

Vibrations des structures & acoustique

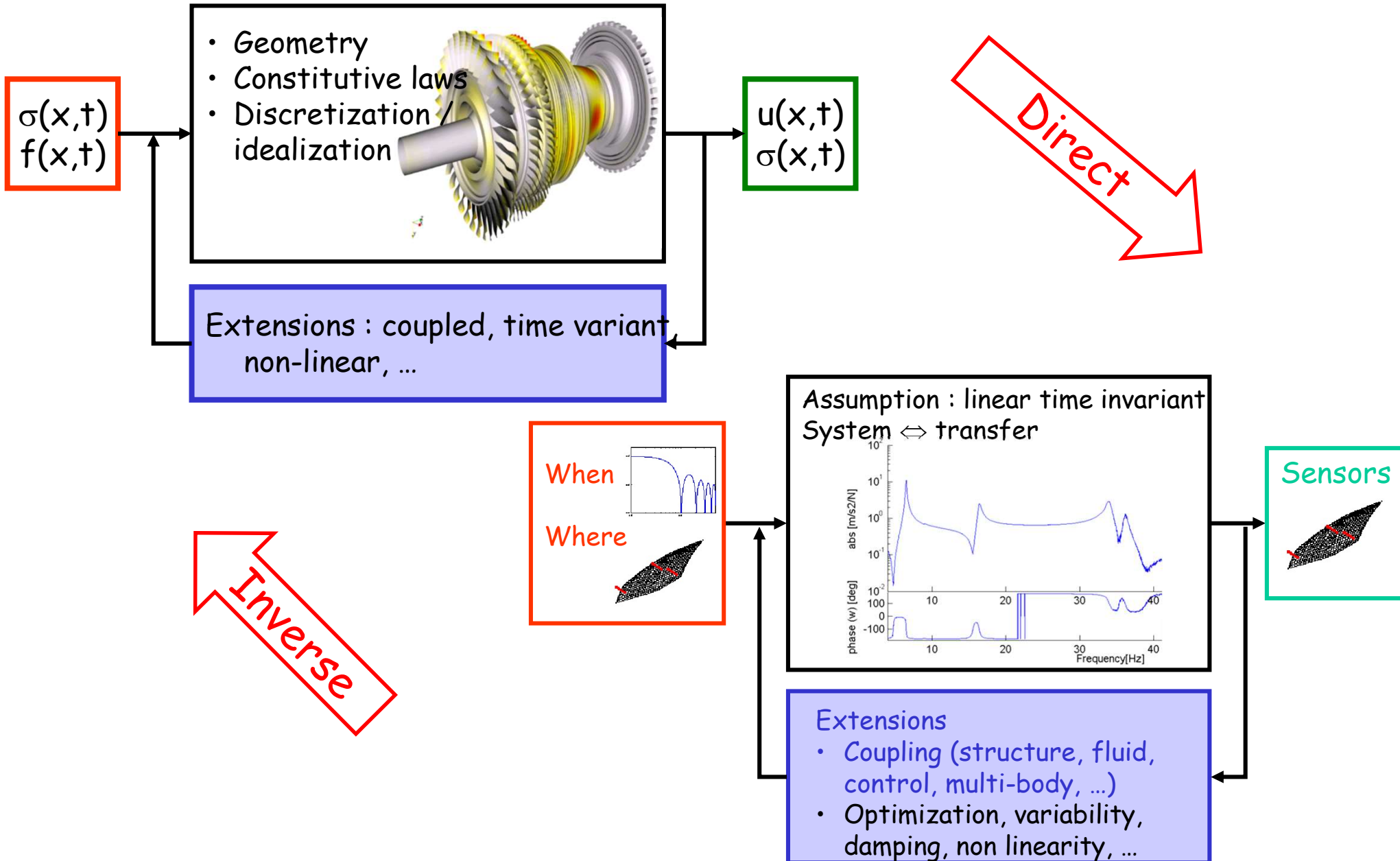
TP1 : modélisation

- modèles système
- problèmes directs et inverses

Etienne Balmès,
ENSAM/PIMM, SDTools

<http://savoir.ensam.eu/moodle/course/view.php?id=2158>
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FEM model / system model

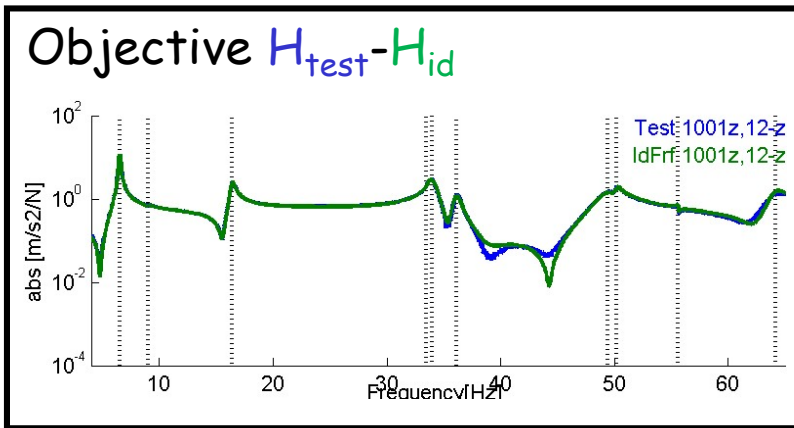
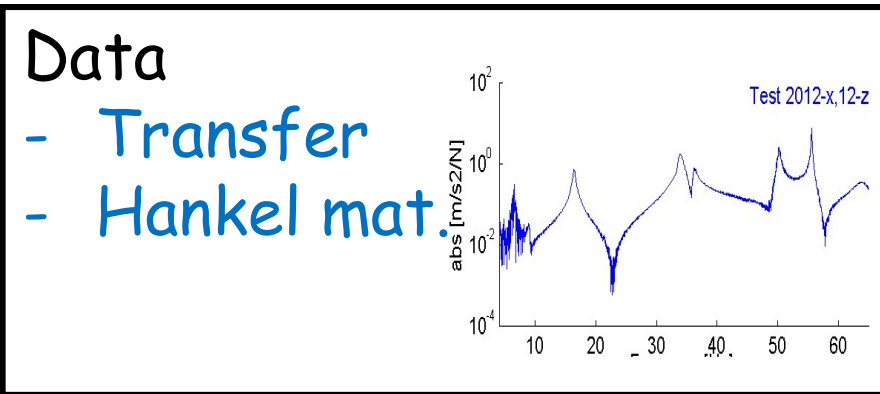


Lab work outline

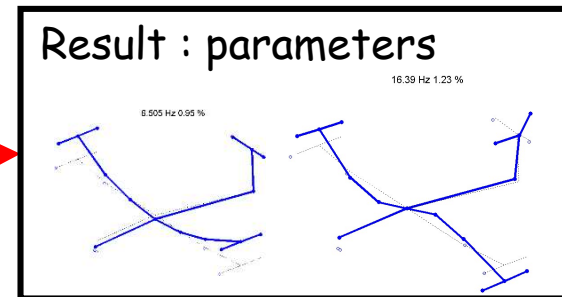
- How are modes measured
 - What is a transfer
 - Identification
 - Content of modal test report
- Test and FEM modes
 - What is a FEM/test mode
 - MAC (modal assurance criterion)
- How are modes predicted
 - Data needed for mode computations
 - Transfers, observability, commandability
- Further steps
 - Damping
 - Sensitivity of frequencies

