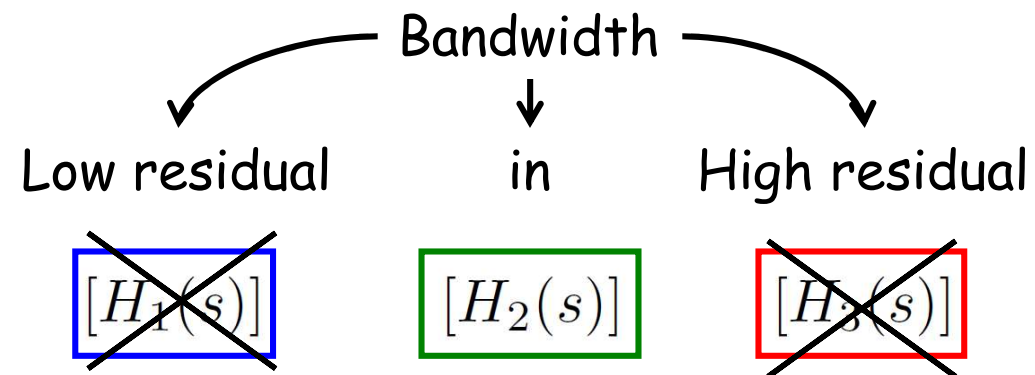
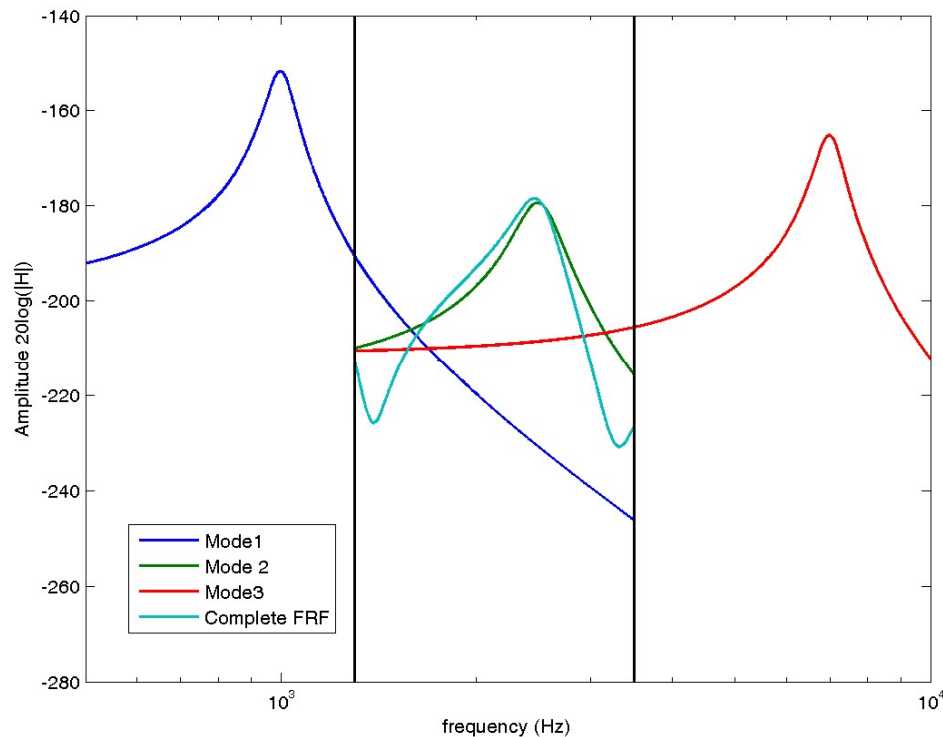


$$[H_{model}(s)] = [H_1(s)] + [H_2(s)] + [H_3(s)]$$

Identification model

- Pole/Residue with high and low residual terms

$$[H_j(s)] = \frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j}$$

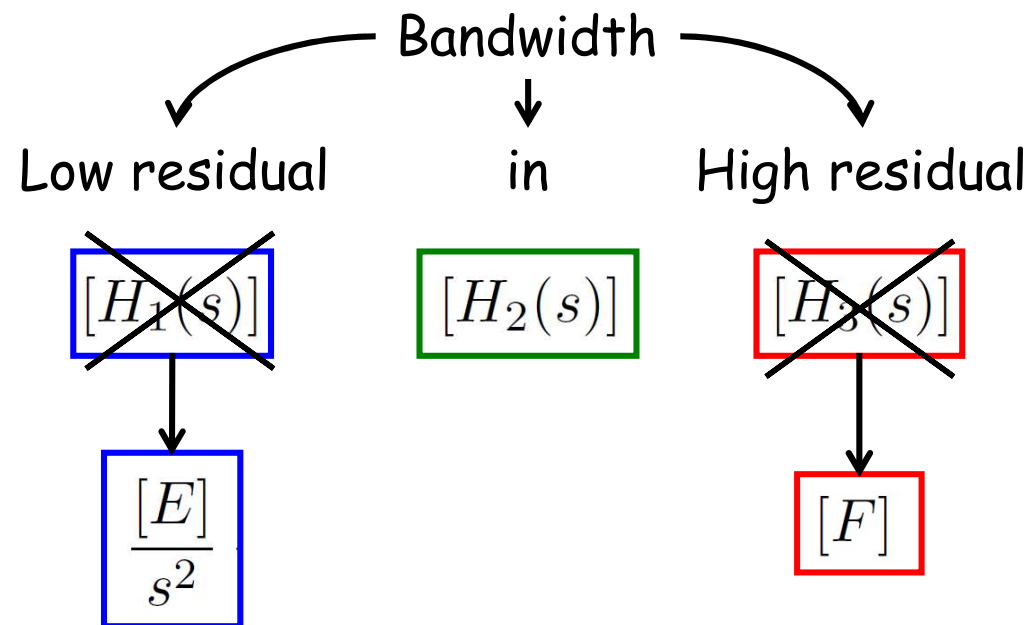
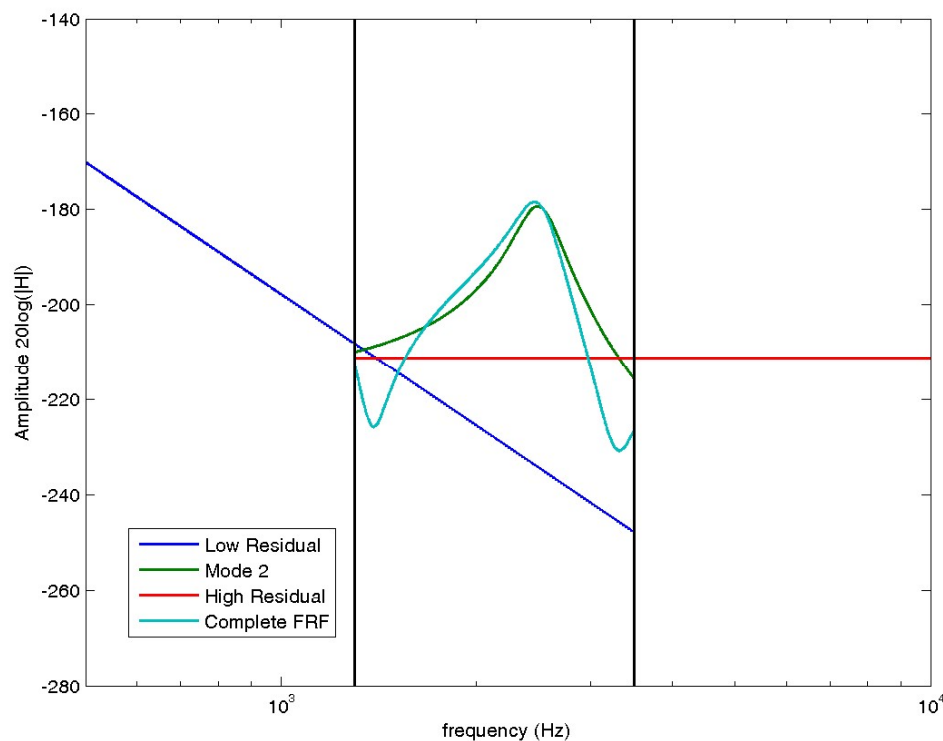