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Experimental modal analysis: an inverse problem



Optimization



Family of models

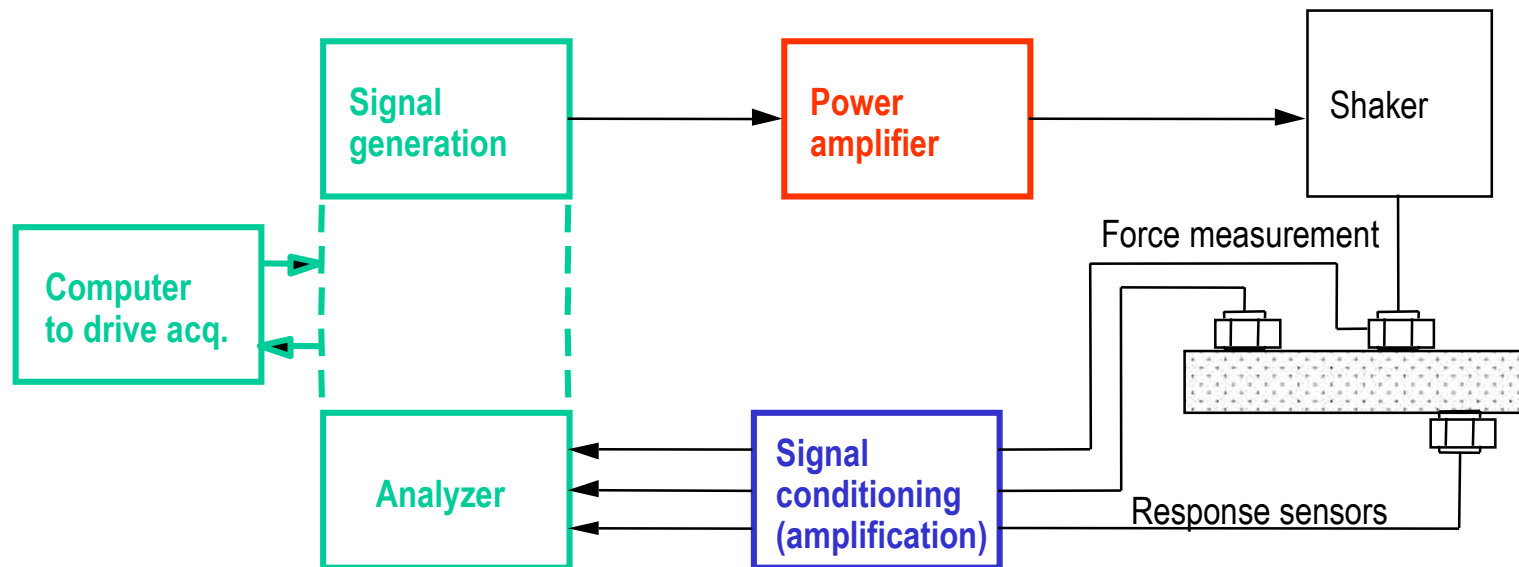
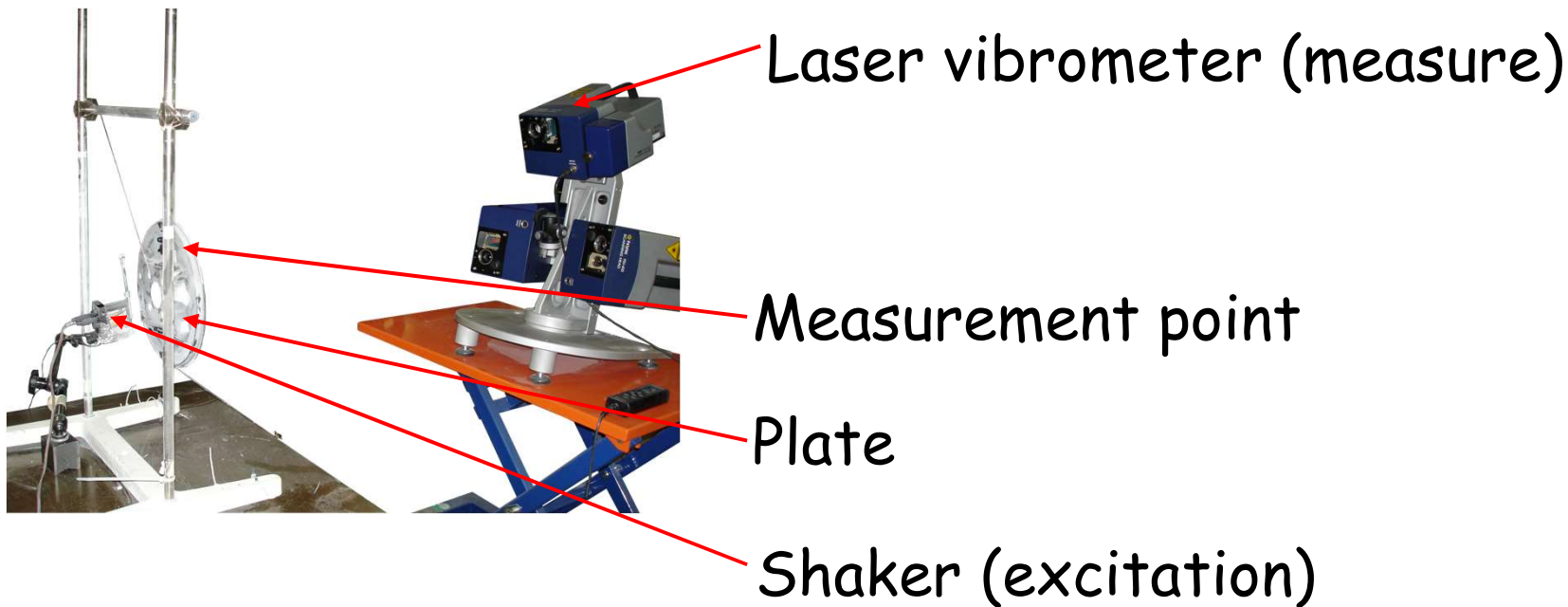
SVD $\sum_i \begin{Bmatrix} U_i \\ \downarrow \\ \text{sensors} \end{Bmatrix} \{a_i(t) \rightarrow \text{time}\}$

State space $\{\dot{x}\} = [A]\{x\} + [B]\{u\}$

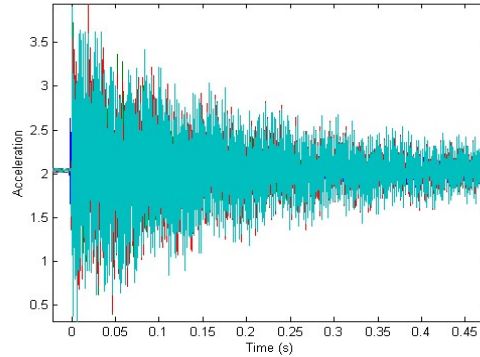
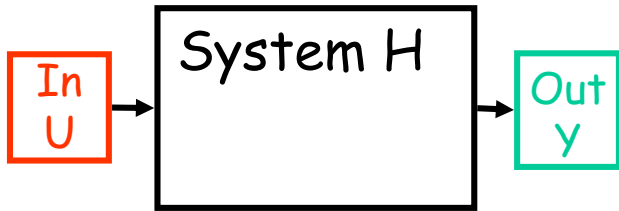
Pole residue $\sum \frac{[R_j]_{NS \times NA}}{s - \lambda_j}$

Rational fraction, modal model, second order ...

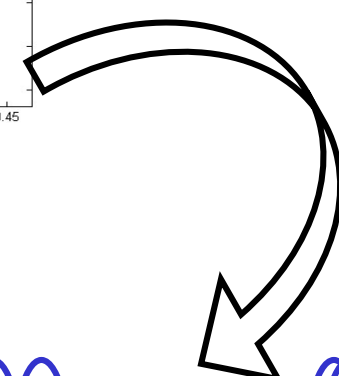
Experimental modal analysis : measurements



Modal analysis : transfers

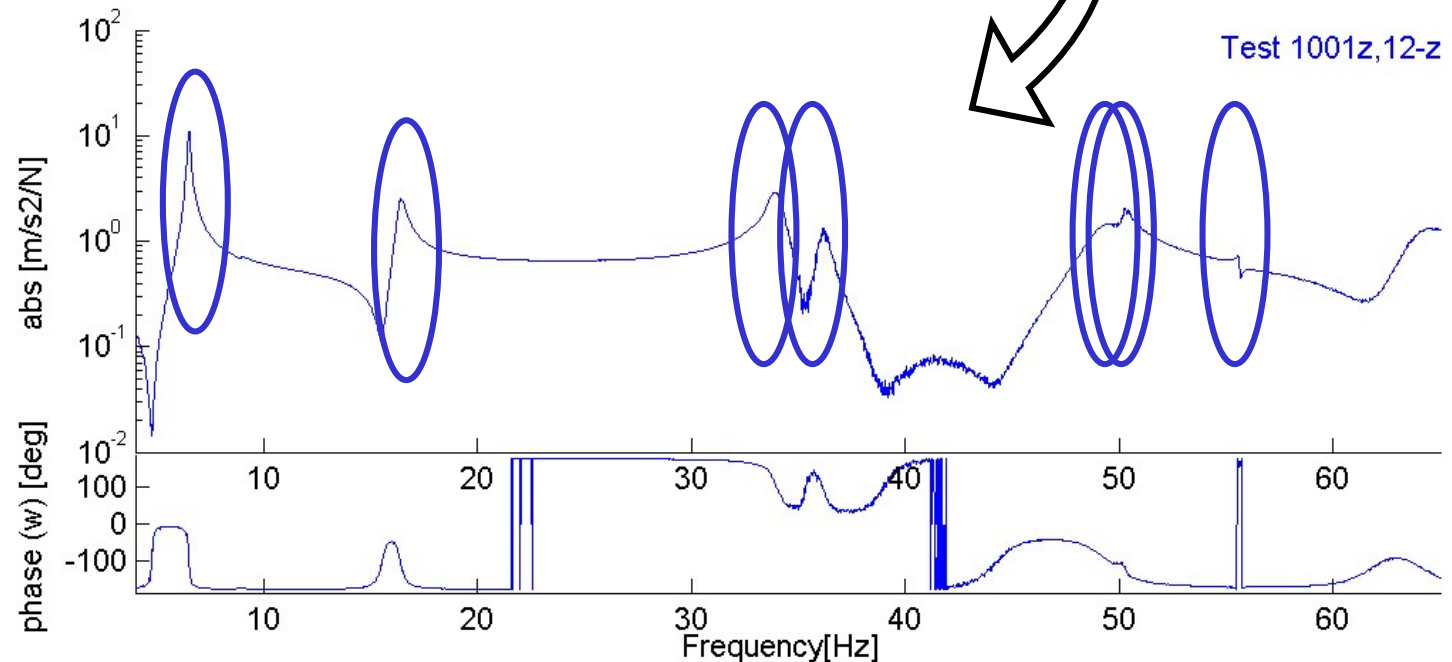


Transfers estimated from time response



ONE input
ONE output

$$\{Y(\omega)\} = [H(\omega)]\{U(\omega)\}$$

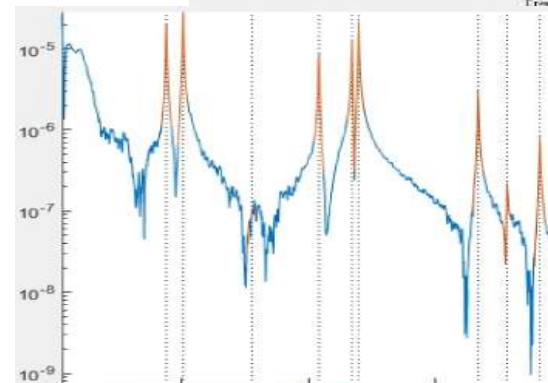
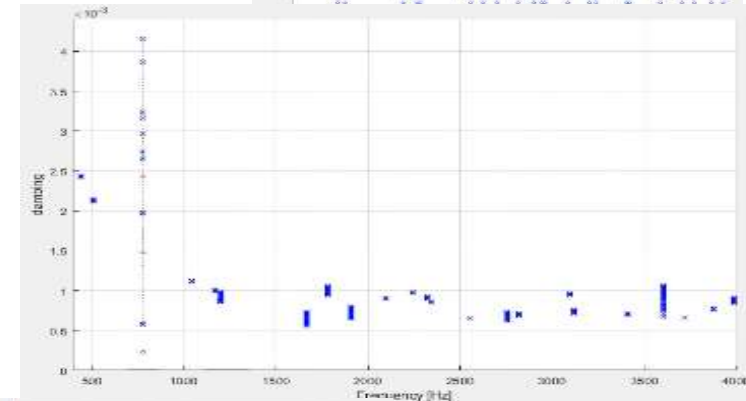
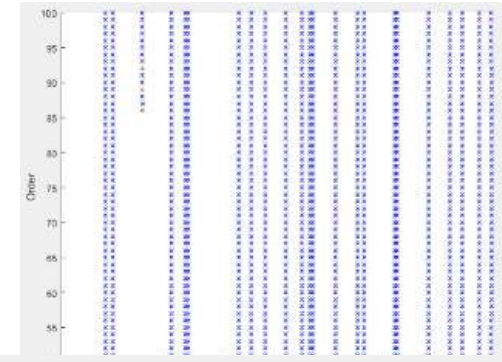


MANY resonances

Bode plot : visualization of transfer function

Identification phases

- Initialization pick from :
 - Single pole estimate [1]
 - Stabilization diagram [2]
- Estimate by band (why ? [1,3])
- How can problems be detected ? [3-4]
- Re-optimize poles (why ?)