$$[H_{model}(s)] = \cancel{[H_1(s)]} + [H_2(s)] + \cancel{[H_3(s)]}$$

Identification model

- Pole/Residue with high and low residual terms

$$[H_j(s)] = \frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j}$$

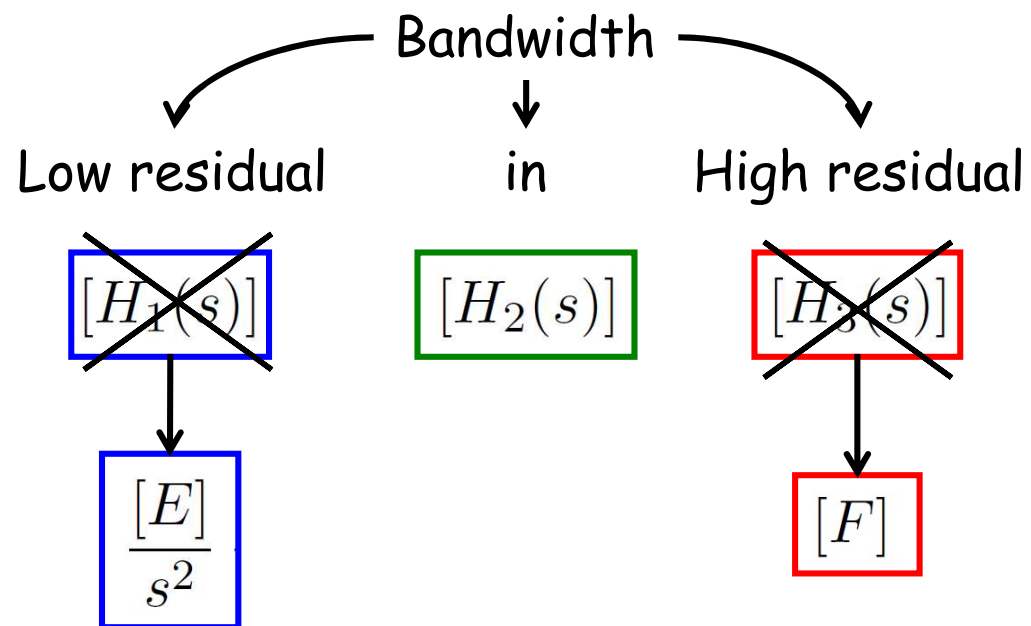
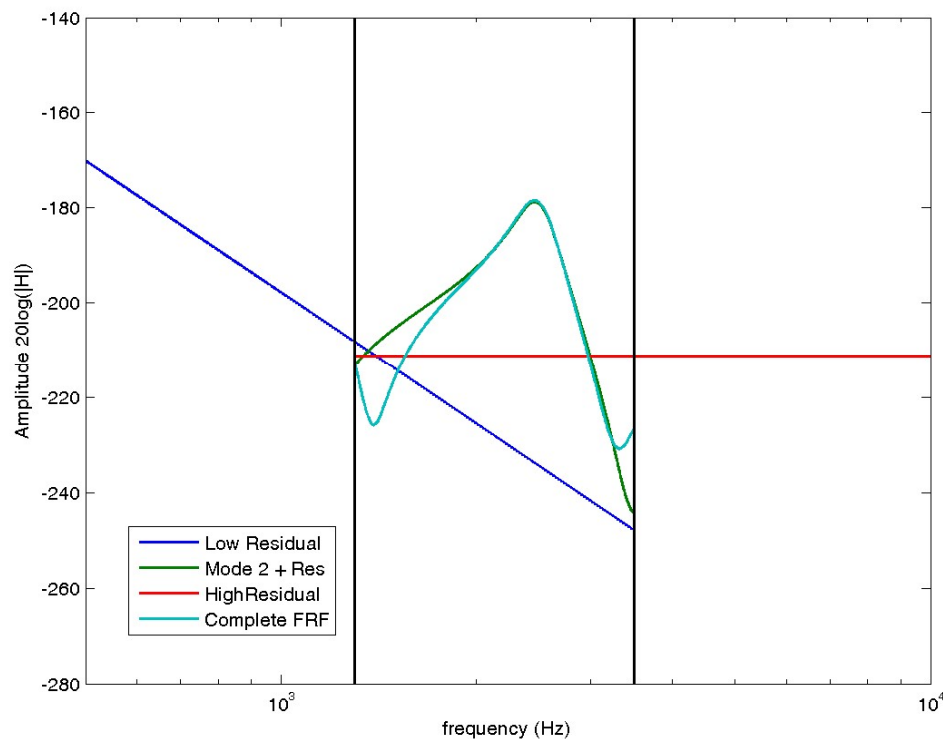


$$[H_{model}(s)] = \cancel{[H_1(s)]} + [H_2(s)] + \cancel{[H_3(s)]}$$

Identification model

- Pole/Residue with high and low residual terms

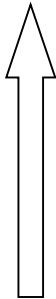
$$[H_j(s)] = \frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j}$$



$$[H_{model}(s)] = \frac{[E]}{s^2} + [H_2(s)] + [F]$$

Identification strategies

$$\min_{\lambda_j} \min_{R_j, E, F} \left| [\alpha(s)]_{test} - \sum_{j \in \text{identified}} \left(\frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j} \right) + [E(s)] \right|^2$$



Freq. domain linear least squares with residuals

almost always used

Unknowns λ_j ($2 \cdot N_p$), R_j ($N_s \cdot N_a \cdot 2 \cdot N_p$), E, F ($N_s \cdot N_a \cdot 2$)

Determination of poles : main differentiator between methods

Two main strategies

- select in over-specified set
- build gradually

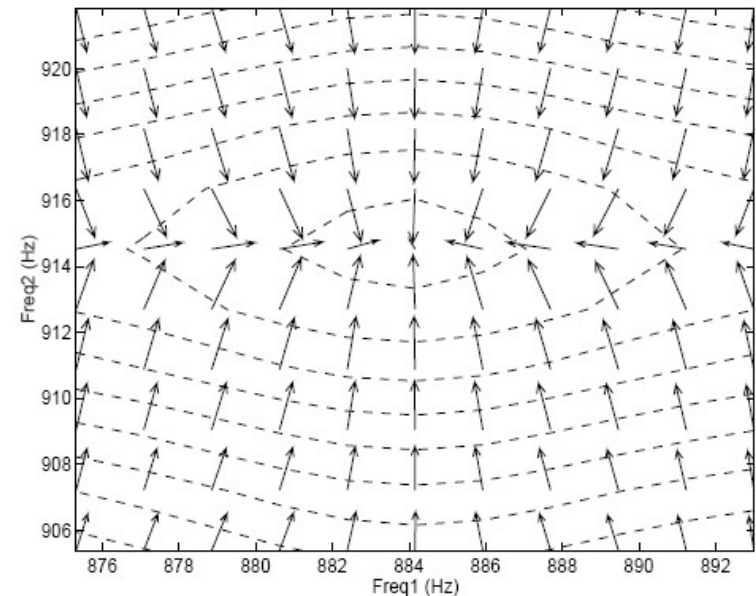
Frequency domain output error

$$\min_{\lambda_j} \min_{R_j, E, F} \left| [\alpha(s)]_{test} - \sum_{j \in \text{identified}} \left(\frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j} \right) + [E(s)] \right|^2$$