- [1] E. Balmes, « Frequency domain identification of structural dynamics using the pole/residue parametrization », IMAC 1996
- [2] P. Verboven, « Frequency-domain system identification for modal analysis », Ph.D. thesis, 2002.
- [3] G. Martin, « Méthodes de corrélation calcul/essai pour l'analyse du crissement », Ph.D. CIFRE SDTools, Arts et Metiers ParisTech, Paris, 2017
- [4] G. Martin, E. Balmes, et T. Chancelier, « Characterization of identification errors and uses in localization of poor modal correlation », MSSP 2017.

Outline

Model validation and verification

CAD Model



Experimental model

19 @ 910 Hz g1e+10, dEk 62 % dY 0 %

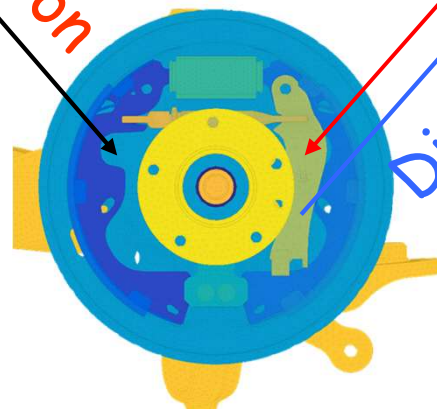


1990 : model updating
2000 : virtual prototype
2019 : digital twin

Continuous model
+interfaces

$$\begin{aligned} \text{On } \Omega & \quad \text{div} \sigma + F_v = \rho \ddot{u} \\ \text{On } \partial \Omega & \quad u(x, t) = u_{\text{given}} \\ & \quad \{T\} = [\sigma] \{n\} \end{aligned}$$

FE/numerical
model

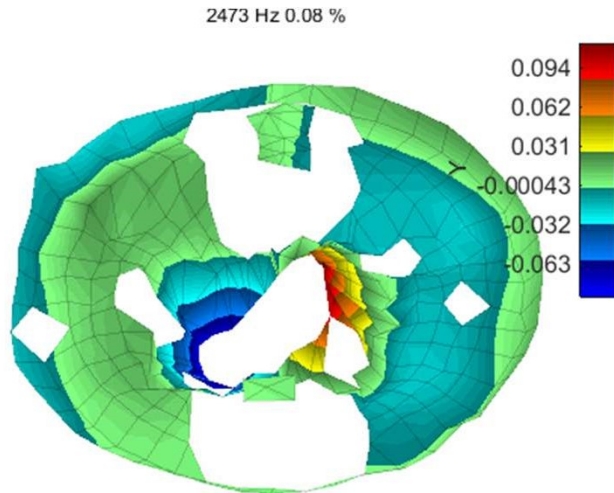


Verification
Design

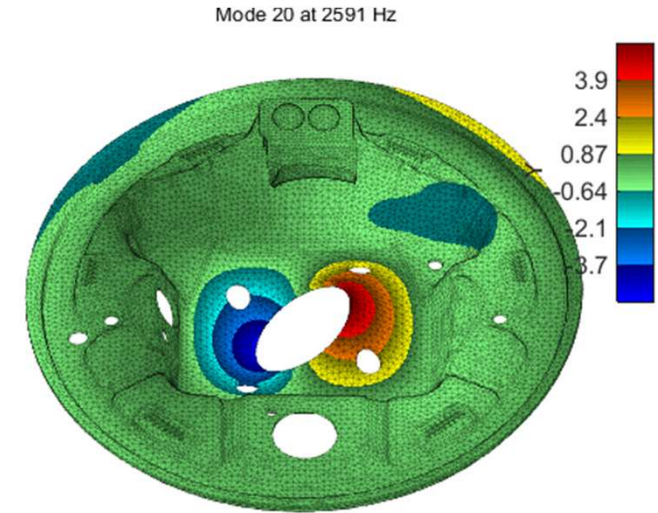
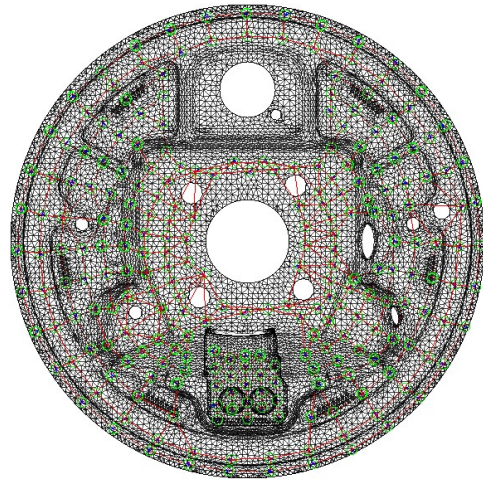
Validation
(Updating)

Dispersion

Comparing test & FEM



Identification
known @ sensors



FEM known @ nodes

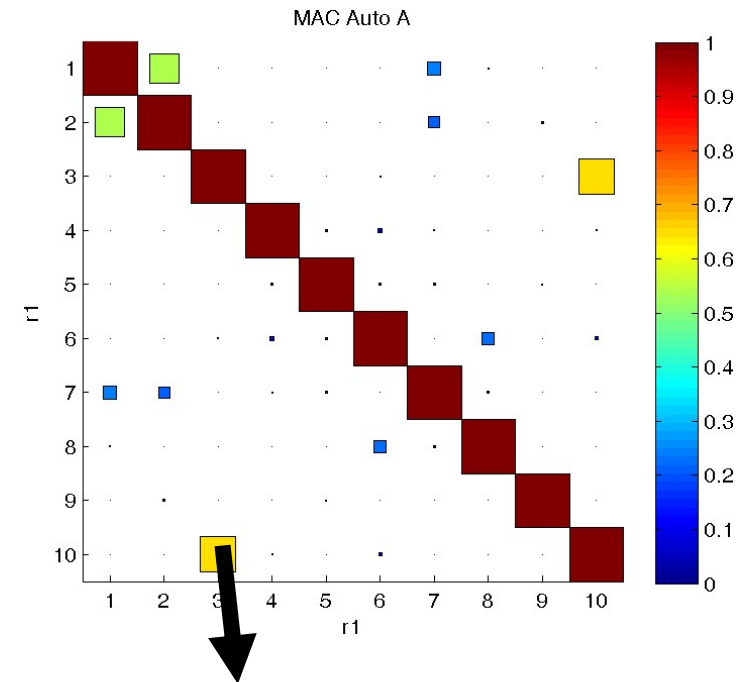
Topology correlation
= observe FEM @ sensors