- R_j, E, F implicit functions of λ_j
- $\frac{\partial J}{\partial \lambda_j} = \frac{\partial R^T R}{\partial \lambda_j} = 2R^T \frac{\partial R}{\partial \lambda_j}$ can be computed analytically
- around a pole gradient not very affected by error on other poles

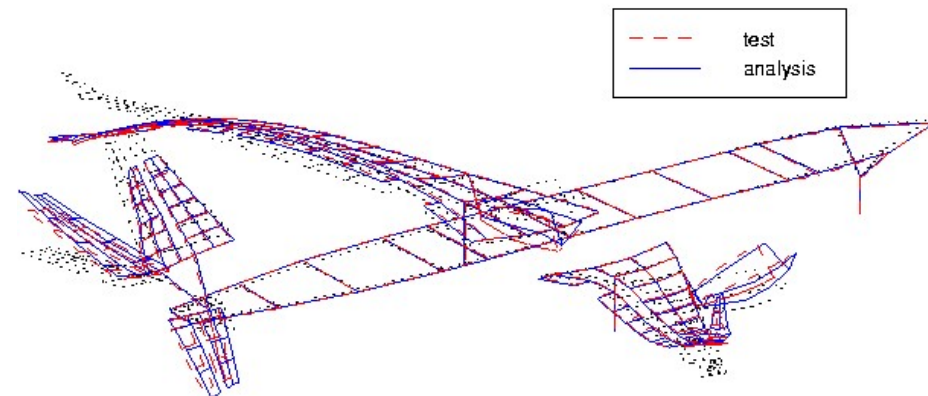
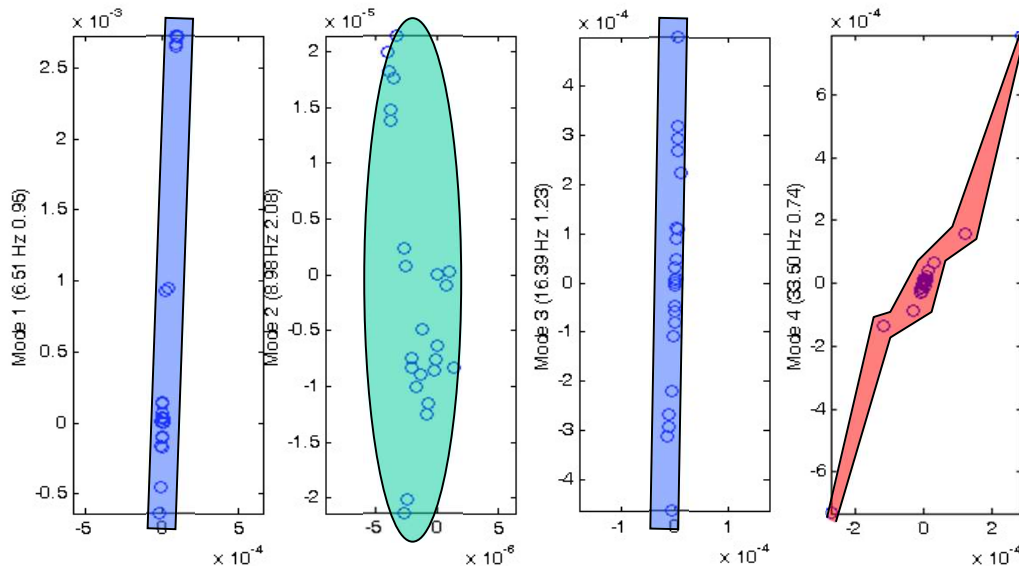
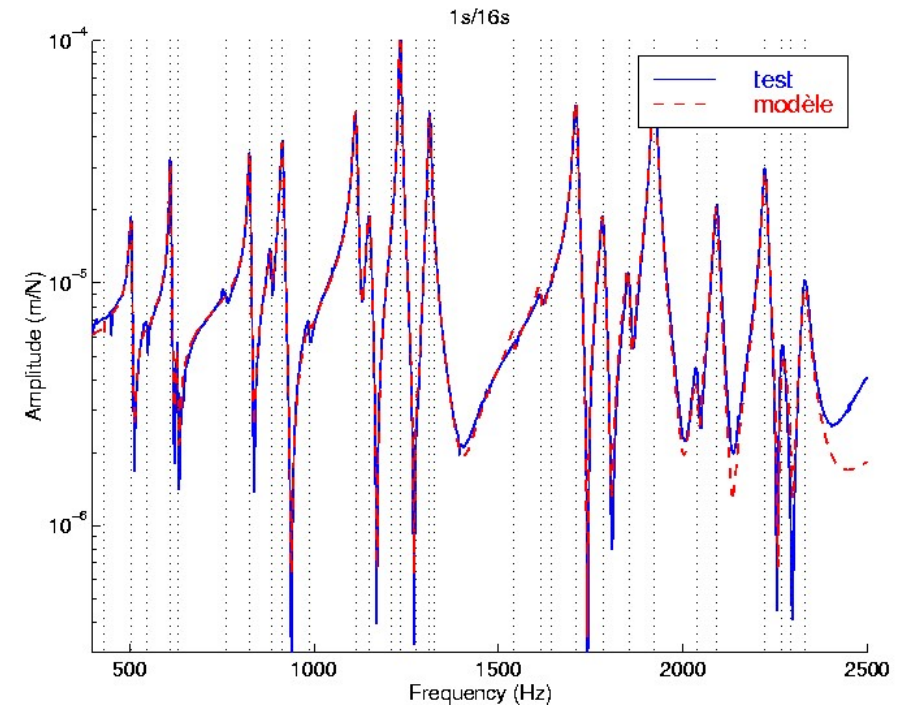
SDT implements

- Fast optimization
- All λ_j changed at each step using gradient only



Validations of results

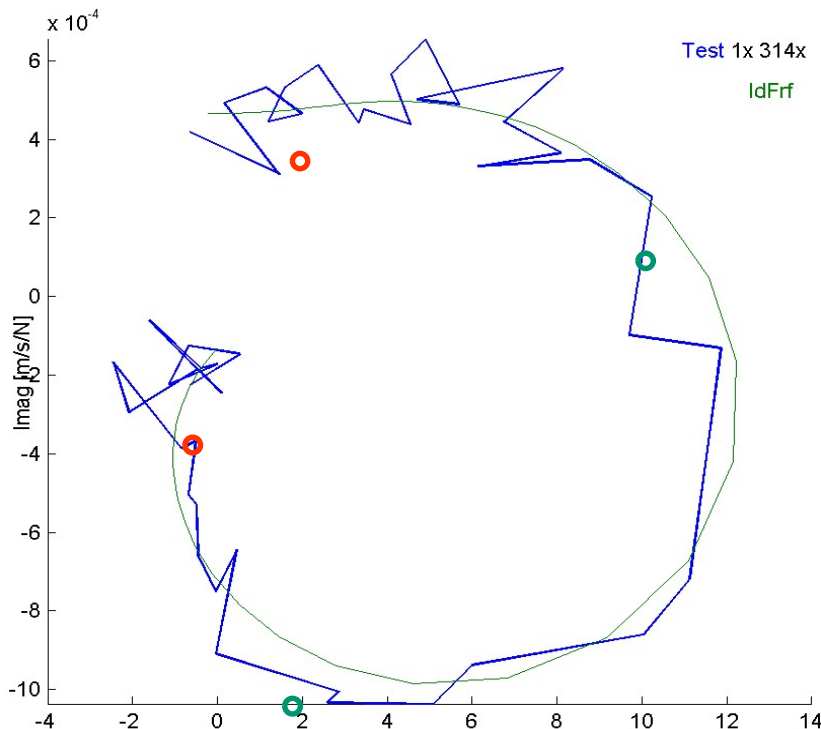
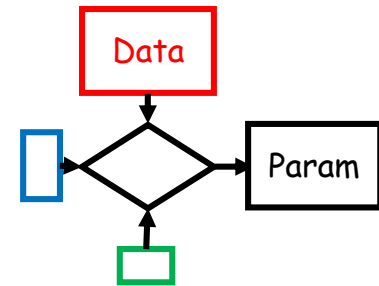
- FRF Superposition
- Modeshape visualization
- Residue properties



Quality : an error criterion

Is the model well identified?

- Superpose measured and identified FRF for **each sensor c**, **around each mode j**
- Compute error on Nyquist



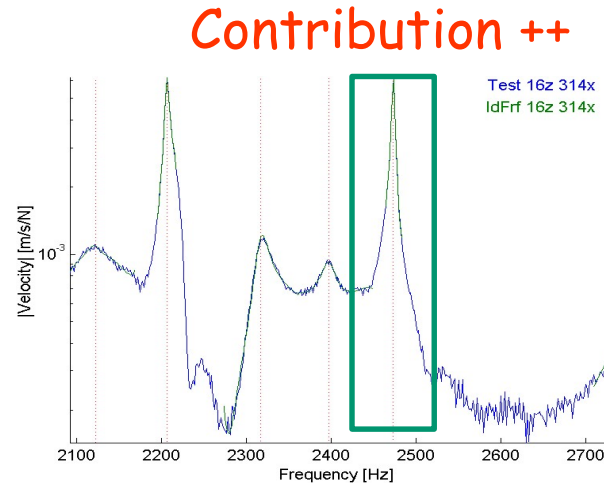
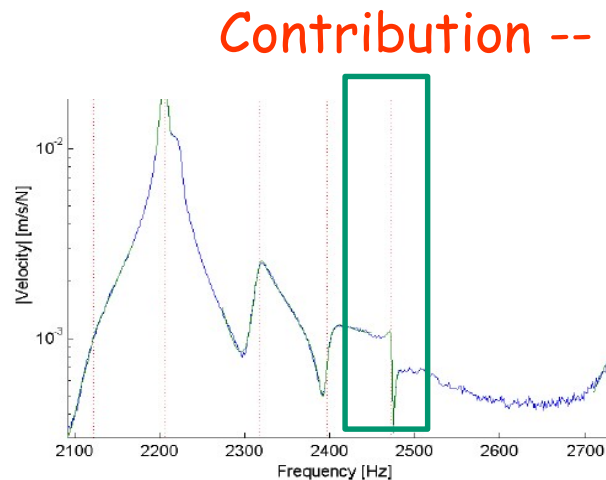
$$e_{jc} = \frac{\int \omega_j (1 + \alpha \zeta_j) |H_{cTest} - H_{cId}|^2}{\int \omega_j (1 + \alpha \zeta_j) |H_{cTest}|^2}$$

"Around each mode" :

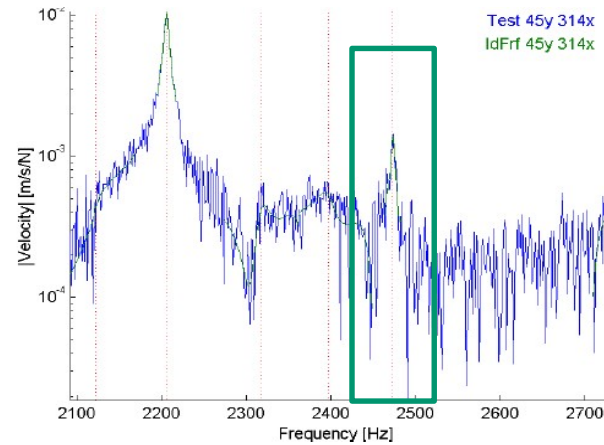
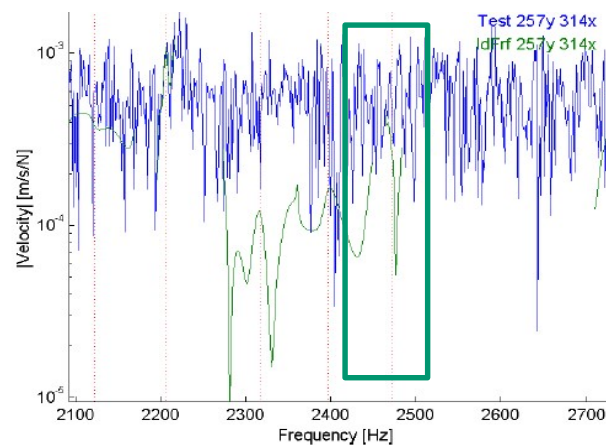
- **Half power** : $\pm \zeta_j \omega_j$
- **Visible peak** : $\pm 5 \zeta_j \omega_j$