$$\{y(t)\} = [c] \{q(t)\}$$

MAC : comparing shapes

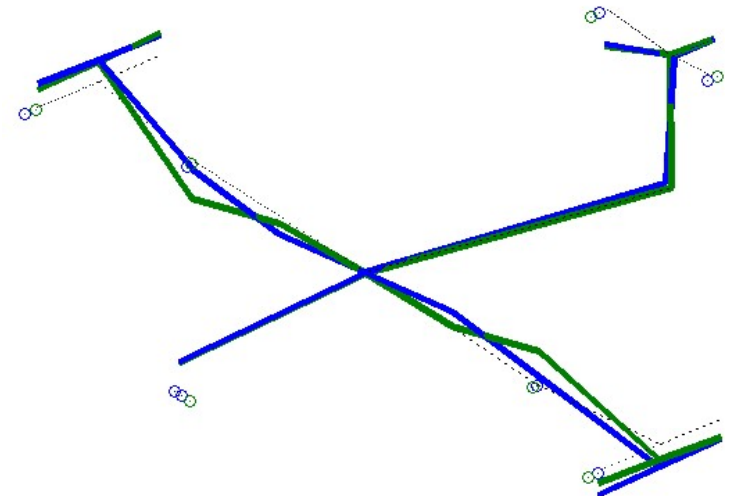
Shapes are compared through correlation coefficient (Modal Assurance Criterion)

$$\text{MAC}(U, V) = \frac{|\{U\}^H \{V\}|^2}{|\{U\}^H \{U\}| |\{V\}^H \{V\}|}$$

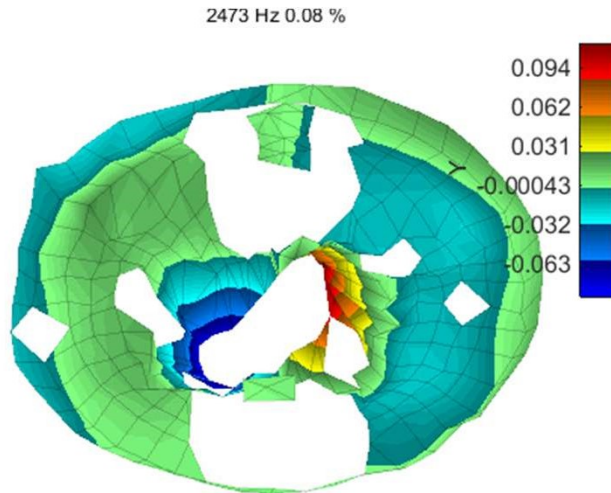


16.39 Hz 1.23 %, 64.16 Hz 1.22 %

Next step : modal updating (*recalage*) = use correlation to correct model parameters

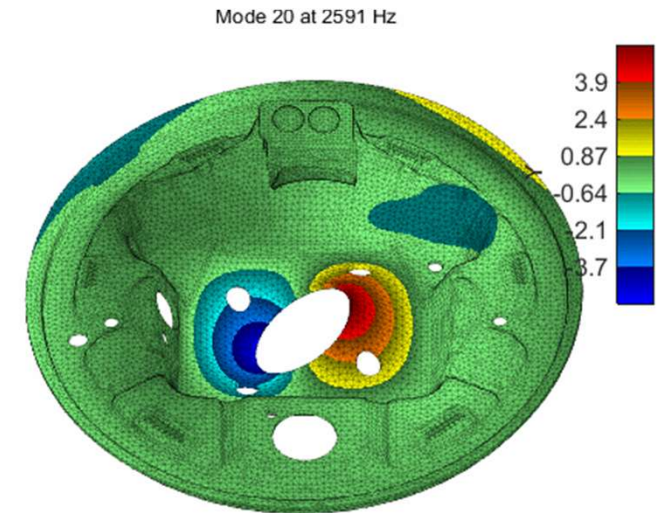


Where is the error?



Topology errors

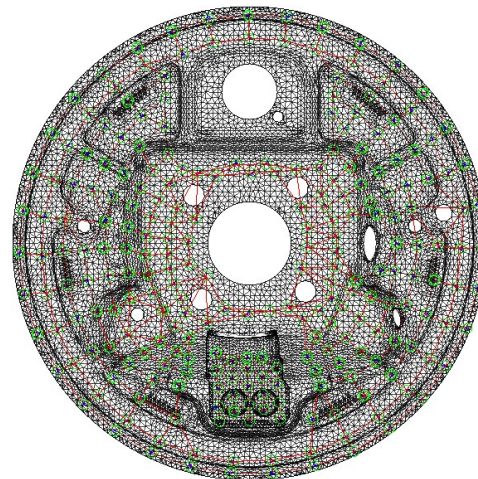
- sensor/act position
- matching



Y
x

Identification error

- Noisy measurements
- Identification bias
- NL, time varying, ...

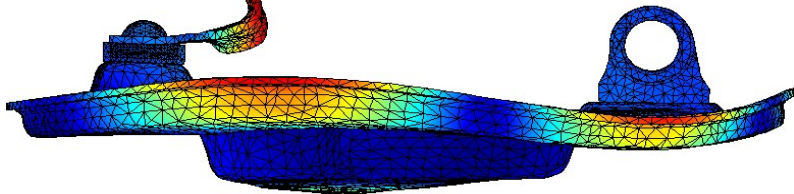
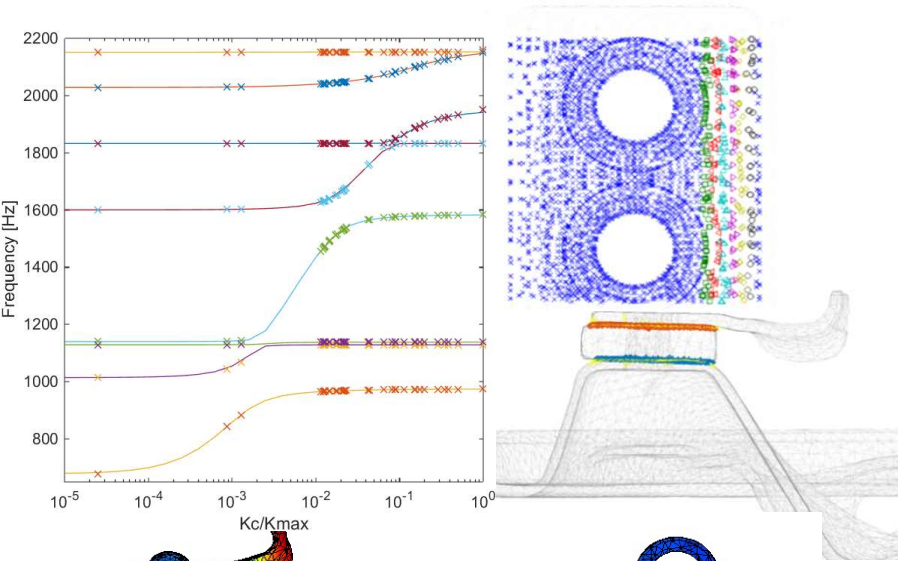