Test/identification error

Error --



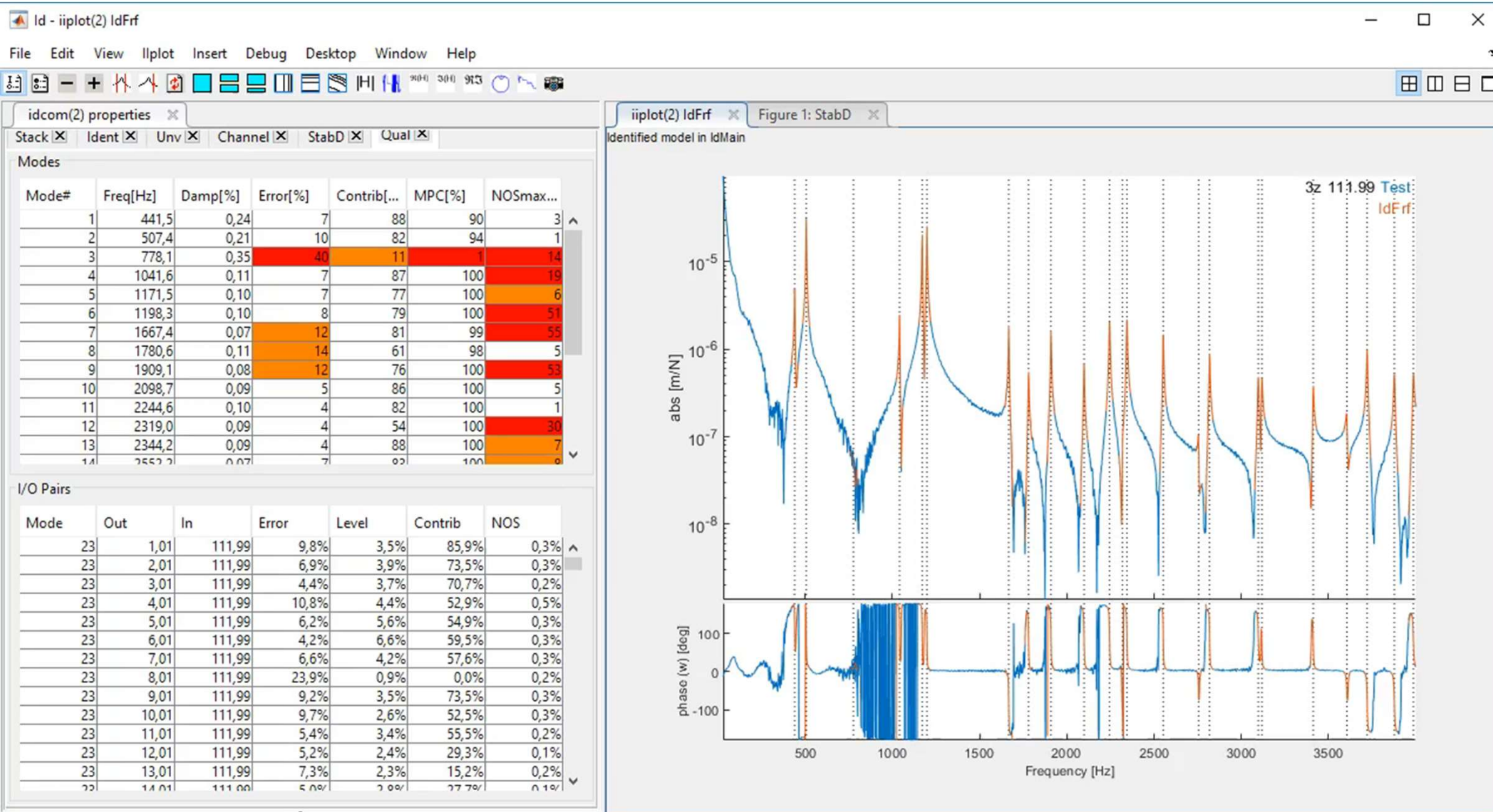
Error ++



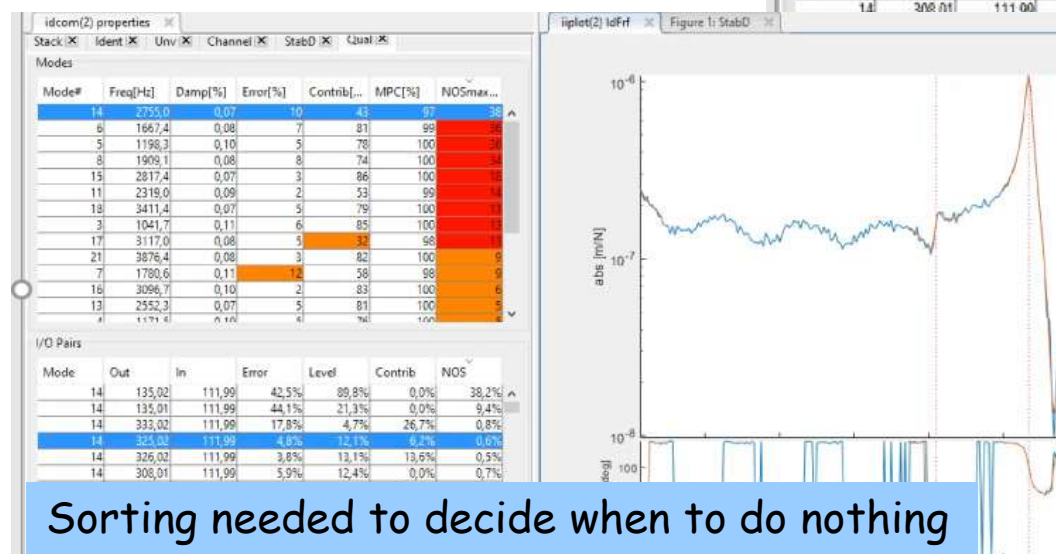
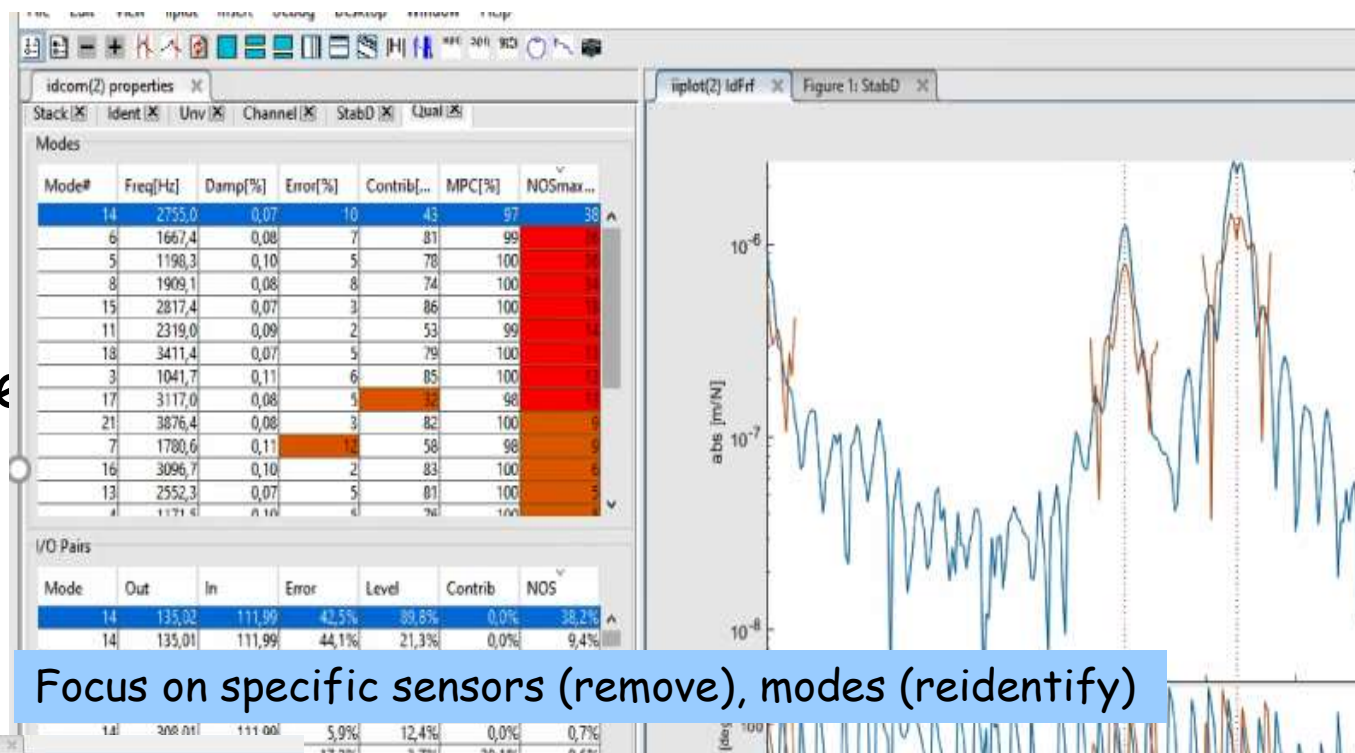
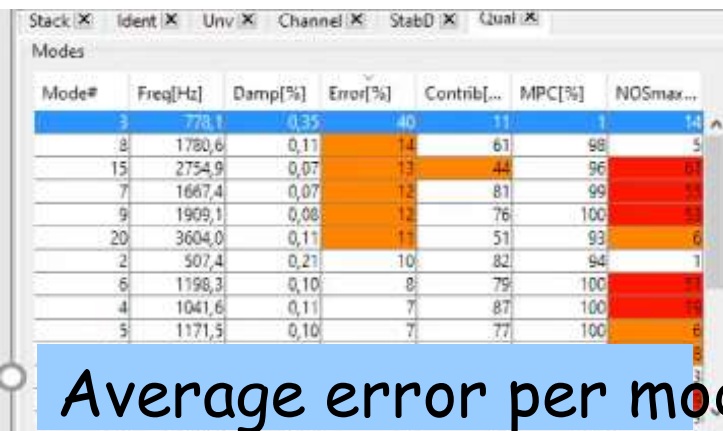
For each sensor, each mode may have strong **error/noise** & low **contribution**