FEM error

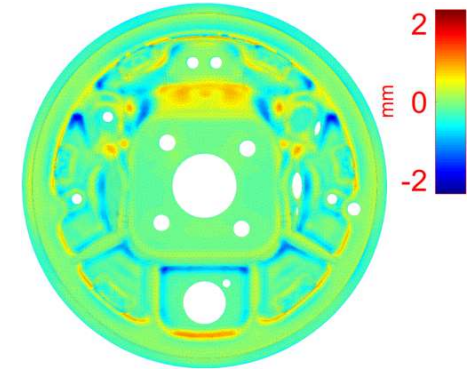
- Geometry
- Material parameters
- Contact properties

Parametrization

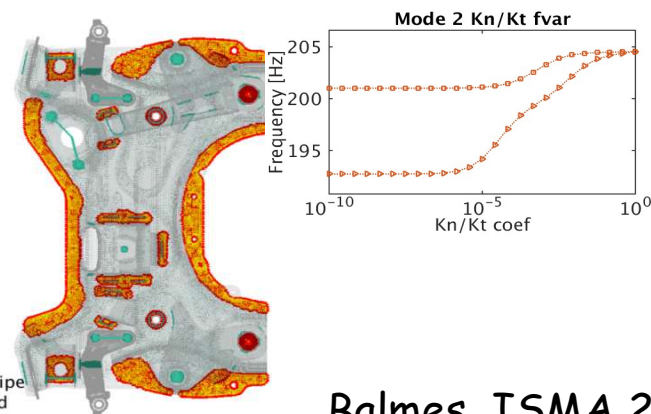
- Variable contact **surface**, **contact**, **sliding**



Chassis Brakes International Eurobrake 2014

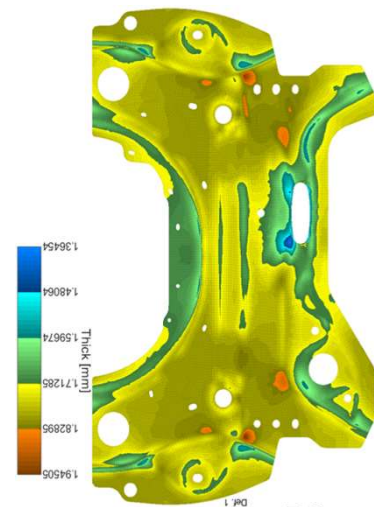


Variable geometry



Berceau_Equipe Fully Coupled

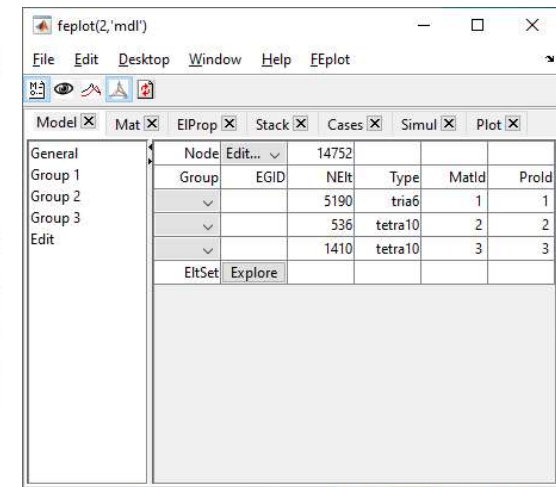
Balmes, ISMA 2016



Outline

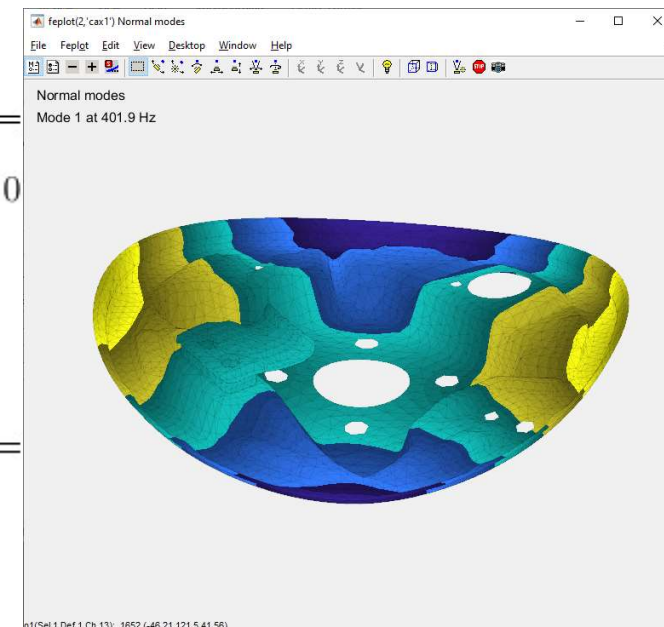
What is a model

- within the domain Ω , one verifies the PDE $\rho A \frac{\partial^2 u}{\partial t^2} - \frac{\partial}{\partial x} (EA \frac{\partial u}{\partial x}) = f_v(x)$. This requires
 - a description of the domain (typically FEM geometry)
 - a problem formulation (you have seen : compression, bending, torsion, continuous mechanics = 3D solid, possibly plate/shell)
 - constitutive parameters to represent the material
- on the edge of the spatial domain $\partial\Omega$, one verifies *boundary conditions* in displacement (clamped), force (free or point load), or combination (added spring). You should remember that boundary conditions provide sufficient data to compute power flow : displacement or force but not both. Or a function that gives force as a function of displacement (constitutive law on edge).
- on the edge of the time domain, one talks about *initial conditions*.

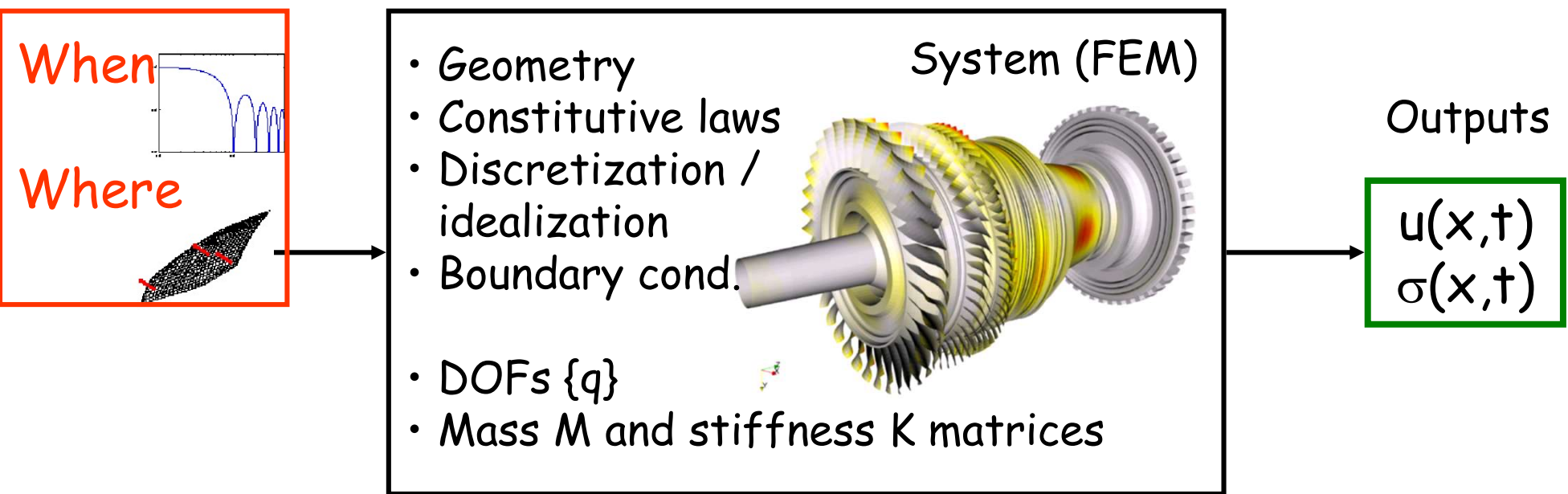


Thus one can compare the equations needed in the strong and weak forms

	strong/local	weak/PVW
On Ω	$\rho A \frac{\partial^2 u}{\partial t^2} - \frac{\partial}{\partial x} (EA \frac{\partial u}{\partial x}) = f_v(x)$	$\int_{\Omega} (\rho A u^* \ddot{u} + u^*_{,x} EA u_{,x} - u^* f_v) = 0$
On $\partial\Omega$		
- Right fixed	$u(x_R) = 0$	$u(x_R) = 0$
- Right free	$F_x = EA u_{,x} _{x_R} = 0$	– (not needed since no work)
- Right loaded	$F_x = EA u_{,x} _{x_R} = F_{ext}$	$P_{ext} = [b] F_{ext}(t) = u^*(x_R) F_{ext}(t)$
- Spring at right	$EA u_{,x}(L) - ku(L) = 0$	$\int_{\Omega} \dots + ku(L) u^*(L) u(L) = F_{ext}$



How are transfers predicted ?



Inputs (when)

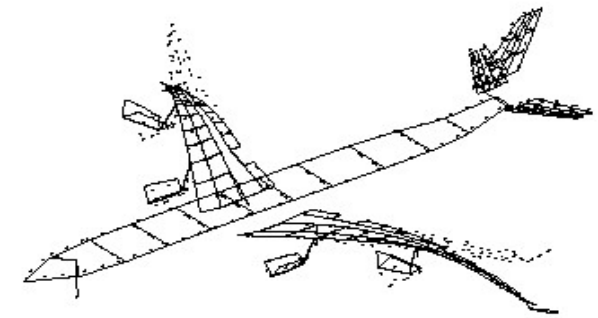
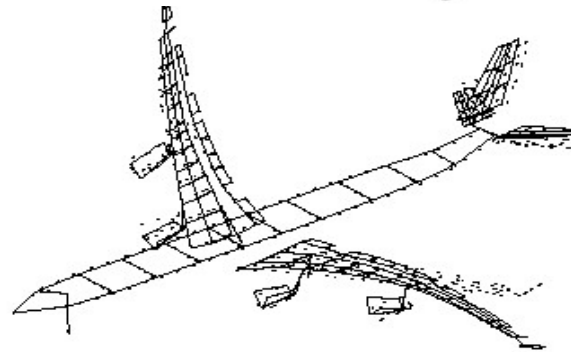
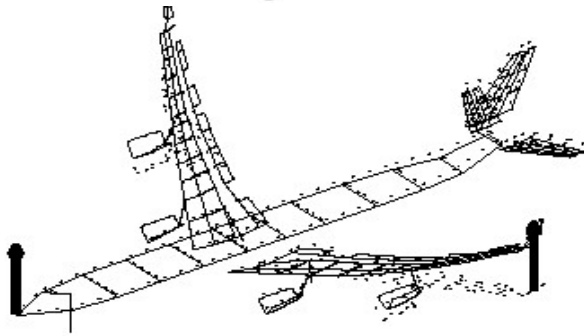
- Unbalance : harmonic at Ω
- Aerodynamic loads ($n\Omega$)
- Rotor/stator contact

Inputs (where)

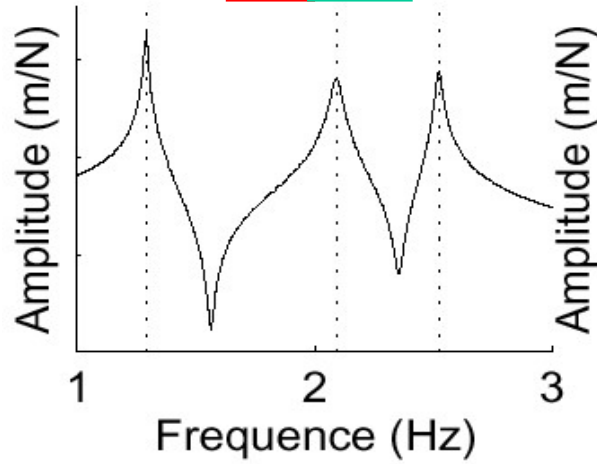
- Point mass
- Distributed pressure
- Blade tip

Observability/controlability

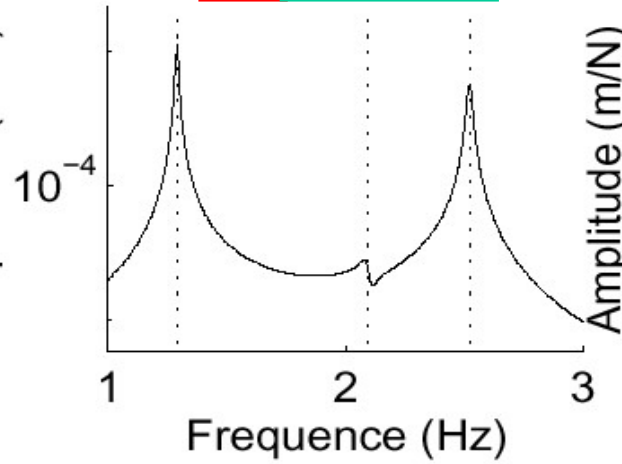
$$H(s) = \sum_{j=1}^N \frac{\boxed{[c]}\{\phi_j\}\{\phi_j\}^T\boxed{[b]}}{s^2 + 2\zeta_j\omega_j s + \omega_j^2}$$