Test error per mode/sensor

Evaluate quality of identification

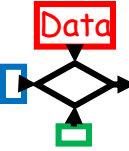


Error analysis usage

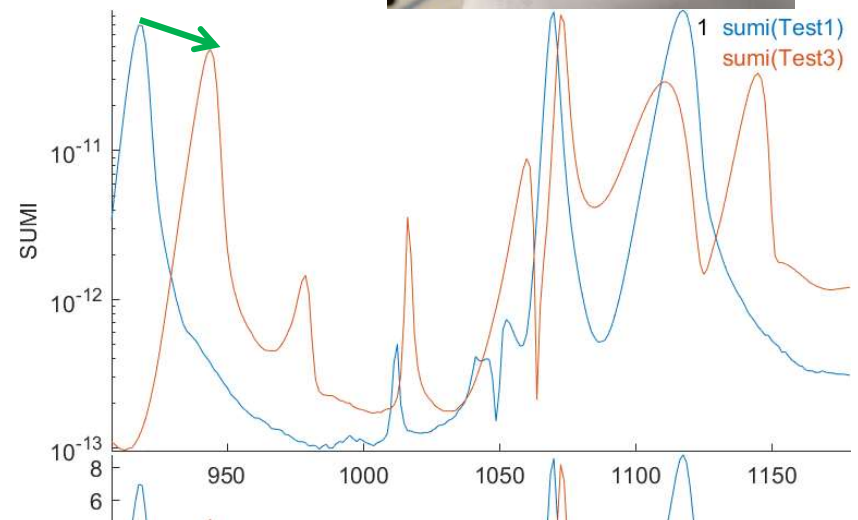
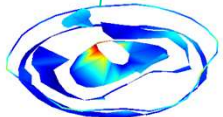
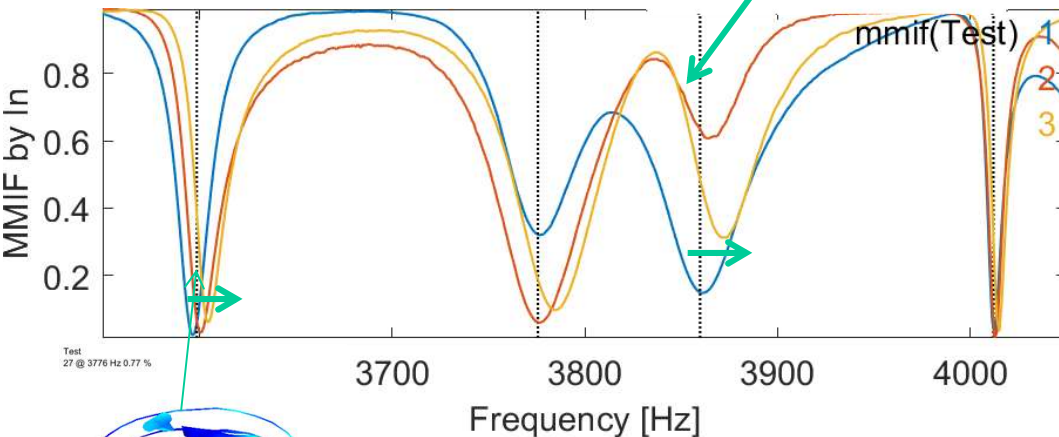
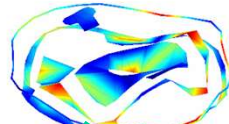


Other errors : NL/LPV $\neq$ one system

- Non linear system : resonance dependent on input point / accurate positioning
- MMIF or identification per impact location shows significant dispersion
- Multiple identification results are not perfectly coherent

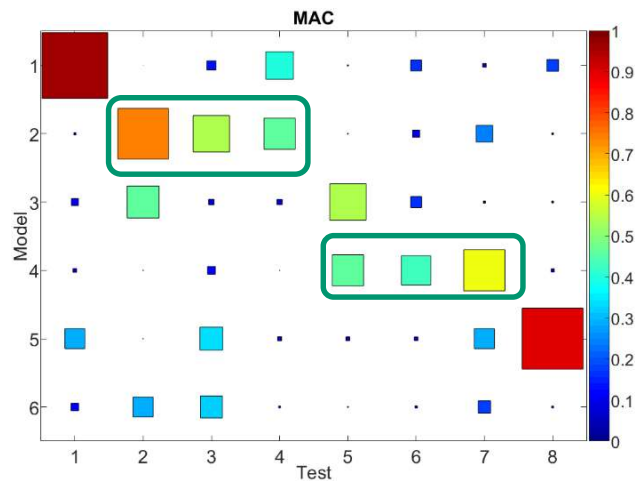


Test
29 @ 4012 Hz 0.12 %

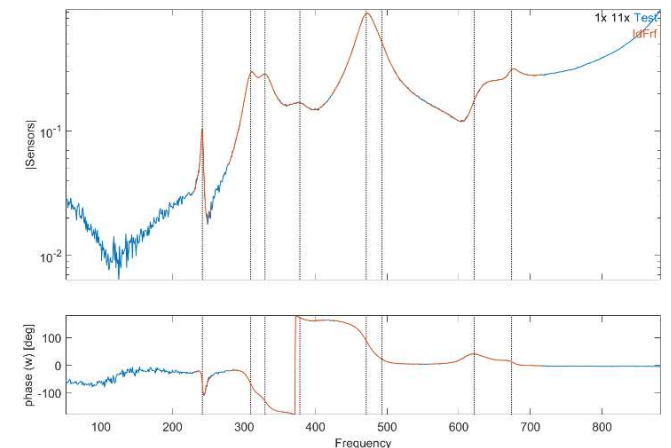
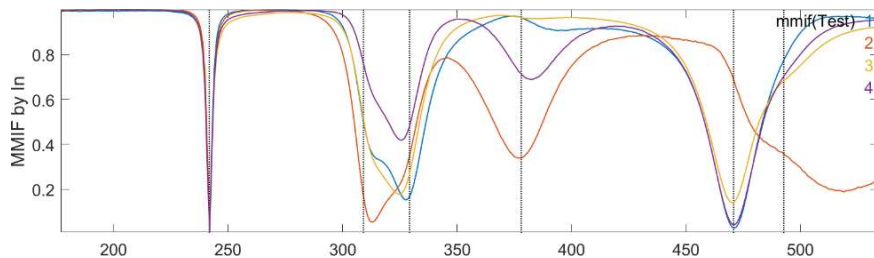


Other errors : local modes

- « Same mode » multiple times



- non structural masses generate global mode duplicates
- « small mass » $\neq$ « tiny peak » (if this were true proof mass dampers would not work)



Subspace methods : IO (ERA)

$$[h(k\Delta t)]_{pq} = \begin{bmatrix} [y(k)] & [y(k + j_1)] & \dots & [y(k + j_{q-1})] \\ [y(k + i_1)] & [y(k + i_1 + j_1)] & \dots & [y(k + i_1 + j_{q-1})] \\ \vdots & \vdots & \ddots & \vdots \\ [y(k + i_{p-1})] & [y(k + i_{p-1} + j_1)] & \dots & [y(k + i_{p-1} + j_{q-1})] \end{bmatrix}$$

$$[h(k\Delta t)]_{pq} = [C]_p [A]^{k-1} [B]_q = \begin{bmatrix} [C] \\ \vdots \\ [C] [A]^{j_{p-1}} \end{bmatrix} [A]^{k-1} \begin{bmatrix} [B] & \dots & [A]^{j_{q-1}} [B] \end{bmatrix}$$

$$[h(k\Delta t)]_{pq} \approx [U] [\backslash S \backslash] [V]^T$$

$$[A] = \left[[\backslash S \backslash]^{-1/2} [U]^T [h(2\Delta t)]_{pq} [V] [\backslash S \backslash]^{-1/2} \right]$$

$$[B] = [\backslash S \backslash]^{1/2} [V]^T \begin{bmatrix} [\backslash I \backslash] \\ [0] \\ [0] \end{bmatrix}_{qNA \times NA}$$

$$[C] = \begin{bmatrix} [\backslash I \backslash] & [0] & [0] \end{bmatrix}_{NS \times pNS} [U] [\backslash S \backslash]^{1/2}$$

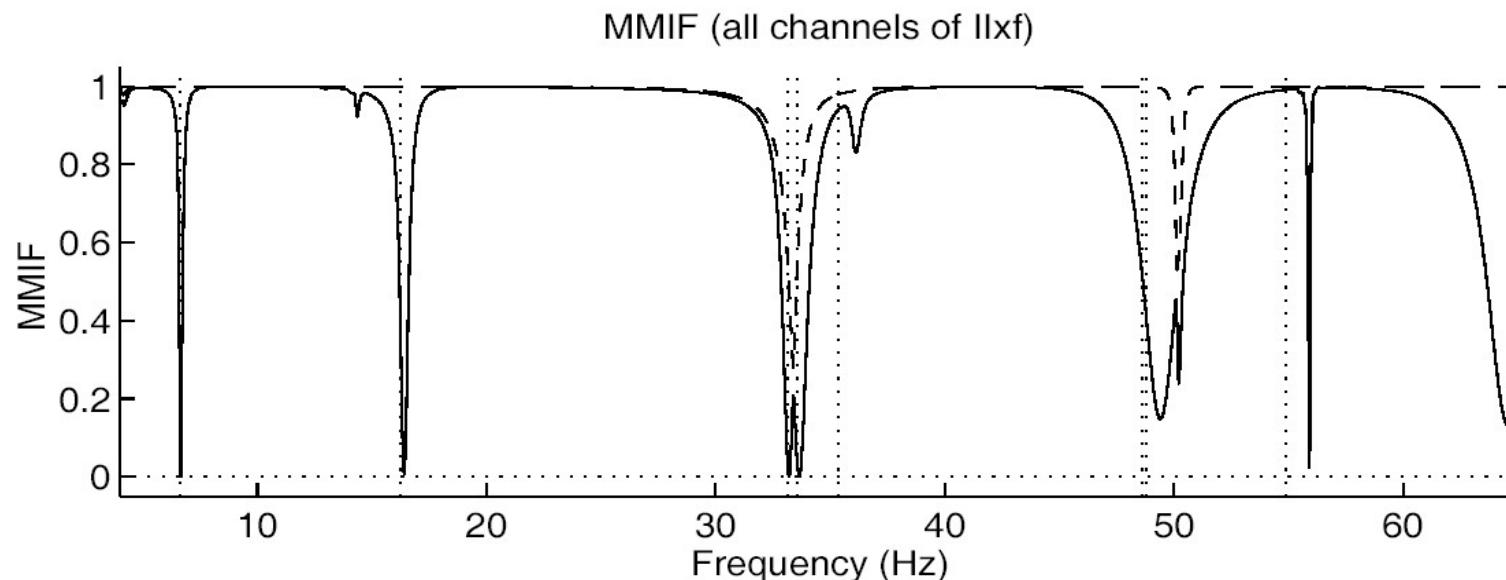
[1] J. N. Juang and R. S. Pappa, "An Eigensystem Realization Algorithm (ERA) for Modal Parameter Estimation and Model Reduction," *J. Guidance, Control, and Dynamics*, vol. 8, no. 5, pp. 620–627, 1985.

[2] E. Reynders, "System identification methods for (operational) modal analysis: review and comparison," *Archives of Computational Methods in Engineering*, vol. 19, no. 1, pp. 51–124, 2012.

Order selection : MMIF

- Mode indicator functions
- Related to **force appropriation** (do by hand)
- Appropriation excites a single mode (allow picture taking)

$$MMIF_i = \min_{u \perp u_{j < i}} \frac{u^T [Re(H)^T Re(H)] u}{u^T [H^H H] u}$$

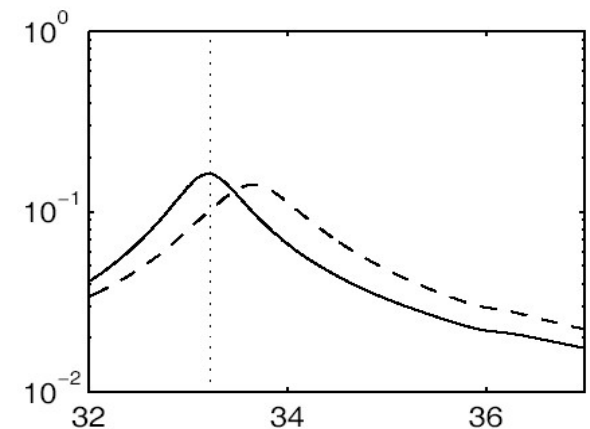
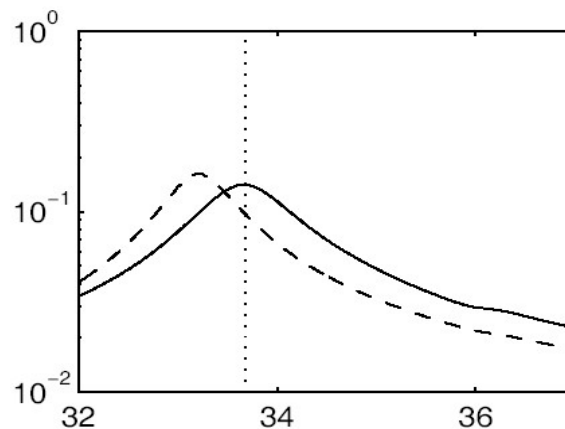
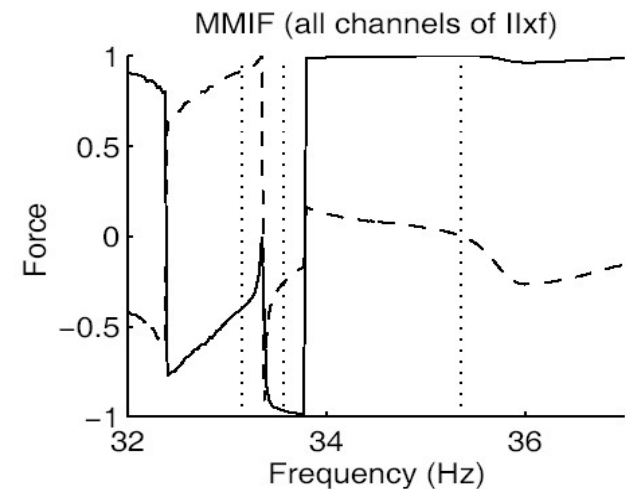
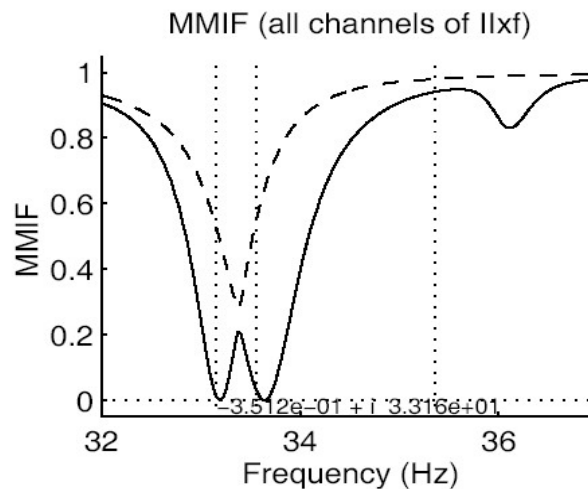