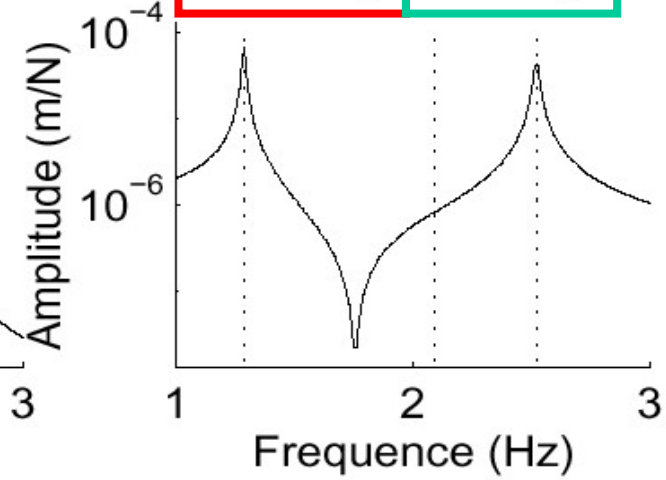
Aile /Aile



Aile /Fuselage



Fuselage /Fuselage



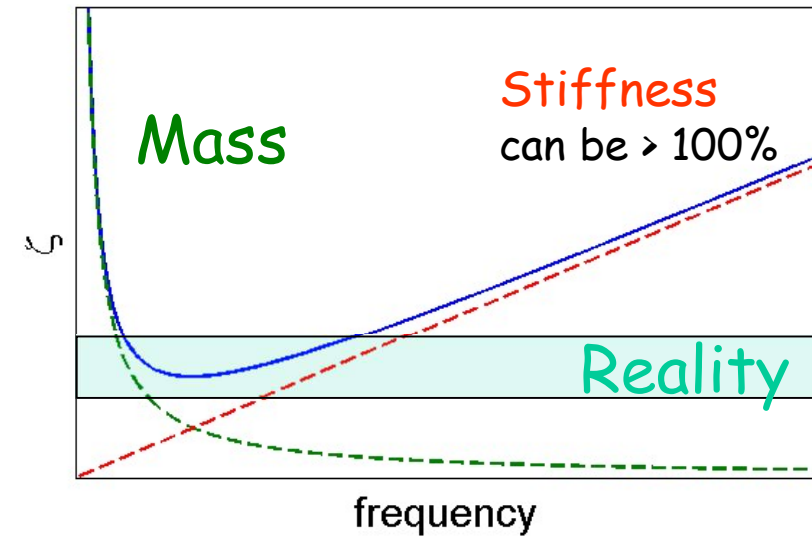
Rayleigh damping (CM2)

- Rayleigh damping:

- Physical domain $[C] = \alpha[M] + \beta[K]$

- Modal domain

$$\zeta_j = \frac{\alpha}{2\omega_j} + \frac{\beta\omega_j}{2}$$



Stiffness perturbation in modal coord. (CM4)

- Stiffness perturbation

$$[M]\{\ddot{q}\} + [K + \Delta K]\{q\} = [b]\{u\}$$

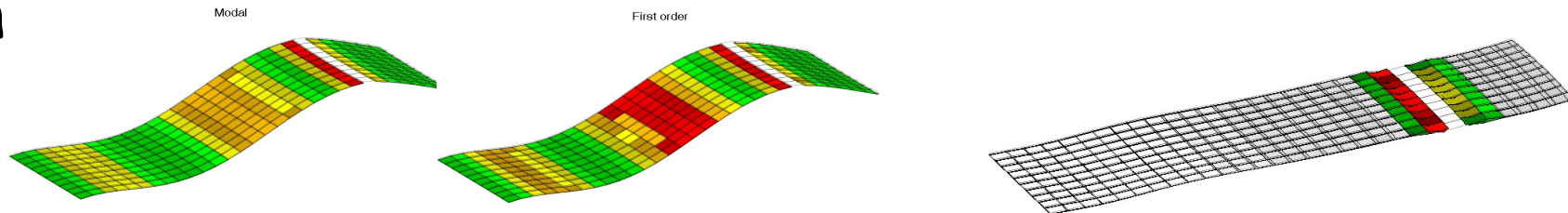
- Modal coordinates **reanalysis** $\{q\} = [\phi]\{q_R\}$

$$[I]\{\ddot{q}_R\} + \left[\begin{matrix} \omega_j^2 \\ \omega_j^2 \end{matrix} \right] + \phi^T \Delta K \phi \{q_R\} = [\phi^T b]\{u\}$$

- Sensitivity on frequency (TD4 question 5)

$$\frac{\partial \omega_j^2}{\partial p} = \{\phi_j\}^T \left[\frac{\partial K}{\partial p} - \omega_j^2 \frac{\partial M}{\partial p} \right] \{\phi_j\}$$

- Need to know :
may be **significantly wrong** without residual terms/static correction



Procédure COVID

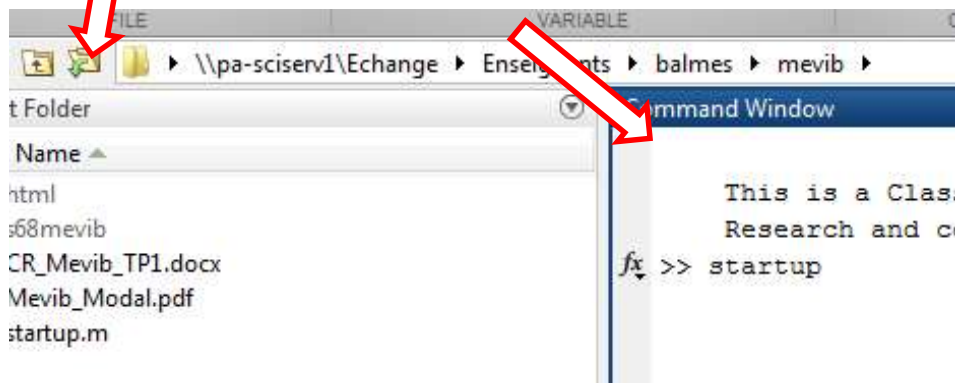
1. Copie slides [Mevib_TpModal.pdf](#)
2. Sujet PDF avec lien vers les films parties [SujetTpModal.pdf](#)
3. Posez des questions dans le chat de la réunion
4. Trame de compte rendu

Vous pouvez utiliser Office365 ou google-docs pour partager la rédaction par binôme.

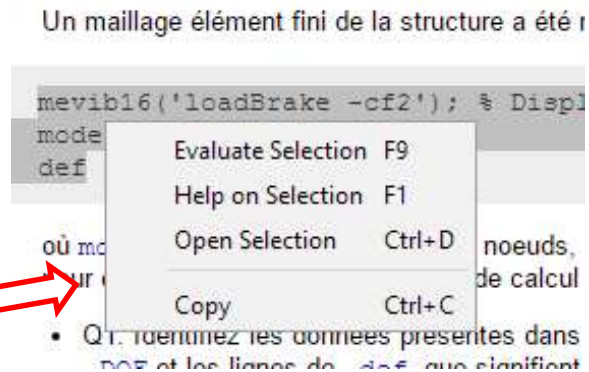
Compte rendu = élément de notation (délais un semaine au plus) :
etienne.balmes@ensam.eu

Ouverture sujet dans Matlab

1. \\intram\paris\Echange\Enseignants\balmes\Mevib
(ou peut être X:\Enseignants\balmes\Mevib)
2. startup



3. Sélection + F9 (evaluate)



Vous pouvez utiliser Office365 ou google-docs
pour partager la rédaction
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