Order selection : MMIF

- Multi exciter allows spatial filter $\{u_i(t)\}_{NA} = \{w_i\}_{NS}a(t)$
- Appropriation : choose $\{w_i\}_{NS}$ to make other modes unobservable
- Keywords

Modal coordinates

Modal filter

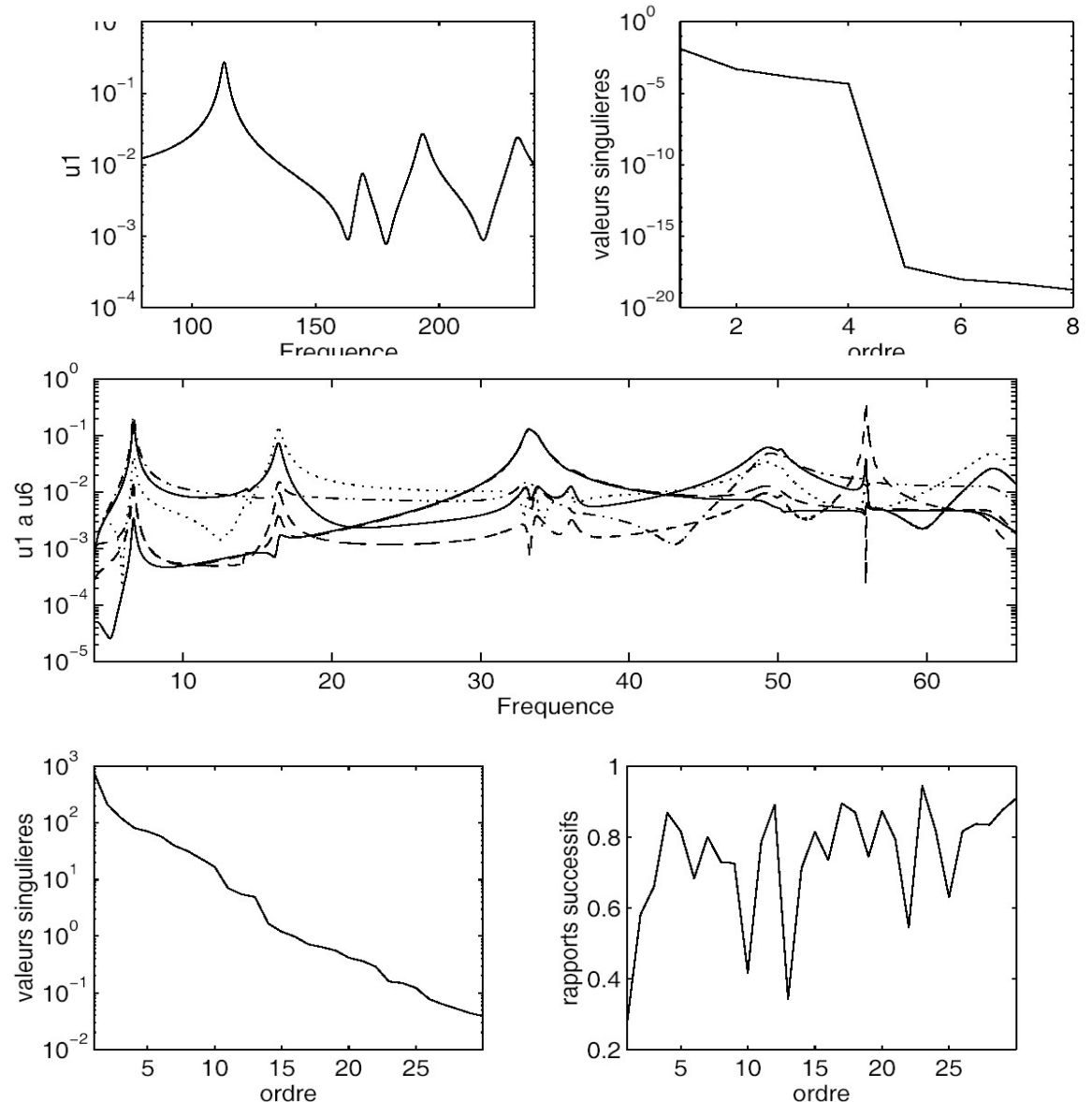


Order selection : SVD

$$[H_{kl}(\omega)]_{NW \times N_{kl}} = \left[\frac{1}{\omega_{1:NW} - \lambda_{1:NM}} \right]_{NW \times NM} [R_{j,kl}]_{NM \times N_{kl}}$$

- Rank of H is limited
- Perfect for simulation
- Problem for real data
- More useful for data volume reduction

P. Guillaume & al. 2005
Improved techniques for
outlier detection



Properties of residues

- Minimal $\frac{R_j}{s - \lambda_j} = \frac{\{c\psi_j\} \{\psi_j^T b\} + \{c\psi_{j+1}\} \{\psi_{j+1}^T b\}}{s - \lambda_j}$
- Reciprocal $([c_{col}] \{\psi_j\})^T = \{\psi_j\}^T [b_{col}]$
- Second order model (modal damping)

$$\frac{[R_j]}{s - \lambda_j} + \frac{[\bar{R}_j]}{s - \bar{\lambda}_j} = \frac{2(s + \zeta_j \omega_j) \operatorname{Re}(R_j) + 2\omega_j \sqrt{1 - \zeta_j^2} \operatorname{Im}(R_j)}{s^2 + 2\zeta_j \omega_j s + \omega_j^2}$$

$$[R_j] = \frac{T_j}{i\omega_j \sqrt{1 - \zeta_j^2}} = [c] \frac{\phi_j}{\sqrt{i\omega_j \sqrt{1 - \zeta_j^2}}} \frac{\phi_j^T}{\sqrt{i\omega_j \sqrt{1 - \zeta_j^2}}} [b]$$

- Modal mass (reciprocal, second order)

$$\frac{1}{\mu_j(c_{col})} = c_{col} \phi_j \phi_j^T b_{col} = (c_{col} \phi_j)^2 = T_{jcol} = i\omega_j \sqrt{1 - \zeta_j^2} R_{jcol}$$

- Second order non proportional damping